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COMPUTER-AIDED CONTROL DESIGN SYSTEMS

A THESIS
PRESENTED IN PARTIAL FULFILMENT
OF THE REQUIREMENTS FOR THE DEGREE
OF
MASTER OF TECHNOLOGY IN INDUSTRIAL
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NEW ZEALAND.

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SYNOPSIS

This thesis is concerned with the application of interactive computer-aided design techniques. These techniques are graphical and based upon conventional control theory. They are the time-response, frequency-response and root-locus techniques.

The design philosophy based on the above design techniques is that an appropriate compensator is inserted and the overall system performance is studied. The graph representing the system performance after each adjustment of the compensator is adjusted until the performance specifications are met.

The Computer techniques are developed which, having specified the fixed configurations of the original system, permit varied choice of the compensator. For each selection, the control system may be investigated in either open- or closed-loop form. The software for computer-aided control system design is programmed in Algol and operates on the Massey University B6700 computer system.

The program is operated via a Tektronix 4010 visual display terminal in an online interactive mode. The control designer, at the Tektronix console, is able to specify certain designs and evaluate their effectiveness in terms of various graphical analyses produced by the computer and displayed on the screen of the Tektronix.

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CHAPTER ONE

INTRODUCTION

This thesis is concerned with the application of computer-aided design to classical control theory. In classical control theory, much of the effort of control system design is concerned with the generation of charts for such aspects as time response, root locus and frequency response characteristics. These charts are time consuming to prepare by hand, even though short cut techniques are often available to obtain sketches which closely approximate the actual characteristics.

A digital computer is able to assist in these conventional design procedures by generating the appropriate charts for inspection by the designer, thereby taking much of the manual effort out of the design processes.

This thesis describes the development of an interactive design package. Software for computer-aided control system design has been programmed to operate, via a visual display terminal, in an online interactive mode. The control designer, using the terminal console, is able to specify certain designs and evaluate their effectiveness in terms of the various graphical analyses produced by the computer and displayed on the screen of the terminal.

The computer-aided approaches to control system design are applicable to linear models. Although this assumption of linearity is not exactly valid for most practical systems, it is a realistic approach for three reasons:

1. Many systems are essentially linear under normal operating conditions.
2. Slightly nonlinear systems usually may be approximated with sufficient accuracy by assuming linearity.

3. At present adequate mathematical methods for solving non-linear equations have not been developed.

In this thesis, the system characteristics are determined by time-response, frequency-response and root-locus methods. The time-response method is a direct way of considering the system performance in the time-domain. The output response is derived for given inputs. The direct solutions of the differential equation of the system following a step input are plotted on an "output versus time" graph. The system performance is specified in terms of the information obtainable from this graph.

For the frequency-response methods, the frequency of sinusoidal oscillation of the input signal is varied over a certain range and the resulting frequency-response is studied. The results are plotted on graphs such as the Bode diagrams, Nyquist diagrams and Nichols charts. The root-locus method is a graphical method used to evaluate the roots of the characteristics equation of the system transfer function and determine the influence of the system parameters on these roots. This method provides a very effective way to carry out many design problems with requirements specified in the time-domain.

The design philosophy based on the above graphical approaches is that the graph representing the system performance, obtained by any of the above methods, is shaped and reshaped until the performance specifications are met. To achieve this, appropriate compensation networks are usually required within the control system structure.

Computer techniques have been developed which, having specified the fixed configurations of the control system, permit varied choice of the compensator. For each selection, the control system may be investigated - either in open- or closed-loop form - using

any of the above methods. The software for computer-aided control system design is programmed in Algol and operates on the Massey University B6700 computer system. The interactive terminal is via a Tektronix 4010 visual display unit. The designer, at the Tektronix console, is able to adjust the designs in order to improve the system performance by shaping various graphs displayed on the screen of the Tektronix. Given a particular control system structure, which may be either open-loop or closed-loop, the designer may request the following options:

- A display of the system time response following a step input.
- A Bode diagram display of the system frequency response.
- A Nyquist diagram display of the system frequency response.
- A Nichols chart display of the system frequency response.
- § - A Root-locus plot display of the roots of the system characteristic equation as the system gain varies.

The designer seeks to satisfy all performance specifications by means of educated trial-and-error repetition. Although the design is based on a trial-and-error procedure, the experience of the designer plays an important role in a successful design. An experienced designer will be able to achieve an acceptable system design using fewer trials.

Chapter Two in this thesis describes the analysis and design techniques that have been utilized, namely time-domain, frequency-domain and root-locus techniques. Chapter Three outlines the structure of the software for computer-aided control system design, detailing the methods employed to implement some of the concepts discussed in chapter two, and also specifies how the system is used. Chapter Four presents several illustrative design examples, along with the computer generated graphs.

CHAPTER TWO

SYSTEM ANALYSIS AND DESIGN

2.1 INTRODUCTION

This chapter describes some of the analysis and design techniques for linear control systems. These techniques are graphical and based upon conventional control theory.

A computer is able to assist in these design procedures by generating the appropriate charts for inspection by the designer, thereby taking much of the manual effort out of the design processes.

To design a control system to meet given performance specifications, it is necessary to determine the characteristics of the system that is to be controlled either in the time domain or frequency domain. Given such characteristics, several graphical methods are available for analysis and design, namely time, response, root-locus and frequency response methods.

The frequency response methods consist of the Bode, Nyquist and Nichols chart techniques. The design philosophy based on these graphical approaches is that the graph representing the system performance, obtained by any of the above methods, is shaped and reshaped until the performance specifications are met. To achieve this objective, appropriate compensation networks are usually required. The control system configuration that is considered is shown in Figure 2.1 (it is assumed that $G(s)$ and $H(s)$, are fixed configurations over which the designer has no control).

Computer techniques have been developed which, having specified $G(s)$ and $H(s)$, permit varied choice of $G_c(s)$. For each selection, the above system may be investigated - either in open-loop or closed-loop form - using any of the above methods.

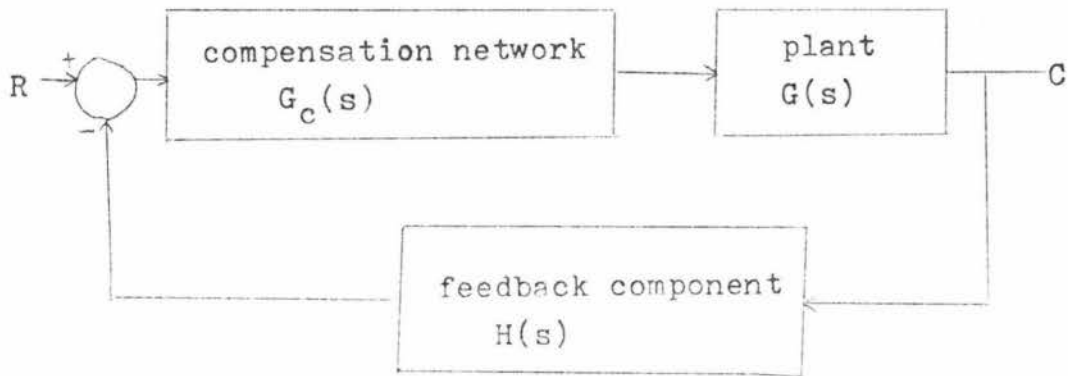


Fig. 2.1 Control system

In the following sections, different design techniques are described with most attention focussed on root-locus and frequency-response design techniques. The time-response technique, though it can be used for control system design with trial and error methods, is more useful for checking time-domain performance following the implementation of a design based upon frequency or root-locus techniques.

2.2 TIME-RESPONSE METHOD

2.2.1 *Time response analysis*

The most direct method of analysis is to consider the system performance in the time domain. The output response is derived for given inputs. The step function is the most useful form of input since control system performance is normally described in terms of step function response.

Direct solutions of the system differential equation following a step input, are plotted on an "output versus time" graph, with the condition that the system is at rest initially. This graph can be broken into two parts, steady-state terms which are directly related to the input, and the transient terms which are either exponential or oscillatory with an envelope of exponential form.

The required transient-response characteristics of a control system following a step input; are commonly specified in terms of time delay, rise time, overshoot and settling time. These time domain specifications are described in Appendix B.

A satisfactory system performance has:

- very small rise time, delay time and settling time.
- an overshoot not exceeding about 30 percent
- a subsidence ratio of about 3 : 1. The subsidence ratio is the ratio of the amplitudes of successive cycles in a decaying oscillation and no offset (or steady-state error).

Since overshoot and rise time conflict with each other, high accuracy and good stability are incompatible, and a good design is a compromise between the two. It is therefore desirable that the transient response must be sufficiently fast and sufficiently damped.

2.2.2 *Time response design*

The computer method employing the time response design technique is a trial and error approach. Given the transfer functions $G(s)$ and $H(s)$ of the fixed configurations of the system in Figure 2.1, a compensator $G_c(s)$ is inserted and the transient response of the overall control system to a unit-step input is then generated.

The resulting response provides information of the system performance, such as overshoot, rise time, delay time, etc.... This information permits the selection of $G_c(s)$ in a trial and error manner, i.e., the designer seeks to satisfy all the performance specifications by means of educated trial and error repetition.

Setting the gain is the first step in adjusting the system for satisfactory performance. In many practical cases, however, the adjustment of the gain alone may not provide sufficient alteration of the system behaviour. As is frequently the case, increasing the gain value will improve the steady-state behaviour but will result in poor stability or even instability. It is then necessary to redesign the system by incorporating additional compensators.

Since the effects of the basic control action, such as proportional, integral and derivative control are known¹, their combination such as proportional-plus-derivative (P.D), proportional-plus-integral (P.I) and proportional-plus-integral-plus-derivative (P.I.D) control actions may be employed to provide various modifications to the behaviour of the original system. The compensator transfer function $G_c(s)$ may take any of the following

forms:

$$G_c(s) = K_p \left(p + \frac{1}{T_I s} \right) \quad (\text{proportional-integral control})$$

$$G_c(s) = K_p (1 + T_d s) \quad (\text{proportional-derivative control})$$

$$G_c(s) = K_p \left(1 + \frac{1}{T_I s} + T_d s \right) \quad (\text{proportional-integral-derivative control})$$

where K_p represents the proportional gain

T_I represents the integral time

T_d represents the derivative time

K_p , T_I and T_d are adjustable.

The selection of $G_c(s)$ may be made as follows:

- If an offset is tolerable within limits, a proportional action may be used since it is simple and least expensive. (This is equivalent to the adjustment of the loop gain.)
- If no offset is tolerable, (P.I) action can be used, but at the expense of a more oscillatory behaviour.
- (P.D) control action increases the stability of the control system by introducing damping.
- (P.I.D.) control action stabilizes the control system and eliminates offset. It also permits increased sensitivity, and consequently increased speed of response.

Overshoot may be reduced by decreasing the value of K_p . This however results in a slower response. Overshoot can also be reduced by increasing T_I at a lesser expense to speed of response. However, overshoot is not lowered by increasing T_d . The speed of response is significantly increased by increasing the value of T_d .¹⁰

Hence, to decrease overshoot without seriously slowing the response, a combination of changes needs to be made. Similarly,

a better control system response with minimum oscillation, lower overshoot, low rise and response time and with no offset is devised, a series of possible combinations should be tried to reduce K_p slightly and to increase T_I and T_D moderately.

In addition to the above basic control actions in combination, other types of compensation, such as lead, lag and lead-lag compensation may be used to improve the control system behaviour. The particular situation determines the type of the compensation to be used.

- Lead compensation essentially yields an appreciable improvement in transient response and a small improvement in steady-state accuracy (with similar properties to derivative control). The transfer function of a lead compensator is

$$G_c(s) = \frac{s + \frac{1}{T}}{s + \frac{1}{\alpha T}} \quad (\text{Appendix C})$$

where $0 < \alpha < 1$, $\alpha = .1$ is a sensible practical value.

Value of T may be adjusted arbitrarily until a satisfactory system performance is reached.

- Lag compensation, on the other hand, yields an appreciable improvement in steady-state accuracy at the expense of increasing the transient response time (with similar properties to integral control).

The transfer function of a lag compensator is

$$G_c(s) = \frac{1}{\beta} \frac{(s + \frac{1}{T})}{s + \frac{1}{\beta T}} \quad (\text{See Appendix C})$$

Lead-lag compensation combines the characteristics of both lead compensation and lag compensation, i.e. lead-lag compensation yields both fast response, and good static accuracy.

The transfer function of a lead-lag network is:

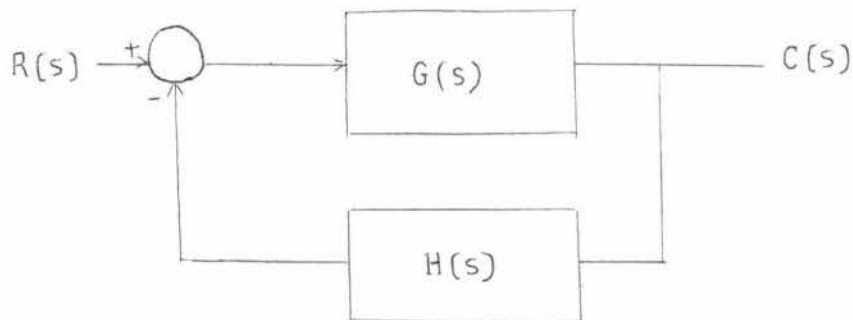
$$G_c(s) = \frac{1}{\beta} \left(\frac{s + \frac{1}{T_1}}{s + \frac{1}{\alpha T_1}} \right) \left(\frac{s + \frac{1}{T_2}}{s + \frac{1}{\beta T_2}} \right) \cdot K_c$$

If α is chosen to be $\frac{1}{\beta}$, then a lead-lag compensator is a combination of lead and lag compensators.

2.3 ROOT-LOCUS METHOD

2.3.1 Root-locus analysis

In the root-locus analysis problem, it is important to locate the closed-loop poles in the s-plane. In the design of closed-loop control system, it is desired to adjust the open-loop poles and zeros so as to place the closed-loop poles and zeros at particular locations in the s-plane. A simple method for finding the roots of the characteristic equation has been developed by W R. Evans and used extensively in control engineering². This method, called the root-locus method, is one in which the roots of the characteristic equations are plotted as the open loop gain constant is varied from zero to infinity. Consider the feedback control system given below:



The closed-loop transfer function of this system is

$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

Let the open loop T.F $G(s)$, $H(s)$ be represented by

$$G(s), H(s) = \frac{KA(s)}{B(s)} = \frac{K(a_0 s^m + a_1 s^{m-1} + \dots + a_{m-1} s + a_0)}{(b_0 s^n + b_1 s^{n-1} + \dots + b_n s + b_0)}$$

where $A(s)$ and $B(s)$ are the finite polynomials of the complex variable s , $m \leq n$ and K is the open-loop gain factor - the closed-loop transfer function then becomes

(overleaf)

$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + \frac{KA(s)}{B(s)}} + \frac{G(s), B(s)}{B(s) + KA(s)}$$

The closed-loop poles are the roots of the characteristic equation

$$B(s) + KA(s) = 0 \quad (2.1)$$

In general the location of these roots in the s-plane changes as the open-loop gain factor K is varied. A locus of these roots plotted in the s-plane as a function of K is called the root-locus.

For K equal to zero, the roots of (2.1) are the roots of the polynomial B(s), which are the same as the poles of the open-loop T, F G(s), H(s).

If K becomes very large, the roots approach those of the polynomial A(s), which are the open-loop zeros. Thus as K is increased from zero to infinity, the loci of the closed-loop poles originate from the open-loop poles and proceed toward and terminate at the open-loop zeros.

For this project, the roots of the characteristic equation are computed and plotted by a digital computer, the rules for constructing root loci from the open-loop roots are therefore not considered.

Stability

The system is unstable for all values of K for which any of the loci pass into the right-hand half of the s-plane, i.e., poles with positive real parts. The system is simple harmonic, in which case the output is a stable amplitude sinusoidal oscillation, for any value of K for which the closed-loop poles are on the imaginary

axis (i.e. poles with zero real parts),

Angle and magnitude condition

In order for a branch of a root-locus to pass through a particular point S_1 in the s -plane, it is necessary that s_1 be a root of the characteristic equation (2.1) for some real value of K .

$$\text{That is} \quad B(s) + KA(s) = 0$$

$$\text{or} \quad 1 + G(s),H(s) = 0$$

$$GH(s_1) = \frac{KA(s)}{B(s)} = -1 \quad (2.2)$$

therefore the complex number $GH(s)$ must have a phase angle of

$$\angle GH(S_1) = \pm 180(2K + 1) \quad (K = 0, 1, 2, \dots)$$

This is the angle condition.

In order for S_1 to be the root-locus, it is necessary that Fig (2.2) be satisfied with regard to magnitude in addition to phase angle.

$$\text{This is the magnitude condition} \quad |GH(s_1)| = 1$$

Dominant closed-loop poles

Those closed-loop poles which have dominant effects upon the transient-response behaviour are called dominant poles. They are the poles nearest to the $j\omega$ axis.

Damping ratio from the root-locus

The gain factor K required to give a specified damping ζ (or vice-versa) is easily determined from the root-locus. A line is drawn from the origin at an angle of plus or minus θ with the negative axis, where

$$\theta = \cos^{-1} \zeta$$

The gain factor at the point of intersection with the root-locus is the required value of K .

2.3.2. Root-locus design

Design problem

The root-locus method can be used very effectively in the design of control systems because it graphically illustrates the variation of the closed-loop poles as a function of the gain factor K . In its simplest form, the design is accomplished by choosing a value of K which results in a satisfactory closed-loop behaviour. If it is not possible to meet system specifications in this way, another form of compensation can be added to the system to alter the root-locus in the required manner.

The closed-loop system transient and frequency responses are determined by the location of the closed-loop poles. If the system has only two closed-loop poles and no finite zeros, performance parameters such as percentage overshoot, rise time, and 3dB bandwidth are directly related to the damping ratio ζ and undamped natural frequency ω_n of the poles. For higher order systems the relationship of these parameters to the closed-loop poles and zeros becomes more complicated and it is necessary to resort to standard curves or tables for this information. However, in many instances a higher order system may be approximated by a second or a third order system by using a dominant pole-zero approximation⁵. Specifications on allowable steady-state errors usually take the form of a minimum open-loop gain factor, expressed in terms of the error constants. (Appendix A)

Cancellation compensation

If the pole-zero configuration of the plant is such that the system specifications cannot be met by an adjustment of the open-loop gain factor, a cascade compensator, as shown in Fig.2.2 can be added to the system to cancel some or all the poles and zeros of the plant. When poles of the plant are cancelled by zeros of the compensator, the compensator also adds new poles to

the forward loop transfer function,

The philosophy of the compensation technique is then to replace undesirable poles. Once the effects on the root -locus of the addition of poles and/or zeros are fully understood, the locations of the pole(s) and zero(s) of the compensator can be readily determined in order to reshape the root-locus as desired.

Effects of the addition of poles

The addition of a pole to the open-loop transfer function has the effect of pulling the root-locus to the right, tending to lower the system's relative stability and slow down the settling of the response. Figure 2.3 shows examples of root-loci illustrating effects of the addition of a pole to a single pole system and the addition of two poles a single pole system.

Effects of the addition of zeros

The addition of a zero to the open-loop T.F has the effect of pulling the root-locus to the left, tending to make the system more stable and to speed up the settling of the response. Figure 2.4 shows the root-locus plots for the system when a zero is added to the open-loop transfer function.

The procedures for designing a lag-lead compensator may be stated as follows:

- (a) From the given performance specifications, determine the desired location for the dominant closed-loop poles.
- (b) In order to have the dominant closed-loop poles at the desired locations, calculate the angle contribution ϕ needed from the phase lead position of the lead-lag network,
- (c) Using the transfer function of the lead-lag compensator:

$$G_c(s) = \frac{s + \frac{1}{T_2}}{s + \frac{\beta}{T_1}} \cdot \frac{s + \frac{1}{T_1}}{s + \frac{1}{\beta T_2}} \cdot K_c$$

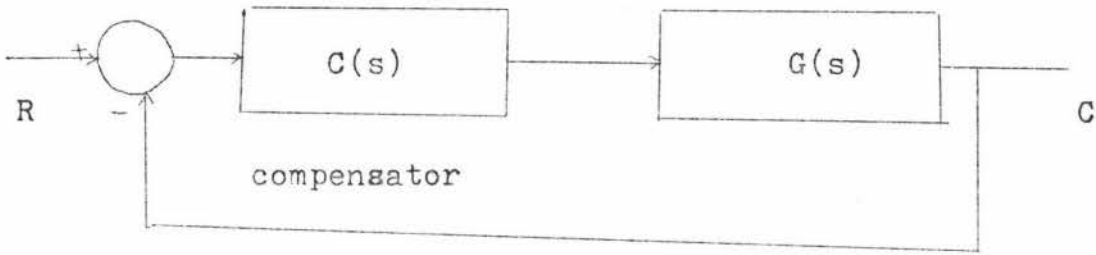


Fig. 2.2 Compensated system

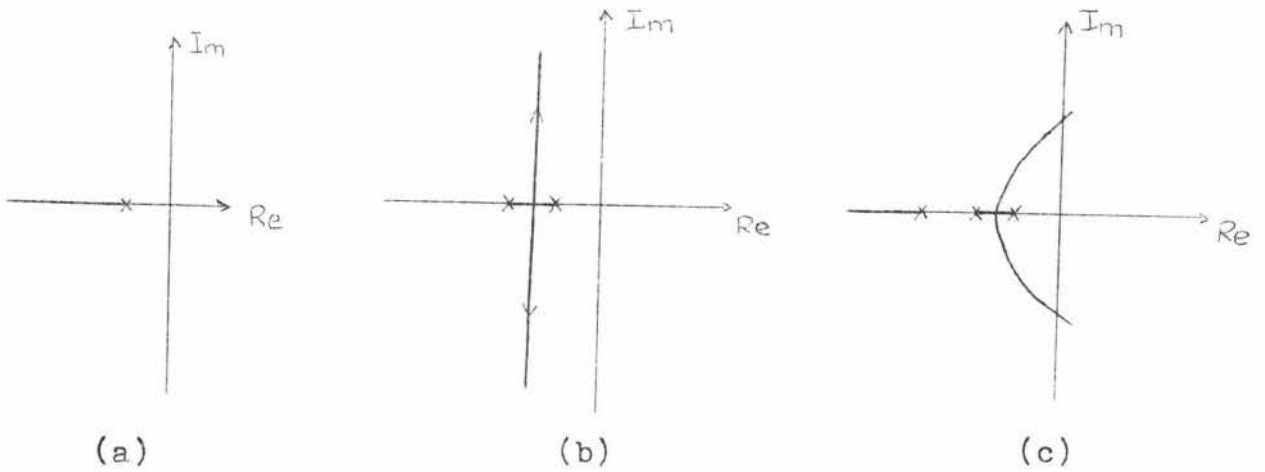


Fig. 2.3

- (a) Root-locus plot of a single-pole system
 (b) Root-locus plot of a two-pole system
 (c) Root-locus plot of a three-pole system

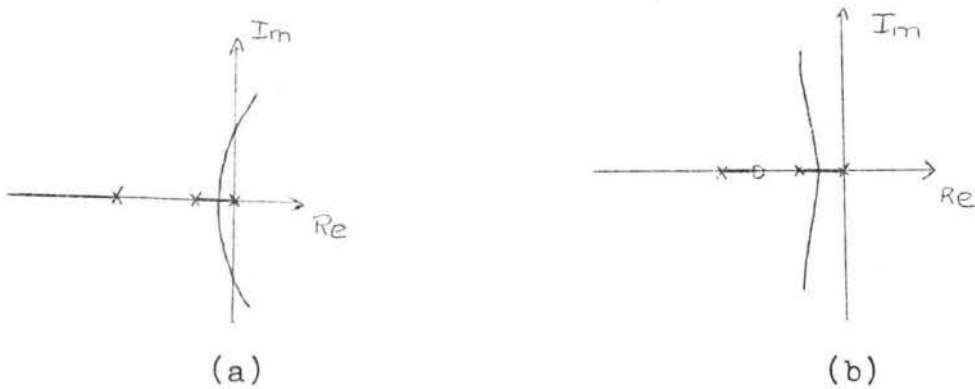


Fig. 2.4

- (a) Root-locus of a three-pole system
 (b) Root-locus showing the effect of addition of a zero to the system

determine the constant K_c from the requirement on the particular error coefficient specified in the design problem.

- (d) For the lead-lag compensator, we choose T_2 sufficiently large so that

$$\frac{s_1 + \frac{1}{T_2}}{s_1 + \frac{1}{\beta T_2}} \quad \text{is approximately unity, where}$$

$s = s_1$ in one of the dominant closed-loop poles.

Determine the values of T_1 and β from the requirements that

$$\left| \frac{s_1 + \frac{1}{T_1}}{s_1 + \frac{1}{\beta T_2}} \right| 1k_c G(s_1) = 1$$

and

$$\angle \frac{s_1 + \frac{1}{T_1}}{s_1 + \frac{1}{\beta T_2}} = \phi$$

- (e) Using value of β just determined, choose T_2 so that

$$\left| \frac{s_1 + \frac{1}{T_2}}{s_1 + \frac{1}{\beta T_2}} \right| \neq 1$$

and $0 < \frac{s_1 + \frac{1}{T_2}}{s_1 + \frac{1}{\beta T_2}} < 3^0$

The value of βT_2 , the largest time constant of the lag-lead network, should not be too large to be physically realized.

By choosing T_2 sufficiently large, Fig.(2.3) becomes

$$G_c(s) = \frac{s + \frac{1}{T_1}}{s + \frac{1}{\beta T_1}} K_c \quad (2.4)$$

which is the transfer function of a lead-compensation network.

Steps a,b,c, (with $G_c(s)$ from Fig.(2.4) and d may be followed to design a lead compensator.

Similarly, by choosing T_1 sufficiently large, Fig.(2.3) becomes

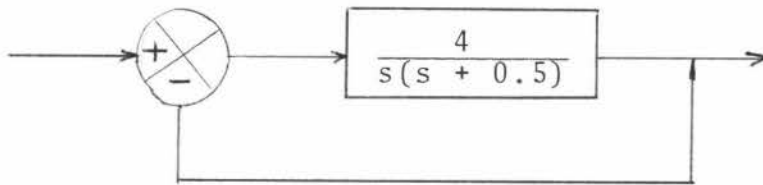
$$G_c(s) = \frac{s + 1/T_2}{s + 1/\beta T_2} K_c \quad (2.5)$$

which is the transfer function of a lag-compensation network.

To design a lag compensation network, choose β between 1 and 15 ($\beta = 10$ is a good choice) and follow steps a, c (with $G_c(s)$ from Fig.(2,5) and e.

Example: The following example illustrates the design of a lead-lag compensator, following the stated procedures.

Consider the control system shown below



The feed forward transfer function is

$$G(s) = \frac{4}{s(s + 0.5)}$$

It is desired to make the damping ratio equal to 0.5 and to increase the undamped natural frequency to 5 rad/sec and the static velocity error coefficient to 50sec^{-1} . A lead-lag compensator is designed to meet all the performance specifications.

From the performance specifications, the dominant closed-loop poles must be

$$s = -2.5 \pm j 4.33$$

since

$$\left. \frac{4}{s(s + 0.5)} \right|_{s = -2.5 + j 4.33} = -235^\circ$$

the phase lead portion of the lead-lag network must contribute 55° so that the root locus passes through the desired location

of the dominant closed-loop poles.

The compensated system will have the transfer function

$$G_c(s) G(s) = \frac{s + 1/T_1}{s + 1/T_1} \frac{s + 1/T_2}{s + 1/\beta T_2} K_c G(s)$$

The static velocity error coefficient therefore becomes

$$K_v = \lim_{s \rightarrow 0} s G_c(s) G(s) = \lim_{s \rightarrow 0} s K_c G(s)$$

The requirement on the static velocity error coefficient is

$$K_v = 50 \text{sec}^{-1}$$

Thus

$$\begin{aligned} K_v &= \lim_{s \rightarrow 0} \frac{s 4 K_c}{s(s + 0.5)} \\ &= 8 K_c = 50 \end{aligned}$$

hence $K_c = 6.25$ and compensated open-loop transfer function becomes

$$G_c(s) G(s) = \left(\frac{s + 1/T_1}{s + \beta/T_1} \right) \left(\frac{s + 1/T_2}{s + 1/\beta T_2} \right) \left(\frac{25}{s(s + 0.5)} \right)$$

since T_2 is chosen large enough that

$$\frac{s + \frac{1}{T_2}}{s + 1/\beta T_2} \approx 1$$

if it is required that the closed-loop poles lie at $s = -2.5 \pm j4.33$ the magnitude condition becomes

$$\begin{aligned} \left| G_c(s) G(s) \right|_{s = -2.5 + j4.33} &= \left| \frac{s + 1/T_1}{s + \beta/T_1} \right| \left| \frac{25}{s(s + 0.5)} \right|_{s = -2.5 + j4.33} \\ &= \left| \frac{s + \frac{1}{T_1}}{s + \beta/T_1} \right| \frac{5}{4.77} = 1 \end{aligned}$$

and the angle condition becomes

$$\angle \left. \frac{s + 1/T_1}{s + \beta/T_1} \right|_{s = -2.5 + j4.33} = 55^\circ$$

It is a simple matter to graphically determine the values of T_1 and β that satisfy these magnitude and angle conditions. Referring to Fig. (2.5), we can easily locate points A and B so that

$$\angle_{APB} = 55^\circ, \quad \frac{\overline{PA}}{\overline{PB}} = \frac{4.77}{5}$$

Graphically, from Fig. (2.5)

$$\overline{AO} = 0.5, \quad \overline{BO} = 5$$

Hence

$$-1/T_1 = -0.5, \quad -\beta/T_1 = -5$$

Thus $T_1 = 2, \beta = 10$

Therefore, the phase lead position of the lag-lead network becomes

$$\frac{s + 0.5}{s + 5}$$

For the phase lag position of the lag-lead network, it is required that

$$\left| \frac{s + 1/T_2}{s + 1/10T_2} \right|_{s = -2.5 + j4.33} \# 1$$

$$0 < \left| \frac{s + 1/T_2}{s + 1/10T_2} \right| < 3^\circ$$

$$s = -2.5 + j4.33$$

Choose $T_2 = 10$

Then the transfer function of the lead-lag compensator is

$$G_c(s) = \left(\frac{s + 0.5}{s + 5} \right) \left(\frac{s + 0.1}{s + 0.01} \right) \quad 6.25$$

And the compensated system will have the open-loop transfer function

$$G_c(s), G(s) = \frac{25 (s + 0.1)}{s(s + 5) (s + 0.01)}$$

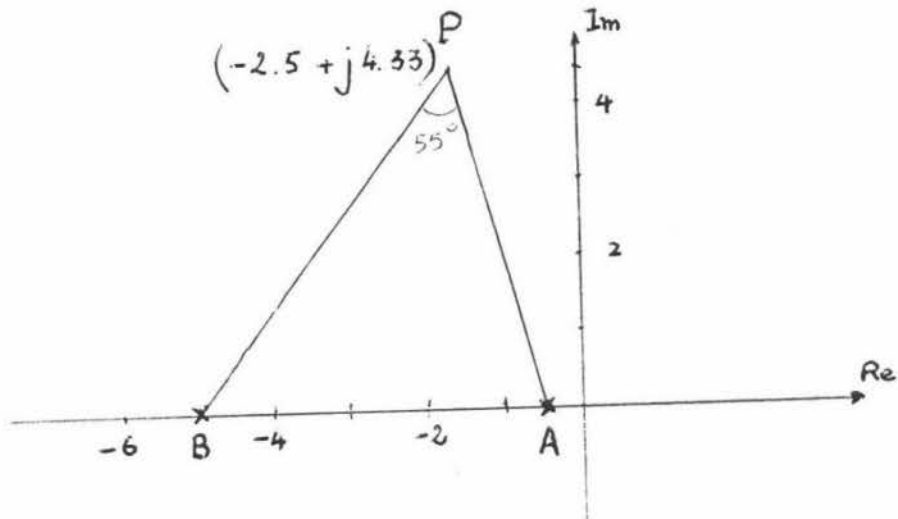
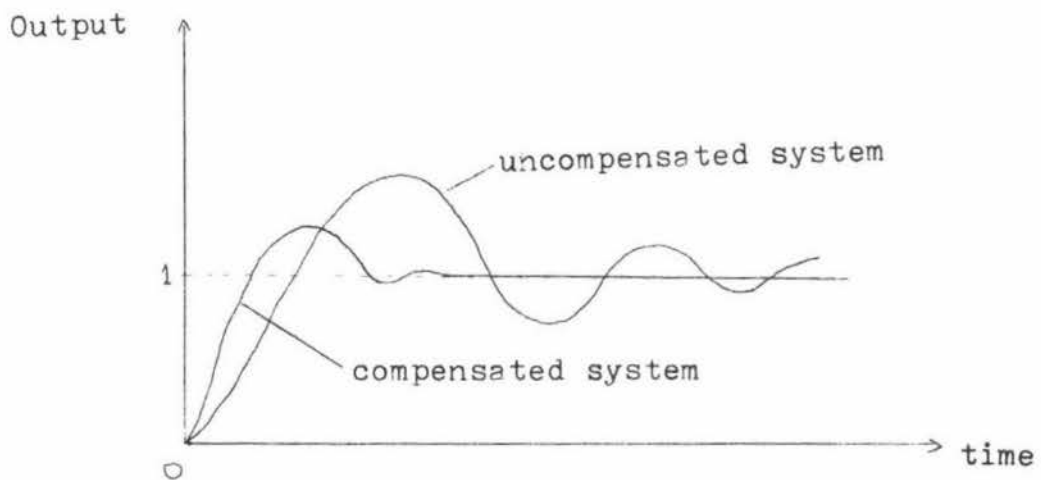


Fig. 2.5

Fig. 2.6 Unit step response of the uncompenstate
-d and compensated system

Because the angle contribution of the phase lag portion of the lead-lag network is quite small, there is no appreciable change in the location of the dominant closed-loop poles from the desired location, $s = 2.5j \pm j4.33$. Therefore, the compensated system meets all the required performance specifications.

Thus, given the unalterable plant transfer function $G(s)$, the following procedure is applied to choose the necessary compensating network $C(s)$:

1. Plot the root-locus of the given plant transfer function $G(s)$.
2. Examine the loci to see if it is necessary to "pull" it to the left, making the system more stable and speeding up the settling of the response. To achieve this, the dominant pole(s) which results in oscillation, is approximately cancelled by the compensating network zero(s), replacing it with insignificant pole. The effect is to distinctly alter the root locus, moving the oscillatory poles to the left.
3. If poles with small damping ratios are present in the plant transfer function, they may be cancelled and replaced with poles which have larger damping ratios.

Although this compensation technique is simple, the difficulty encountered in applying it is that it is not always apparent what open-loop pole-zero configuration is desirable from the standpoint of meeting specifications on the closed-loop system performance.

Lead-lag compensation

Lead compensation essentially yields an appreciable improvement in transient-response and a small improvement in steady-state accuracy.

Lead compensation on the other hand, yields an appreciable improvement in steady-state accuracy at the expense of increasing

the time response. Lead-lag compensation combines the characteristics of both lead and lag compensation.

The transfer function of the lead-lag network is

$$G_c(s) = \left(\frac{s + 1/T_1}{s + \beta/T_1} \right) \cdot \left(\frac{s + 1/T_2}{s + 1/\beta T_2} \right) K_c \quad (2.3)$$

2.4 FREQUENCY-RESPONSE METHOD

2.4.1 *Introduction*

The term "frequency-response", means the steady-state response of a system to a sinusoidal input. For frequency-response methods, the most conventional methods available to control engineers for analysis and design of control systems, the frequency of the input signal over a certain range is varied and the resulting frequency-response is studied. The results are plotted on graphs such as Nyquist diagrams, Bode plots and Nichols chart plots. A digital computer is of tremendous assistance in accomplishing the tedious task of plotting point by point, especially when the system transfer function is of high order.

Design in the frequency-domain is simple and straightforward. The frequency-response plot indicates clearly the manner in which the system should be modified in order to obtain the desired transient-response characteristics.

Bode diagrams are convenient to use if we desire a certain phase or gain margin. On the other hand, if a certain value of the resonant peak value M_r is required, the Nyquist or Nichols chart plot is more convenient to use. Usually Nyquist design techniques provide rapid evaluation of stability of the system and they are thus used to supplement other methods.

2.4.2 *Bode design*

Bode design techniques employ graphical representation of the open-loop transfer function $GH(j\omega)$, both plotted as a

function of frequency ω , Logarithmic scales are usually used for the frequency axes because the magnitude and phase angle may be graphed over a greater range of frequencies than with a linear frequency axis. The magnitude $G(j\omega)$ for any value of ω is plotted on a linear scale in decibel (db) unit, where

$$\text{db} = 20 \log_{20} G(j\omega)$$

The phase angle, in degrees, is plotted on a linear scale.

Fig. 2.7 shows the Bode representation of a second-order transfer function. From Bode diagrams, information such as gain and phase margins, bandwidth, cut-off frequency, resonant peak and resonant frequency may be obtained. These frequency-domain specifications are described in Appendix B.

Stability of the system

In most cases positive gain and phase margins obtained from the o/p-loop T-F will ensure stability of the closed-loop system. If the open-loop gain is increased by the gain margin the overall gain at 180° phase shift will be unity (0db), and the system is on the verge of instability.

Negative gain and phase margins indicate an unstable system.

For satisfactory performance, the phase margin should be between 30° and 60° , and the gain margin should be greater than 6db.

Bode design,

Design of a control system using Bode techniques entails shaping and reshaping the Bode magnitude and phase angle plots until the system specifications are satisfied. Setting the gain is the first step in shaping the Bode plot for better performance. In many practical cases, however, the adjustment of the gain alone may not provide sufficient alteration of the system behaviour. It is then necessary to shape the Bode plots by adding cascade or feedback compensation in order to alter the

overall behaviour so that the system will behave as desired.

Lead compensation

The lead compensator, presented in Appendix B, has the following Bode form of frequency response function.

$$G_c(j\omega) = \frac{j\omega + 1/T}{j\omega + 1/\alpha T} \quad (\alpha < 1)$$

The primary function of the lead compensator is to reshape the frequency-response curve to provide sufficient lead angle to offset the excessive phase lag associated with the components of the fixed system.

Assume that a unity feedback system is given. It is desired to satisfy the performance requirements, which are given in terms of phase margin, gain margin, error coefficients, etc. The procedures for designing a lead compensator may be stated as follows:

- (a) Determine the open-loop gain to satisfy the requirement on the error coefficients.
- (b) Using the gain K thus determined, plot the Bode diagram of the open-loop system and evaluate the phase margin of the uncompensated system.
- (c) Determine the necessary phase lead angle ϕ to be added to the system.
- (d) Determine the attenuation factor α by use of Eq.

$\sin\phi_m = \frac{1-\alpha}{1+\alpha}$ (Appendix C). Determine the frequency at which the magnitude of the uncompensated system is equal to $-20\log_{10}\left(\frac{1}{\alpha}\right)$ (Appendix C). Select this frequency as the new gain crossover frequency. This frequency corresponds to ω_m and the maximum phase shift ϕ occurs at this frequency.

(e) Determine the corner frequencies of the lead network from

$$\omega = 1/T ; \omega = 1/\alpha T$$

Finally, insert an amplifier with gain equal to $\frac{1}{\alpha}$, or increase the gain of the existing amplifier by a factor of $\frac{1}{\alpha}$.

A design example using lead compensation techniques based on the frequency-response approach is presented in Chapter 4.

Lag compensation

The lag compensation network has the following transfer function

$$G_c(j\omega) = \frac{1}{\beta} \frac{s + 1/T}{s + 1/\beta T} \quad (\beta > 1)$$

The primary function of a lag network is to provide attenuation in the high-frequency range in order to give a system sufficient phase margin. The procedures for designing a lag compensator by the frequency-response approach may be stated as follows:

(It is assumed that the system has unity feedback.)

- (a) Determine the open-loop gain such as the requirement on the particular error coefficient is satisfied.
- (b) Using the gain thus determined, draw Bode diagram of the uncompensated system and determine the phase and gain margins of the uncompensated system.
- (c) If the specifications on the phase and gain margins are not satisfied, then find the frequency point where the phase angle of the open-loop transfer function is equal to 180° plus the required phase margin. The required phase margin is the specified phase margin plus 5° to 12° . (The addition of 5° to 12° compensates for the phase lag of the lag network.) Choose this frequency as the new gain crossover frequency.
- (d) Choose the corner frequency $\omega = \frac{1}{T}$ corresponding to the zero of the lag network) one decade below the new gain crossover frequency.

(e) Determine the attenuation necessary to bring the magnitude curve down to 0db at the new gain crossover frequency.

Noting that this attenuation is $-20 \log_{10} \beta$, determine the value of β . Then the other corner frequency (corresponding to the pole of the lag network) is determined from $\omega = \frac{1}{\beta T}$.

Lead-lag compensation

If the value of α for the lead network must be equal to the reciprocal of the value of β for the lag network, a lead-lag compensator may be designed by combining the individually designed lead and lag compensators. Apart from lead, lag and lead-lag compensators, other conventional control action such as P.I, P.D or P.I.D control action may be employed to compensate for the deficiencies of the system performance.

2.4.4. Nyquist design

Nyquist analysis, a frequency response method, is essentially a graphical procedure for determining absolute and relative stability of closed-loop control systems. Information about stability is available directly from a graph of the sinusoidal open-loop transfer function $GH(j\omega)$.

The open-loop transfer function $GH(s)$ may be written in an equivalent form

$$GH(j\omega) = X(\omega) + jY(\omega)$$

A Nyquist plot of $GH(j\omega)$ is a graph of $Y(\omega)$ versus $X(\omega)$ in the finite position of the $G(j\omega)$ -plane for $-\alpha \leq \omega \leq +\alpha$

Fig. 2.9 shows the typical Nyquist diagram of

$$GH(j\omega) = \frac{\omega_1^2}{(j\omega)^2 + 2j\omega(j\omega) + \omega_n^2}$$

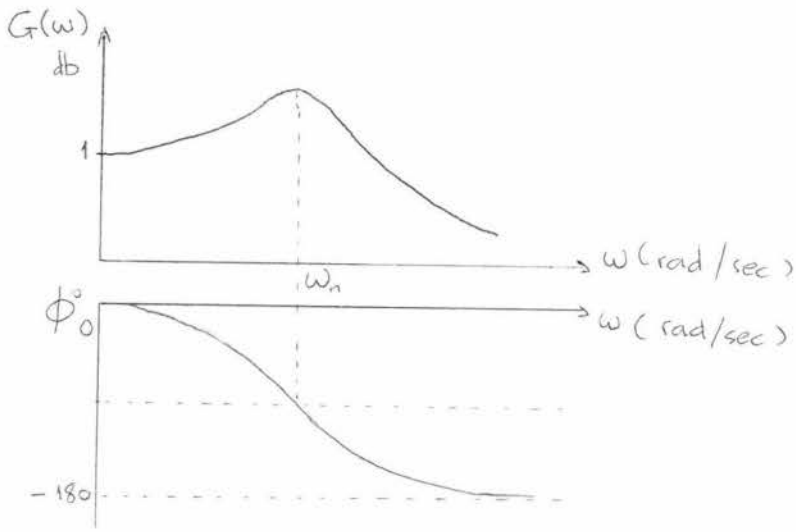
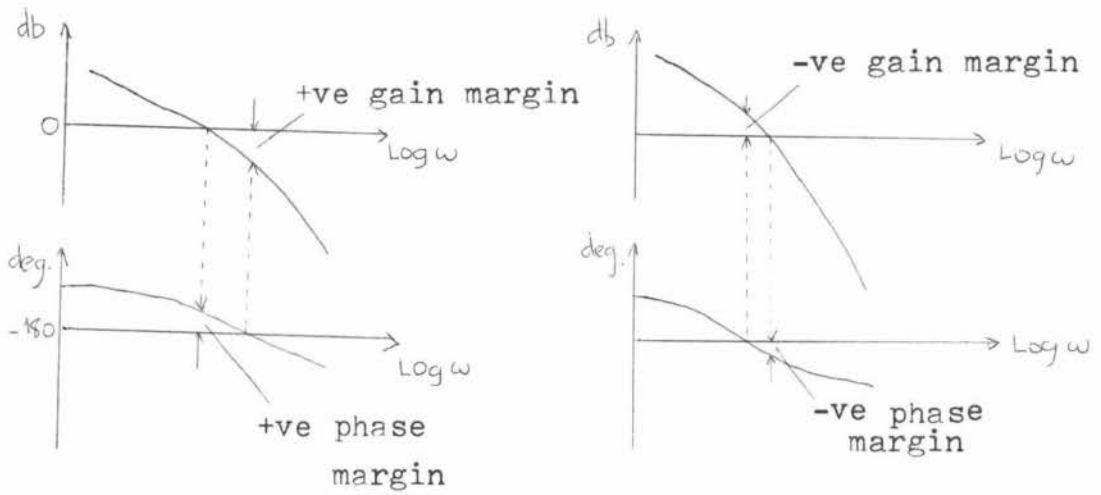


Fig. 2.7 Bode diagram of

$$\frac{\omega_n^2}{(j\omega)^2 + 2\zeta\omega_n(j\omega) + \omega_n^2}$$



(stable)

(unstable)

Fig. 2.8 phase and gain margins of stable and unstable system.

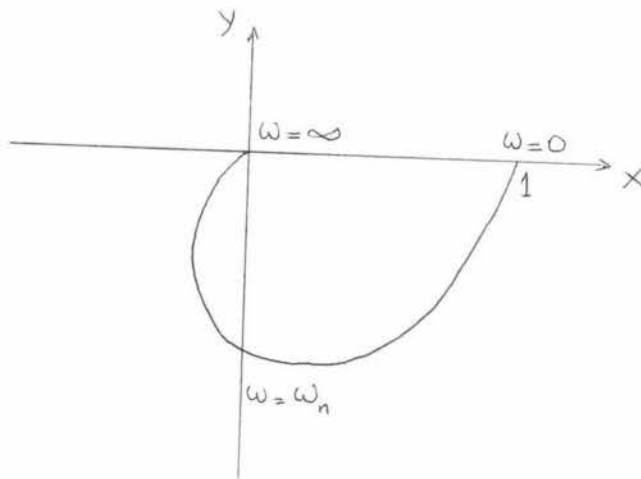


Fig. 2.9 Nyquist diagram of

$$\frac{\omega_n^2}{(j\omega)^2 + 2\zeta\omega_n(j\omega) + \omega_n^2}$$

Nyquist stability criterion

The simplified Nyquist stability criterion states that for a closed-loop system to be stable the locus of the open-loop frequency response, plotted on a Nyquist diagram must not enclose the point $(-1, j0)$ as ω varies from zero to infinity. Enclosing the point $(-1, 0)$ may be interpreted as passing to the left of the point.

Fig. 2.10 shows three systems with similar time constant but different gain constants. When the loop is closed, system (a) is stable, (b) is just stable and (c) is unstable. The complete Nyquist stability criterion states that for a closed-loop system to be stable it is necessary and sufficient that the contour of the open-loop frequency-response $GH(j\omega)$, describes a number of counterclockwise encirclements of the point $(-1, j0)$ as ω varies from $-\infty$ to $+\infty$, not less than the number of poles of $GH(j\omega)$ with positive real parts.

Let P_0 = number of poles with =ve real part of $GH(j\omega)$

N = total number of counterclockwise encirclements of the point $(-1, j0)$

$N > 0$ for counterclockwise encirclements

$N < 0$ for clockwise encirclements

For the closed-loop system to be stable $N \geq P_0$

A system with any clockwise net rotation is always unstable.

Nyquist design

Since it is not easy to generalize methods for the design of lead, lag or lead-lag compensator using Nyquist design technique, this method is normally used to provide rapid evaluation of system stability and to support other methods.

In the following, a method for determining the gain K so that the system will have a specific resonant peak M_p is demonstrated.

Referring to Fig. 2.11, the tangent line drawn from the origin to the desired M_p circle has an angle of ψ .

The value of $\sin \psi$ is

$$\sin \psi = \frac{\left| \frac{M_p}{M_p^2 - 1} \right|}{\left| \frac{M_p^2}{M_p^2 - 1} \right|} = \frac{1}{M_p^3}$$

The line drawn from point P , perpendicular to the negative real axis, intersects this axis at the $-1 + j0$ point. Consider the control system shown in Fig. 2.12. The procedure for determining the gain K so that $G(j\omega) = KG_1(j\omega)$ will have a desired value of M_p (where $M_p > 1$) can be summarized as follows:

1. Draw the Nyquist plot of the normalized open-loop transfer function $G_1(j\omega) = \frac{G(j\omega)}{K}$
2. Draw from the origin the line which makes an angle of $\psi = \sin^{-1}\left(\frac{1}{M_p}\right)$ with the negative real axis.
3. Fit a circle with centre on the negative real axis tangential to both the $G_1(j\omega)$ locus and the line PO .
4. Draw a perpendicular line to the negative real axis from point P , the point of tangency of this circle with the line PO . The perpendicular line PA intersects the negative real axis at point A .
5. In order that the circle just drawn corresponds to the desired M_p circle, point A should be the $-1+j0$ point.

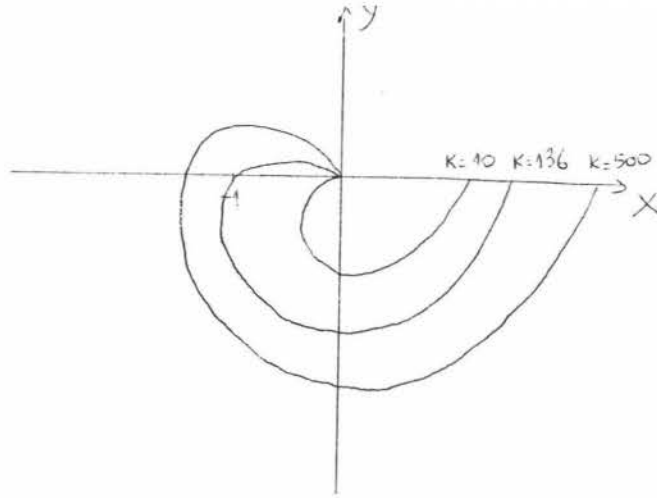


Fig. 2.10 Nyquist diagram of

$$\frac{K}{(1+2j\omega)(1+j\omega)(1+10j\omega)}$$

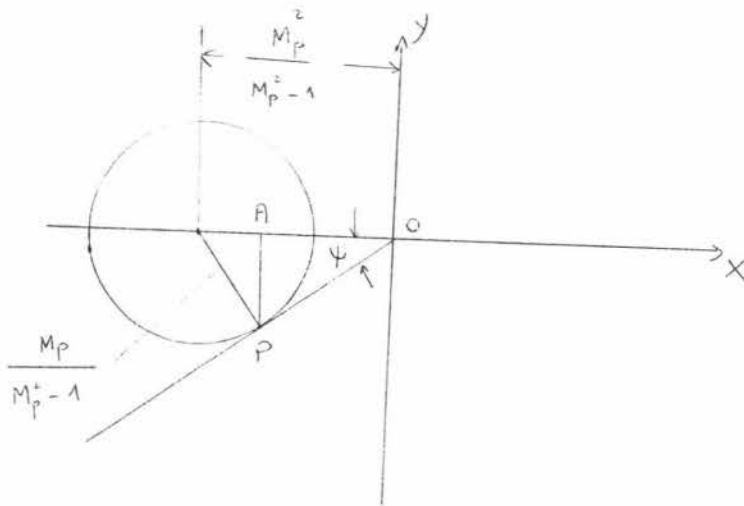


Fig. 2.11 M circle

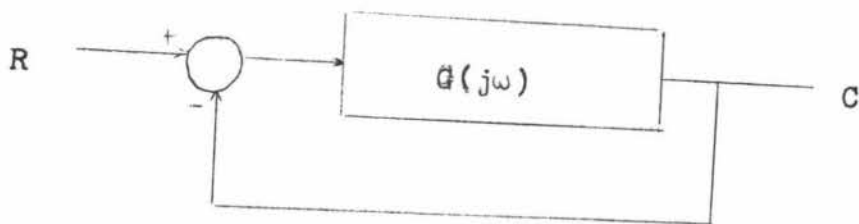


Fig. 2.12 Control system

6. The desired value of the gain K is that value which changes the scale so that point A becomes the $-1 + j\omega$ point.

Thus,

$$K_{M_p} = \frac{1}{OA}$$

Note that the resonant frequency ω_p is the frequency of the point at which the circle is tangent to the $G(j\omega)$ locus. The present procedure may not yield a satisfactory value for ω_p . If this is the case, the system must be compensated in order to increase the value of ω_p without changing the value of M_p .

2.4.5 Nichols chart design

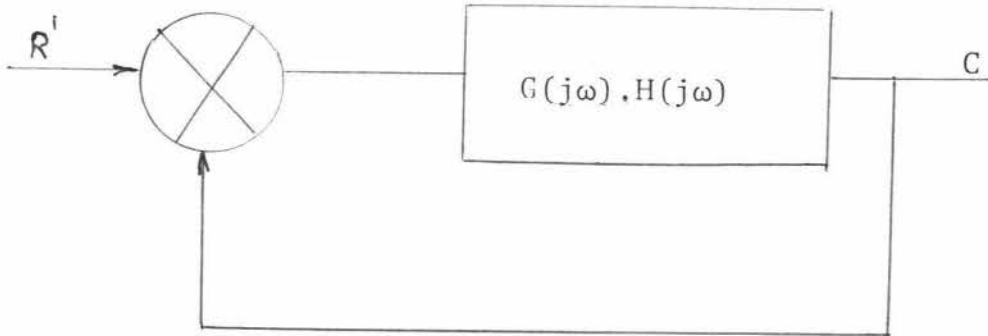
The closed-loop transfer function of the control system shown in Figure 2.13 is

$$\begin{aligned} \frac{C(j\omega)}{R(j\omega)} &= \frac{G(j\omega)}{1 + G(j\omega) \cdot H(j\omega)} \\ &= \frac{G(j\omega) \cdot H(j\omega)}{1 + G(j\omega) \cdot H(j\omega)} \cdot \frac{1}{H(j\omega)} \\ &= \frac{1}{H(j\omega)} \cdot \frac{C(j\omega)}{R'(j\omega)} \end{aligned}$$

with

$$\frac{C(j\omega)}{R'(j\omega)} = \frac{G(j\omega) \cdot H(j\omega)}{1 + G(j\omega) \cdot H(j\omega)}$$

The term $\frac{C(j\omega)}{R'(j\omega)}$ is the transfer function of a direct feedback system with a forward transfer function of $G(j\omega) \cdot H(j\omega)$ as shown below



Put $G(j\omega) = G_2(j\omega) = G(\cos \phi + j \sin \phi)$

Where G and ϕ are function of ω

Hence the closed-loop transfer function of the direct feedback system is

$$\begin{aligned} \frac{C(j\omega)}{R'(j\omega)} &= \frac{G_2(j\omega)}{1 + G_2(j\omega)} \\ &= \frac{G(\cos \phi + j \sin \phi)}{1 + G(\cos \phi + j \sin \phi)} \end{aligned}$$

Then the closed-loop amplitude of the direct feedback system is

$$\begin{aligned} M &= \frac{G}{\sqrt{(1 + G \cos \phi)^2 + G^2 \sin^2 \phi}} \\ &= \frac{G}{\sqrt{1 + 2G \cos \phi + G^2}} \end{aligned}$$

$$\begin{aligned} \therefore M^2 (1 + 2G \cos \phi + G^2) &= G^2 \\ \therefore \cos \phi &= \frac{-(M^2 + G^2)(M^2 - 1)}{2M^2 G} \end{aligned}$$

Similarly, the closed-loop phase angle of the direct feedback is

$$\delta = \angle G_c(j\omega) - 1 + \angle G_c(j\omega)$$

$$\delta = \phi - \text{TAN}^{-1} \frac{G \sin \phi}{1 + G \cos \phi} \quad 2.7$$

Equation 2.6 relates M to $\phi(\omega)$ and $G(\omega)$ so that contours for constant M can be plotted as $20 \log_{10} G(\omega)$ against $\phi(\omega)$.

Similarly Eq. 2.7 relates δ with $20 \log_{10} G(\omega)$ and $\phi(\omega)$ so that the constant δ contours can also be drawn.

These constant gain and phase contours are plotted on the db magnitude-degree phase diagram called the Nichols chart.

The magnitude-phase angle of the open-loop transfer function $G(j\omega) H(j\omega)$ plotted on a Nichols chart is called a Nichols chart plot of $GH(j\omega)$.

The Nichols chart applies only to direct feedback loops. The chart can be used to find $\frac{C(j\omega)}{R(j\omega)}$ and multiplying by $\frac{1}{H(j\omega)}$, as illustrated below:

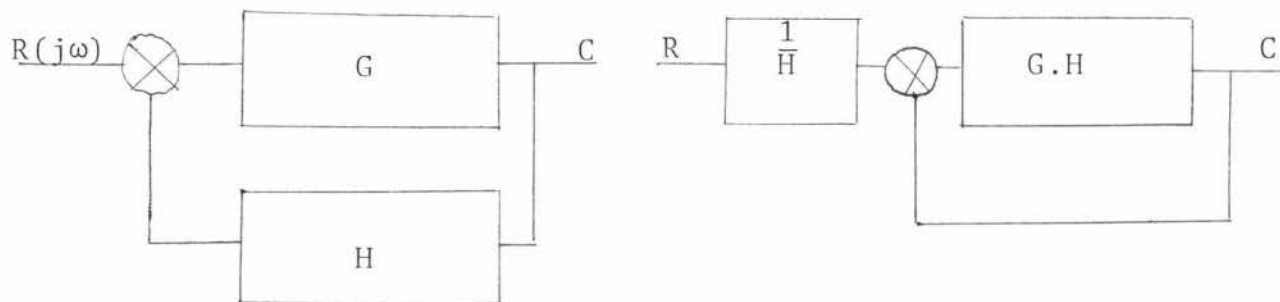


Fig. 2.14 shows the plot of $GH(j\omega)$ superimposed on a Nichols chart.

The Nichols chart is symmetric about the -180° axis. The magnitude-phase angle of the closed-loop transfer function of the direct feedback system $\frac{C}{R}(j\omega) = \frac{GH(j\omega)}{1 + GH(j\omega)}$ is obtained directly from the chart as the plot of the points where the graph of $GH(j\omega)$ intersects the graphs of loci of constant $\left| \frac{C}{R}(j\omega) \right|$ and $\arg \frac{GH(j\omega)}{1 + GH(j\omega)}$.

If the open-loop locus does not intersect the $M = M_r$ locus but is tangent to it, then the resonant peak value of M of the closed-loop frequency response is given by M_r . The resonant frequency is given by the frequency at the point of tangency. (Note that the closed-loop plot $\frac{C}{R}, (j\omega)$ can be directly generated by the computer

The stability of the system based on the Nichols chart plot is the same as the Bode diagram, i.e.: positive gain and phase margins yield a stable system. Fig. 2.15 shows the gain and phase margins of a stable and an unstable system.

Nichols chart design

Design by analysis in the frequency domain using Nichols chart techniques is performed in the same general manner as the design methods described in previous sections. Appropriate compensation networks are introduced in the forward and/or feedback paths and the behaviour of the resulting system is critically analysed. In this manner, the Nichols chart plot is shaped and reshaped until the performance specifications are met.

Gain-factor compensation using constant amplitude curves

The Nichols chart may be used to determine K_β (for a unity feedback system) for a specified resonant peak M_p (db). The following procedure is applied:

- (a) Draw only constant amplitude M_p curve and the magnitude-phase angle plot of $G(j\omega)$ for $K_\beta = 1$ on the Nichols chart.
- (b) Trace magnitude-phase angle plot of $G(j\omega)$ on tracing paper.
- (c) Slide the plot up or down until it is just tangent to the constant amplitude curve of M_p db. The amount of shift in db is the required value of K_β .

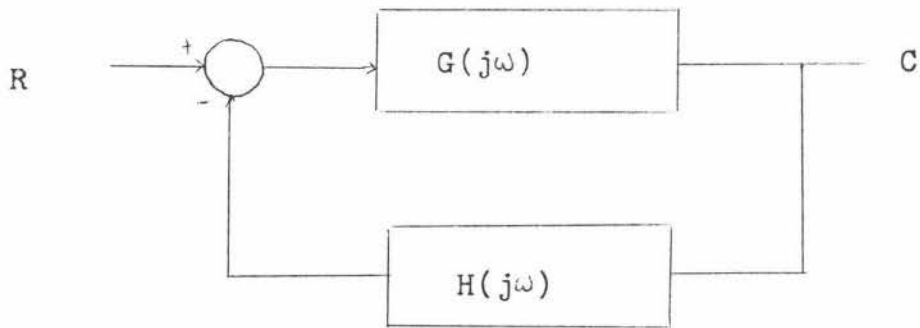


Fig. 2.13 Control system

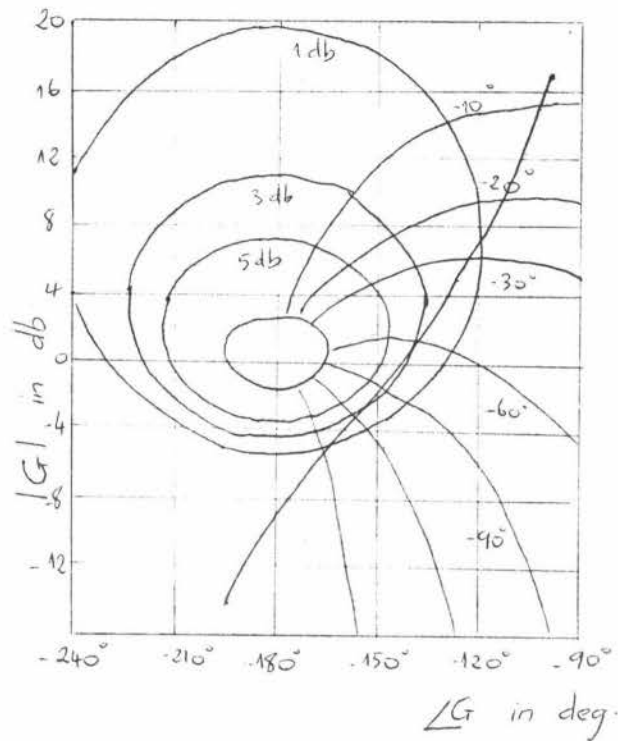
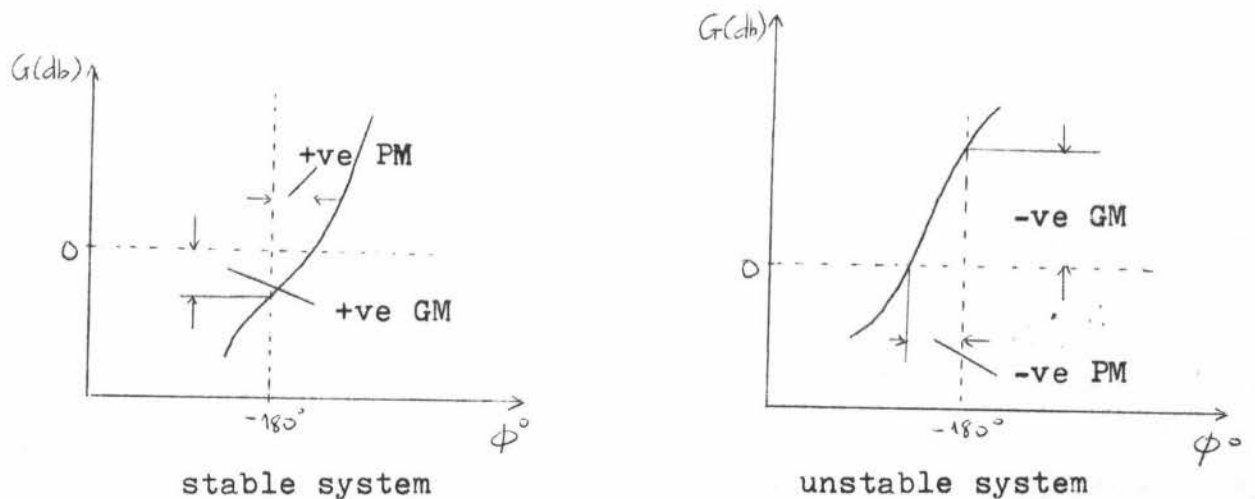
Fig. 2.14 Plot of $G(j\omega) \cdot H(j\omega)$ superimposed on a Nichols chart

Fig. 2.15 Phase and gain margins of stable and unstable system

Lead compensation

The transfer function for a lead network is $G_C(s) = \frac{j\omega + \frac{1}{T}}{j\omega + \frac{1}{\alpha T}}$
 where $\alpha < 1$.

For some systems in which lead compensation in the forward loop is applicable, appropriate choice of a and b permits an increase in K_β , providing greater accuracy and less sensitivity, without adversely affecting transient performance. The important properties of a lead network compensator are its phase-lead contribution in the low to medium frequency range (the vicinity of resonant frequency ω_p) and its negligible attenuation of high frequencies. If a very large phase-lead is required, several lead networks may be cascaded. Lead compensation generally increases the bandwidth of a system. The use of Nichols chart for selection of phase lead network is particularly suited to problems where the specification is given as a peak M_{\max} (db) occurring at a resonant frequency ω_R . Consider the uncompensated system $G(j\omega)$ of a direct feedback control system. The procedure for designing a lead compensator using Nichols chart techniques is as follows:

- From the Nichols chart, find the value of phase shift, ϕ , corresponding to the peak of the specified M_{\max} contour.
- Find the value of phase shift for the uncompensated system when $\omega = \omega_R$ (resonant frequency), say ϕ_R lag.
- The required value of phase lead then is

$$\phi_C = \phi_R - \phi \quad \text{lead}$$

If compensation is required, $\phi_R > \phi$.

(d) Determine the attenuation factor α by use of Fig.

$$\sin \phi_M = \frac{1 - \alpha}{1 + \alpha} \quad (\text{Appendix C})$$

Determine the frequency where the magnitude of the

uncompensated system is equal to $-20 \log_{10} \left(\frac{1}{\sqrt{\alpha}} \right)$ (Appendix C) Select this frequency as the new gain crossover frequency. This frequency corresponds to ω_m and the maximum phase shift ϕ_m occurs at this frequency.

(e) Determine the corner frequencies of the lead network from

$$\omega = \frac{1}{T}, \quad \omega = \frac{1}{\alpha T}$$

(f) It now remains to find K_c in cascade with the lead network to yield the control system with the required resonant peak M_{\max} . Plot $K_c G(j\omega)$ on the Nichols chart with $K_c = 1$. It can now be seen by how much the curve must be moved vertically to make the open-loop curve tangential to the M_{\max} contour. The vertical movement is $20 \log_{10} K_c \text{ db}$, hence K_c .

Lag compensation

The lag compensator has the following transfer function:

$$G_c(j\omega) = \frac{1}{\beta} \frac{s + 1/T}{s + 1/\beta T} \quad (\beta > 1)$$

The lag network provides compensation by attenuating the high frequency position of the db magnitude-phase angle plot.

Several general effects of lag compensation are:

- The bandwidth of the system is unusually decreased
- The predominant time constant τ of the system is unusually increased, producing a more sluggish system,
- For a given constant, relative stability is improved.

The procedure for using lag compensation in Nichols chart

techniques is essentially the same as that for lag compensation in Bode design technique.

Lead-lag compensation

The design of a lead-lag network compensator is based on the combination of the design techniques discussed under lead compensation and lag compensation.

2.5 DISCUSSION

This chapter has described various graphical techniques used in control system analysis and design, namely the time response, root-locus and frequency-response methods.

- The root-locus method is most convenient to use when specifications are given in terms of time domain quantities, such as overshoot, rise time, etc.
- The Bode design technique is the most popular of the frequency response methods if specifications are given in terms of frequency domain quantities, such as gain and phase margins.
- The Nichols chart design technique is used when a certain value of the resonant peak M_r is required.
- The Nyquist design technique provides rapid evaluation of stability of the control system and it is thus used to supplement other methods.

The time response method, though it can be used for control system design with trial and error approaches, is more useful for checking time domain performance following the implementation of a design based upon frequency or root-locus techniques.

However, whatever design technique is used, it is important to note that no single compensation scheme is universally applicable. The particular situation determines the type of compensation that is appropriate. Chapter 3 describes the computer software that has

been written to aid the design procedures described in this chapter. This software is able, given the generalized control system structure of Fig. 2.1, to generate, for either the open- or closed-loop form, any of the specified design charts. The software operates interactively to give the designer wide flexibility of choice in terms of the system structure to be analysed and the technique to be applied. It is planned that the facility will be enhanced to incorporate automatic design procedures for compensation networks.

-ooOoo-

CHAPTER THREE

PROGRAMME DESCRIPTION

3.1 INTRODUCTION, OUTLINE OF PROGRAMME

The software for computer aided control system design has been programmed in Algol and operates on the Massey University B6700 computer system. This chapter outlines the structure of the programme, detailing the methods employed to implement some of the concepts discussed in Chapter 2, and also specifies how the system is used.

The program is operated via a Tektronix 4010 visual display terminal. It is designed to function in an online interactive mode. The control system designer, at the Tektronix console, is able to specify certain designs and evaluate their effectiveness in terms of the various graphical analyses produced by the computer and displayed on the screen of the Tektronix. The designer can then adjust the designs in order to improve performance. Given a particular control system structure, which may be either open-loop or closed-loop, the designer may request the following options:

1. A display of the system time response following a step input.
2. A Bode diagram display of the system frequency response.
3. A Nyquist diagram display of the system frequency response.
4. A Nichols chart display of the system frequency response.
5. A root-locus plot of the roots of the system characteristic equation as the system gain varies.

The designer is able to select the scales and ranges of these displays so that any particular portion of a graph can be isolated and magnified if required.

3.2 EVALUATION OF SYSTEM TRANSFER FUNCTION

The control system is defined with the general structure illustrated in Fig. 3.1. The three transfer functions of Fig. 3.1 are given as the polynomial ratios.

$$G(s) = \frac{G_1(s)}{G_2(s)} = \text{the process transfer function} \quad (3.1)$$

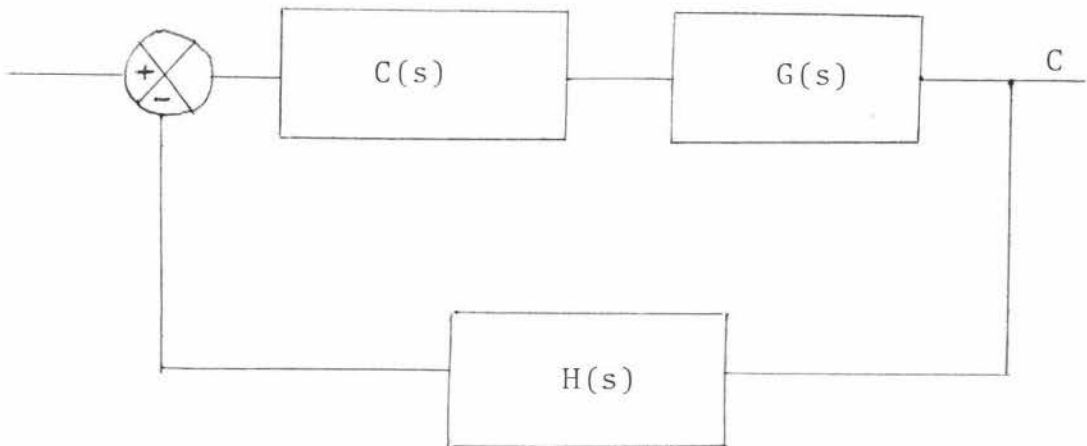


Figure 3.1 Control system

$$H(s) = \frac{H_1(s)}{H_2(s)} = \text{the measurement transfer function} \quad (3.2)$$

and
$$C(s) = \frac{C_1(s)}{C_2(s)} = \text{the controller transfer function} \quad (3.3)$$

As mentioned above, the control designer is able to select both open-loop and closed-loop analyses. To achieve this end the program has been designed to be flexible so that the following combinations of the above transfer functions can be studied by request, at the console:

i. open-loop structure

$$G(s), K \cdot C(s) \cdot G(s) \text{ and } K \cdot C(s) \cdot G(s) \cdot H(s) \quad (3.4)$$

ii. the full closed-loop structure of Fig. 3.1 for which the transfer function may be devised as

$$\frac{K C_1(s) \cdot G_1(s) \cdot H_2(s)}{C_2(s) \cdot G_2(s) \cdot H_2(s) + K C_1(s) \cdot G_1(s) \cdot H_1(s)} \quad (3.5)$$

subject to the option chosen, the software is able to manipulate the constituent polynomials to obtain a final transfer function of the form

$$R(s) = \frac{A(s)}{B(s)} \quad (3.6)$$

which is that analysed.

3.3 FREQUENCY RESPONSE ANALYSIS

Frequency response analysis required evaluation of the following frequency response function

$$R(j\omega) = \frac{A(j\omega)}{B(j\omega)} \quad (3.7)$$

It is necessary that polynomials $A(j\omega)$ and $B(j\omega)$ and divided into their real and imaginary parts, with

$$A(j\omega) = RA(\omega) + j IA(\omega) \quad (3.8)$$

and

$$B(j\omega) = RB(\omega) + j IB(\omega) \quad (3.9)$$

A procedure has therefore been developed to establish the real and imaginary parts of a general polynomial, of degree N , given the coefficients $A_0, A_1, A_2, \dots, A_n$. i.e.,

$$A(j\omega) = A_0(j\omega)^N + A_1(j\omega)^{N-1} + \dots + A_{N-1}(j\omega) + A_N \quad (3.10)$$

$$= A_N + j\omega A_{N-1} - \omega^2 A_{N-2} - j\omega^3 A_{N-3} + \omega^4 A_{N-4} \\ + j\omega^5 A_{N-5} - \omega^6 A_{N-6}$$

$$= (A_N + \omega^4 A_{N-4} + \omega^8 A_{N-8} + \dots)$$

$$- (\omega^2 A_{N-2} + \omega^6 A_{N-6} + \dots) \quad (3.11)$$

$$+ j(\omega A_{N-1} + \omega^5 A_{N-5} + \dots)$$

$$- j(\omega^3 A_{N-3} + \omega^7 A_{N-7} + \dots)$$

Therefore the real and imaginary components are derived as

$$RA(\omega) = (A_N + \omega^4 A_{N-4} + \omega^8 A_{N-8} + \dots) - (\omega^2 A_{N-2} + \omega^6 A_{N-6} + \dots) \quad (3.12)$$

and

$$IA(\omega) = (\omega A_{N-1} + \omega^5 A_{N-5} + \dots) - (\omega^3 A_{N-3} + \omega^7 A_{N-7} + \dots) \quad (3.13)$$

Terms in the above equations are included only if the indices for A are not negative,

$$\text{i.e. } A_{-1} = A_{-2} = \dots = 0$$

A procedure has been developed to derive the real and imaginary components by evaluating the above expressions for RA and IA. Difficulties were encountered in this evaluation because of particular properties of the supplied polynomial arithmetic procedures. This necessitated the development of a slightly less straightforward method of derivation than that indicated above. For the Bode display, the gain is plotted as

$$G = 20 \log_{10} |R(j\omega)| = 20 \log_{10} \sqrt{\frac{RA^2(\omega) + JA^2(\omega)}{RB^2(\omega) + IB^2(\omega)}} \quad \text{db} \quad (3.14)$$

and the phase is

$$P = \text{TAN}^{-1} \left(\frac{IA(\omega)}{RA(\omega)} \right) - \text{TAN}^{-1} \left(\frac{IB(\omega)}{RB(\omega)} \right) \quad \text{degrees} \quad (3.15)$$

The above expressions are also used for derivation of the Nichols chart display.

$R(j\omega)$ can also be written as

$$R(j\omega) = X(\omega) + jY(\omega)$$

with

$$X(\omega) = \frac{RA(\omega) \cdot RB(\omega) + IA(\omega) \cdot IB(\omega)}{RB^2(\omega) + IB^2(\omega)} \quad (3.16)$$

and

$$Y(\omega) = \frac{IA(\omega) \cdot RB(\omega) - RA(\omega) \cdot IB(\omega)}{RB^2(\omega) + IB^2(\omega)} \quad (3.17)$$

The Nyquist display is a plot of $Y(\omega)$ vs $X(\omega)$ as ω varies.

Constant magnitude and phase contours may be drawn on the Nichols display. These are derived using the expressions:

$$\cos \phi = - \frac{(M^2 + R^2(M^2 - 1))}{2M^2R} \quad (3.18)$$

and

$$\delta = \phi - \text{TAN}^{-1} \frac{R \sin \phi}{1 + R \cos \phi} \quad (3.19)$$

R and ϕ , functions of ω , are the magnitude and phase angle of the open loop transfer function of a direct feedback control system ($H(s) = 1$). M and δ are the magnitude and phase angle of the closed-loop transfer function of a direct feedback system. The M and δ contours are plotted on the Nichols display to generate the standard configuration.

3.4 ROOT-LOCUS ANALYSIS

A root-locus display is established by plotting the roots of the characteristic equation extracted from expression 3.5, namely:

$$C_2(s).G_2(s).H_2(s) + K.C_1(s).G_1(s).H_1(s) = 0$$

as K varies. Open-loop poles and zeros are established from the roots of

$$C_2(s).G_2(s).H_2(s) = 0$$

and

$$C_1(s).G_1(s).H_1(s) = 0$$

respectively.

3.5 TIME DOMAIN ANALYSIS

The time-response of the system to a step input is obtained by directly solving differential equations representing the system.

Consider the final transfer function described in equation 3.2

$$\text{i.e. } R(s) = \frac{A(s)}{B(s)}$$

$$\text{or } \frac{Y(s)}{U(s)} = \frac{A_0 s^M + A_1 s^{M-1} + \dots + A_M}{B_0 s^N + B_1 s^{N-1} + \dots + B_N} \quad (3.20)$$

where $N-1 \geq M$

Equation 3.20 may be written

$$\frac{Y(s)}{U(s)} = \frac{AA_0 s^M + AA_1 s^{M-1} + \dots + AA_M}{s^N + BB_1 s^{N-1} + \dots + BB_N} \quad (3.21)$$

where

$$AA_0 = \frac{A_0}{B_0} ; AA_1 = \frac{A_1}{B_0} ; \dots AA_M = \frac{A_M}{B_0}$$

and

$$BB_1 = \frac{B_1}{B_0} ; BB_2 = \frac{B_2}{B_0} ; \dots BB_N = \frac{B_N}{B_0}$$

From equation 3.1 we obtain

$$Y(s^N + BB_1 s^{N-1} + \dots + BB_N) = U \left(\frac{AA_0}{s^{N-M}} + \frac{AA_1}{s^{N-(M-1)}} + \dots + \frac{AA_M}{s^N} \right)$$

or

$$Y = U \left(\frac{AA_0}{s^{N-M}} + \frac{AA_1}{s^{N-(M-1)}} + \dots + \frac{AA_M}{s^N} \right) - Y \left(\frac{BB_1}{s} + \frac{BB_2}{s^2} + \dots + \frac{BB_N}{s^N} \right)$$

(3.22)

By introducing additional variables such as the rate of change of output, the same system can be represented by N first-order equations involving N system variables.

The state variable diagram for Eq. 3.22 is shown in Fig. 3.2

From Fig. 3.2 the following system of first-order equations may be written:

$$\begin{aligned}
 Z_1 &= -BB_1 \cdot Z_1 + Z_2 \\
 Z_2 &= -BB_2 \cdot Z_1 + Z_3 \\
 Z_3 &= -BB_3 \cdot Z_1 + Z_4 \\
 &\vdots \\
 &\vdots \\
 Z_{N-M} &= -BB_{N-M} \cdot Z_1 + Z_{N-M+1} + AA_0 U \\
 Z_{N-M+1} &= -BB_{N-M+1} \cdot Z_1 + Z_{N-M+2} + AA_1 U \\
 &\vdots \\
 &\vdots \\
 Z_{N-1} &= -BB_{N-1} \cdot Z_1 + Z_N + AA_{M-1} U \\
 Z_N &= -BB_N \cdot Z_1 + AA_M U
 \end{aligned} \tag{3.23}$$

Expressing in matrix form

$$\begin{bmatrix} Z_0 \\ Z_1 \\ Z_2 \\ \vdots \\ \vdots \\ Z_{N-M} \\ \vdots \\ \vdots \\ Z_{N-1} \\ Z_N \end{bmatrix} = \begin{bmatrix} 0 & 0 & 0 & \dots & \dots & \dots & 0 \\ 0 & -BB_1 & 1 & \dots & \dots & \dots & 0 \\ 0 & -BB_2 & 0 & \dots & \dots & \dots & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ AA_0 & -BB_{N-M} & \vdots & \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ AA_{M-1} & -BB_{N-1} & \vdots & \vdots & \vdots & 1 & \vdots \\ AA_M & -BB_N & 0 & \dots & \dots & \dots & 0 \end{bmatrix} \begin{bmatrix} U \\ Z_1 \\ \vdots \\ \vdots \\ \vdots \\ Z_{N-M} \\ \vdots \\ \vdots \\ Z_{N-1} \\ Z_N \end{bmatrix}$$

$$Z_0 = 0$$

$$Y = Z_1$$

(3.24)

Equations 3.24 are solved numerically using a standard differential equation solving subroutine.

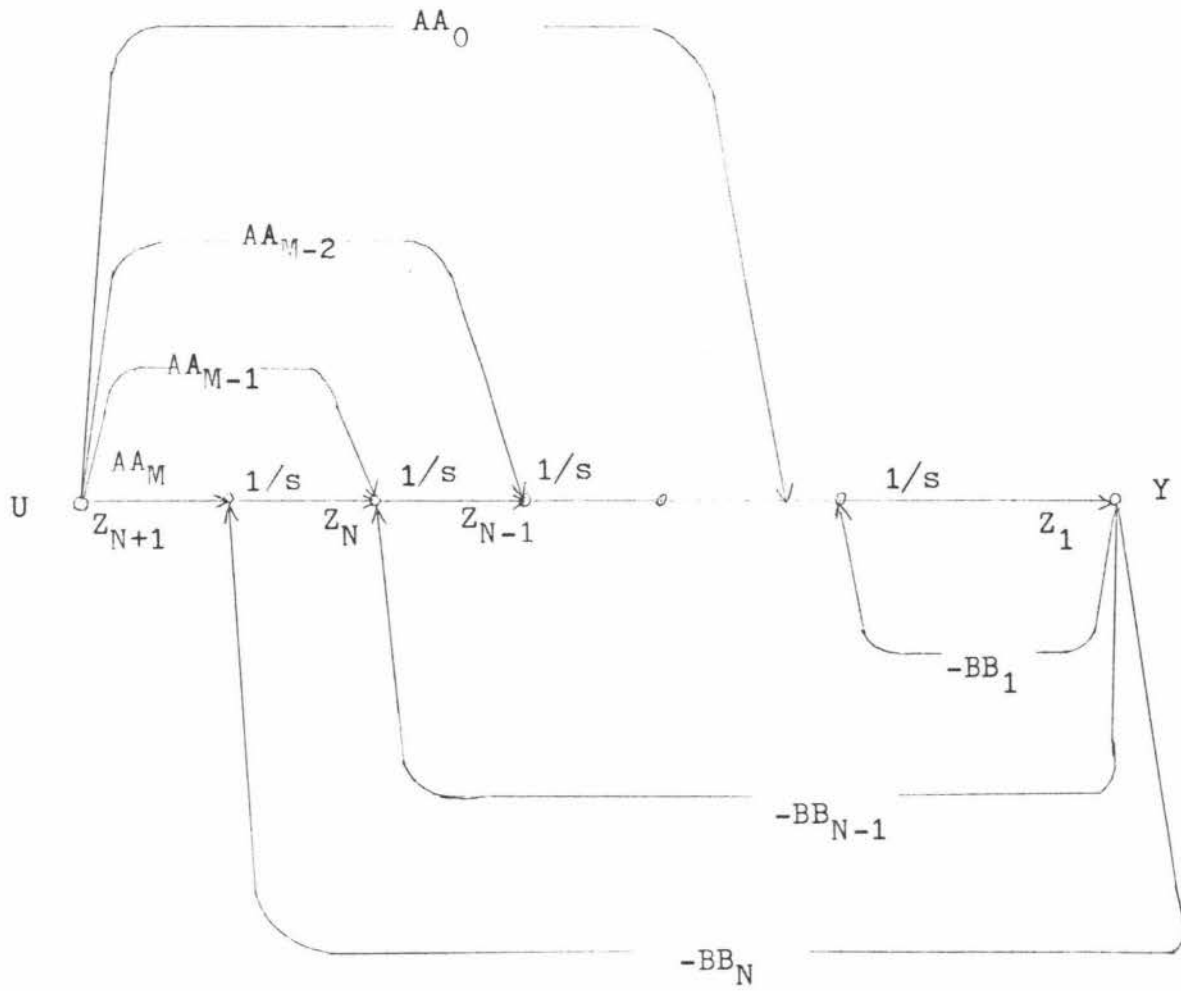


Fig.3.2 State variable diagram of Eq.(3.22)

3.5 USE OF THE PROGRAM

The program is entered directly into the B6700 by command from the Tektronix terminal. When the program is executed, the following steps are performed in order. (Appendix E illustrates a block flow diagram of the program.)

1. The program asks the user to specify the type of response by printing "TIME OR FREQUENCY RESPONSE?" on the screen.
2. The user types "T" for time response and "F" for frequency response.
3. Assuming "F" is typed in, the program then asks for the required option, i.e. open- or closed-loop system, by printing "OPEN- OR CLOSED-LOOP?"
4. The user types in the appropriate answer from the keyboard, "O/P" for open-loop, and "CLOSED" for closed-loop system.
5. The program asks for system gain and coefficients of the polynomials of the constituent transfer functions, i.e. K , $C_1(s)$, $C_2(s)$, $G_1(s)$, $G_2(s)$, $H_1(s)$ and $H_2(s)$. These values are entered from the keyboard as required
6. The program asks the user to specify the type of frequency-response graph to be displayed by printing "WHICH GRAPH? - BODE, NYQUIST, NICHOLS OR R-LOCUS?"

(Note: Root-locus is included as a frequency-response plot to facilitate the programming.)

7. The user enters the appropriate answer to the question in step 6 by typing "B" if Bode diagram is required
 "NY" if Nyquist diagram is required
 "NI" if Nichols chart plot is required
 "R" if Root-locus plot is required.
8. After the variables relating to the required graph are entered, the graph is displayed on the screen. Five seconds after the plotting is finished, the program asks if the

user wants to change the scale of the graph by printing

"DO YOU WANT TO CHANGE THE SCALE?"

9. If "Y" is answered, steps 8, 9 are repeated.
If "N" is answered, the program continues to ask if the user wants any other frequency-response plot, by printing "DO YOU WANT ANY MORE GRAPH?"
10. If "Y" is the answer, step 6, 7, 8, 9, 10 and repeated.
If "N" is the answer, the program asks if the user wants to change to any new values, i.e. new $C_1(s)$, $C_2(s)$, $G_1(s)$, $G_2(s)$, $H_2(s)$, $H_1(s)$ and K , by printing "DO YOU WANT TO CHANGE TO ANY NEW VALUES?"
11. If "Y" is typed in, step 5, 6, 7, 8, 9, 10, 11, are repeated.
If "N" is typed in, the user is asked "DO YOU WANT ANY MORE FREQUENCY-RESPONSE?"
12. If "Y" is entered, step 3, 4, 5, 6, 7, 8, 9, 10, 11, 12 are repeated.
If "N" is entered, the user is asked "WHAT ABOUT TIME-RESPONSE?"
13. If "Y", step 3, 4, 5 are performed and followed by steps 14, 15, 16, 17.
If "N" the program ends.
14. The program asks for the required information to generate the time-response of the system to a step input, such as the value of step-function, the output range and increment, time range and increment,
15. After the plotting of the time-response is finished, the user may change the scale of the graph or value of step-input by typing "Y" in answer to the question "DO YOU WANT TO CHANGE THE SCALE?"
16. If "Y" step 14, 15, 16 are repeated.

If "N", the user is asked "DO YOU WANT TO CHANGE TO ANY ANY NEW VALUES?"

17. If "Y" step 13 is repeated.

If "N" the program ends.

However, if in step 2, the answer to be typed in is "T" instead of "F", i.e. the time-domain response is required first, the procedure to be followed is:

18. Step 3, 4, 5, 14, 15, 16 are performed.

19. If "N" is answered to question "DO YOU WANT TO CHANGE TO NEW VALUES?" from step 16, steps 20, 21, 22 are performed.

If "Y", steps 18, 19 are repeated.

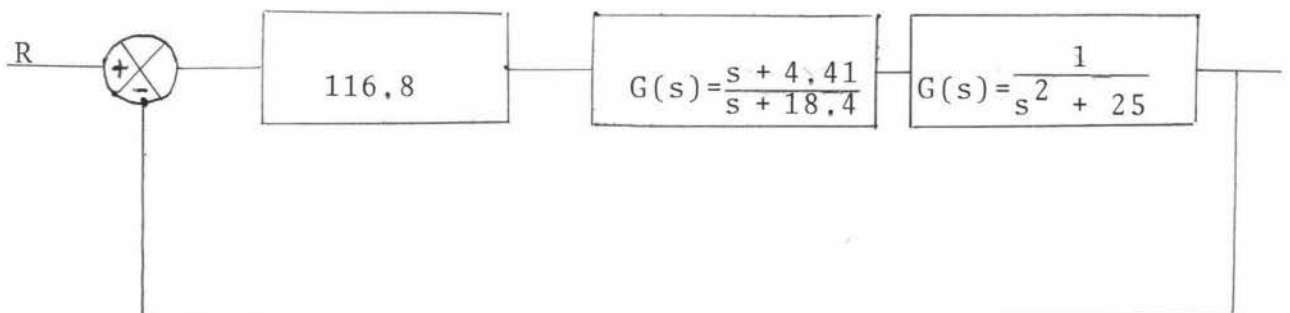
20. The program asks "WHAT ABOUT FREQUENCY-RESPONSE?"

21. If "Y" is entered, steps 3, 4, 5, 6, 7, 8, 9, 10, 11 are performed.

22. If "N" is answered to question "DO YOU WANT ANY MORE FREQUENCY-RESPONSE?" from step 11, program ends

If "Y" is entered, steps 21, 22 are performed

The following example illustrates the use of the program. Consider the direct feedback control system shown below:



If a Bode diagram of the open-loop transfer function of the system is to be generated, the communication between the computer and the user begins as follows:

C: TIME OR FREQUENCY RESPONSE?

U: "F"

C: O/P OR CLOSED-LOOP?

U: "O/P"

C: VALUE OF K?

U: 116.8

C: DEGREE OF C_1 , C_2 , G_1 , G_2 , H_1 , H_2 ?

U: 1, 1, 0, 1, 0, 0

C: COEFFICIENTS OF $C_1(s)$?

U: 1, 4.41

C: COEFFICIENTS OF $C_2(s)$?

U: 1, 18.4

C: COEFFICIENT OF $G_1(s)$?

U: 1

C: COEFFICIENTS OF $G_2(s)$?

U: 1. 2. 0

C: COEFFICIENT OF $H_1(s)$?

U: 1

C: COEFFICIENT OF $H_2(s)$?

U: 1

C: WHICH GRAPH? - BODE, NYQUIST, NICHOLS OR ROOT-LOCUS?

U: "B"

C: FREQUENCY RANGE?

U: 1, 100

C: GAIN RANGE?

U: -40, 40

C: PHASE RANGE?

U: -180, 0

C: GAIN, PHASE AND FREQUENCY INCREMENT?

U: 10, 30, 1

C: IS GRID WANTED?

U: "Y"

After the plotting is finished, the communication between the user and the computer continues:

C: DO YOU WANT TO CHANGE THE SCALE?

U: "N"

C: DO YOU WANT MORE GRAPH?

If the user does not want any more graph she should answer

U: "N"

C: DO YOU WANT TO CHANGE TO NEW VALUES?

If the user wants to change the T.F of the system, she should type "Y", and the process starts from the beginning. If she does not wish to do so, she can type "N", the computer then answers

C: DO YOU WANT ANY MORE FREQUENCY PLOT?

This gives the user option to change from open-loop to closed-loop T.F. or vice versa. If she does not want any more frequency-response plot of the system, either open- or closed-loop, she may type

U: "N"

C: WHAT ABOUT TIME-RESPONSE?

U: "N"

End of the program.

Other graphs, such as time-response, Nyquist diagram, Nichols chart plot and Root-locus may be generated in the same manner, with different questions to answer.

3.7 DISCUSSION

This chapter has decided the software for computer-aided control system design. Specific design examples are described in Chapter 4. These examples further illustrate aspects of the

design software. A full listing of the programme and details of its constituent procedures are presented in Appendices D and G.

The software development exercise has been taken to a certain stage only because of the limited project time. There are many more options open for enhanced development to further facilitate the design procedure. For example

- i. the direct display of the gain and phase margins on the frequency-response charts;
 - ii. the direct display of frequency at any point on the Nichols chart and Nyquist diagram;
 - iii. the direct display of the open-loop gain at any point on the root-locus plot
- and iv. automatic evaluation of the angle and magnitude condition on the root-locus diagram.

Many other features can be incorporated to make the design procedures more automatic.

-ooOoo-

CHAPTER FOUR

RESULTS

Three illustrative design examples are presented in this chapter. These utilize the interactive computer software. The first example is via the trial-and-error approach. The second example is through analysis, has system performance specifications given in terms of time-domain quantities. The third example, through analysis, has system performance specifications given in terms of frequency-domain quantities.

4.1 DESIGN EXAMPLE VIA TRIAL-AND-ERROR APPROACH

Consider the system shown in Figure 4.1. With no stringent performance specification required, it is desired to compensate the system so that the system has satisfactory relative stability and high-precision closed-loop control.

The following information obtained from the computer-generated graphs indicates that the system is unstable when the loop is closed.

From Figure 1 showing the closed-loop time response to a unit-step function input, the transient terms are seen oscillatory with an increasing exponential envelope.

From Figure 2 showing the Nyquist diagram of the open-loop system, the locus encloses the point $(-1(j\omega))$ as ω is varied from .1 to 20 rad/sec.

From Figure 3, showing the Root-locus of the open-loop system, the loci pass into the R-H half of the s-plane (i.e. poles with positive real parts) when the open-loop gain is increased from zero to 30.

From Figures 4, 5 showing the Nichols chart plot and the Bode diagram of the open-loop system, the gain and phase margins are found to be -8.67db and -16° .

Hence, the system is unstable

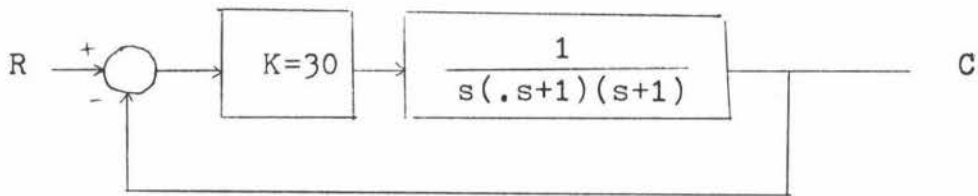


Fig. 4.1 uncompensated system

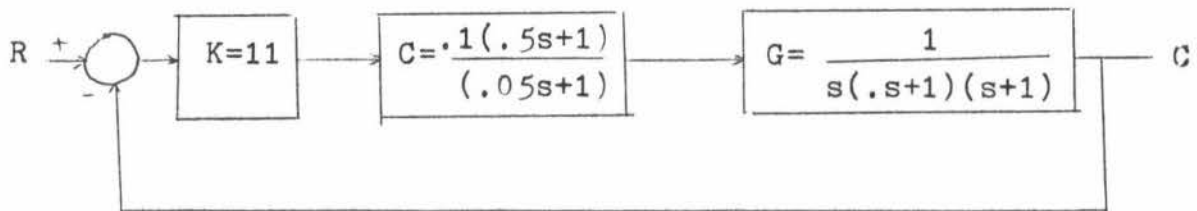


Fig. 4.2 The Lead-compensated system

Reducing the system open-loop gain K is the first step in improving the behaviour of the system. The resulting characteristics of the system are examined and it is found that the system is on the verge of instability for $K = 11$, as seen in the following graphs.

From Figure 6 showing the time response of the closed-loop system to a unit-step input, the output is a stable amplitude sinusoidal oscillation.

From Figure 7 showing the Nyquist diagram of the open-loop system, the locus passes the point $(-1, j0)$ as ω is varied from .01 to 9 rad/sec.

Figure 8 shows the Root-locus of the O/P-loop system. For the O/P-loop gain $K = 11$, the poles of the closed-loop T.F. are found on the imaginary axis (i.e. the real parts $= 0$). The loci will pass into the R H. half of the s -plane if $K > 11$ and the system is then unstable.

Figure 9 shows the Nichols chart plot of the O/P-loop system. The gain and phase margins are found zero.

Figure 10 shows the Bode diagrams of the O/P-loop system. Gain and phase margins are found zero, as calculated from Figure 9.

The above information illustrates the 'marginal stability' of the system, in which case the output oscillates with a stable amplitude.

The O/P-loop gain K may be further reduced to make the above 'marginally stable' system stable. However, by improving the system stability, the speed of response and therefore the delay and settling time are increased.

Thus, high precision closed-loop control and good stability are incompatible, as seen in Figure 11, Figure 12, Figure 13, and Figure 14.

Resulting quantities measuring the performance of the closed-

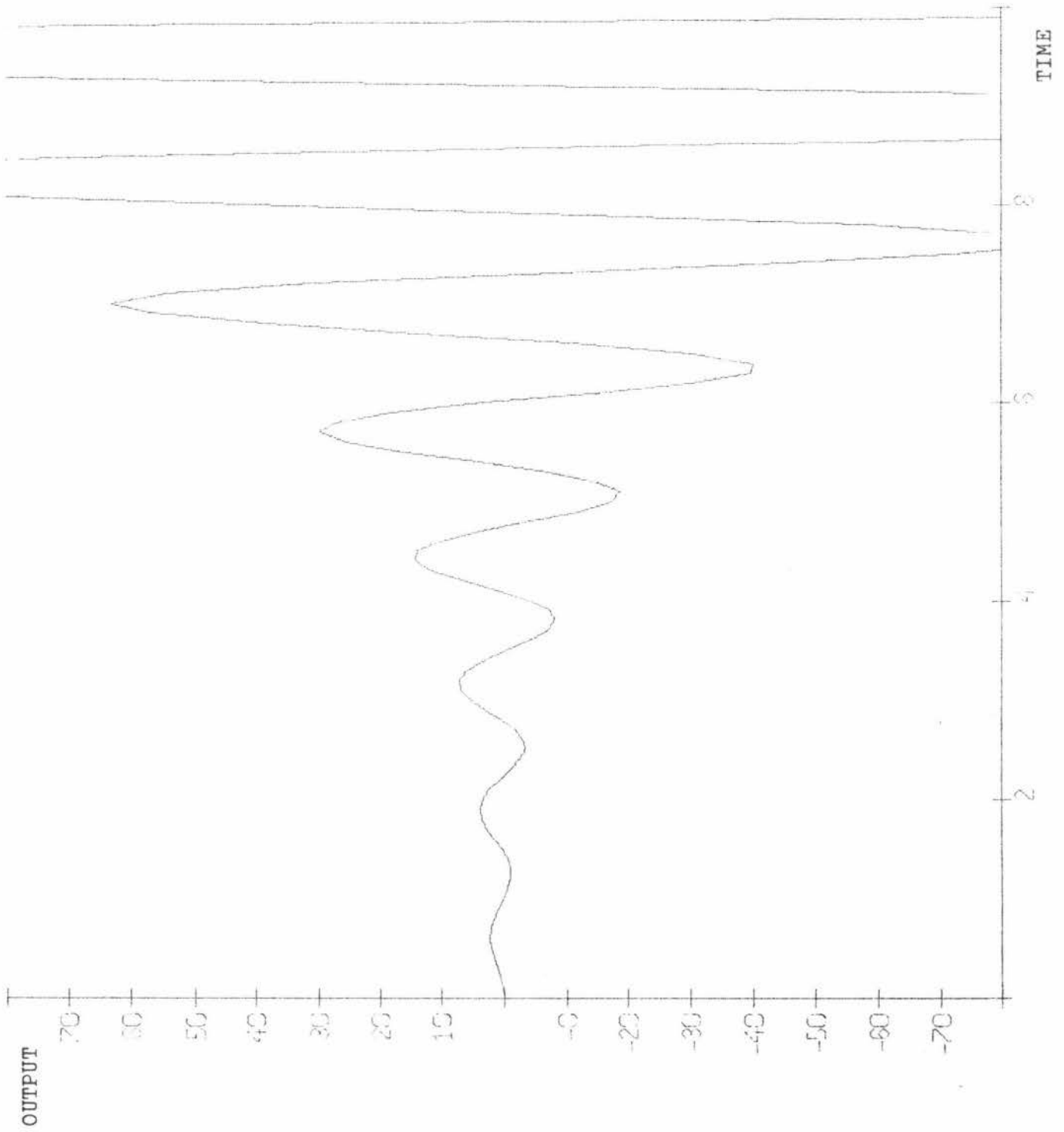


FIG. 1

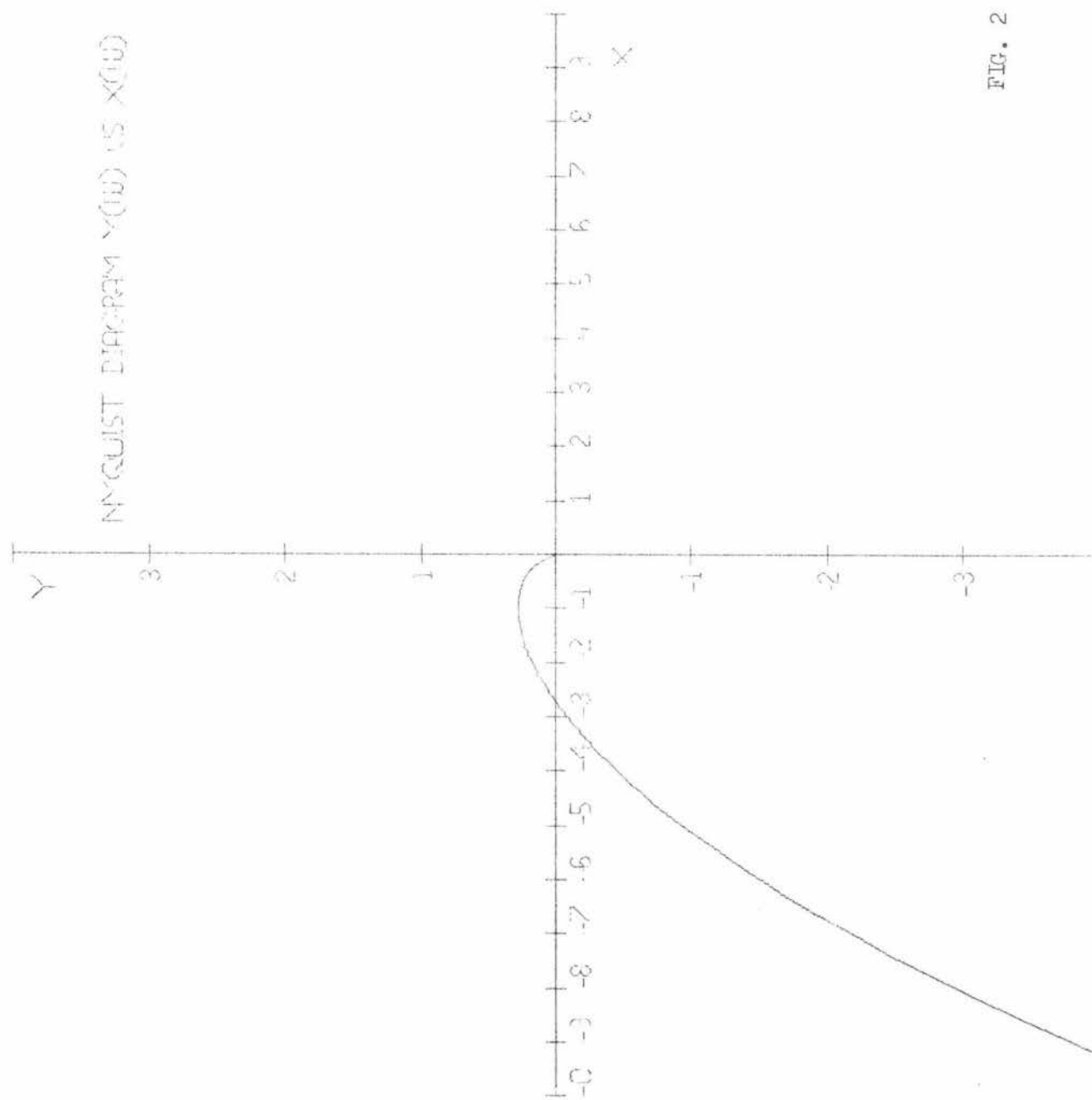


FIG. 2

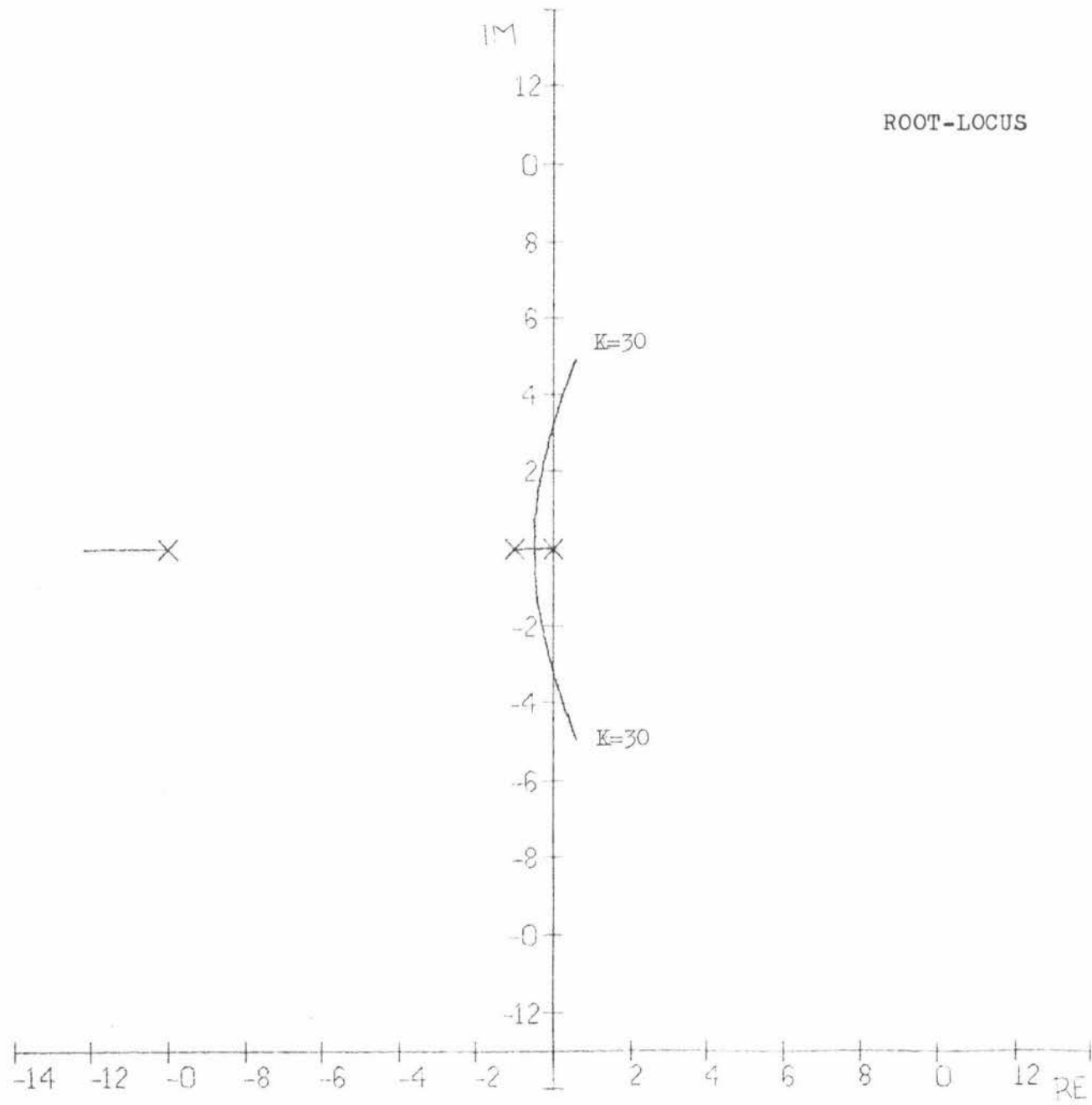


FIG. 3

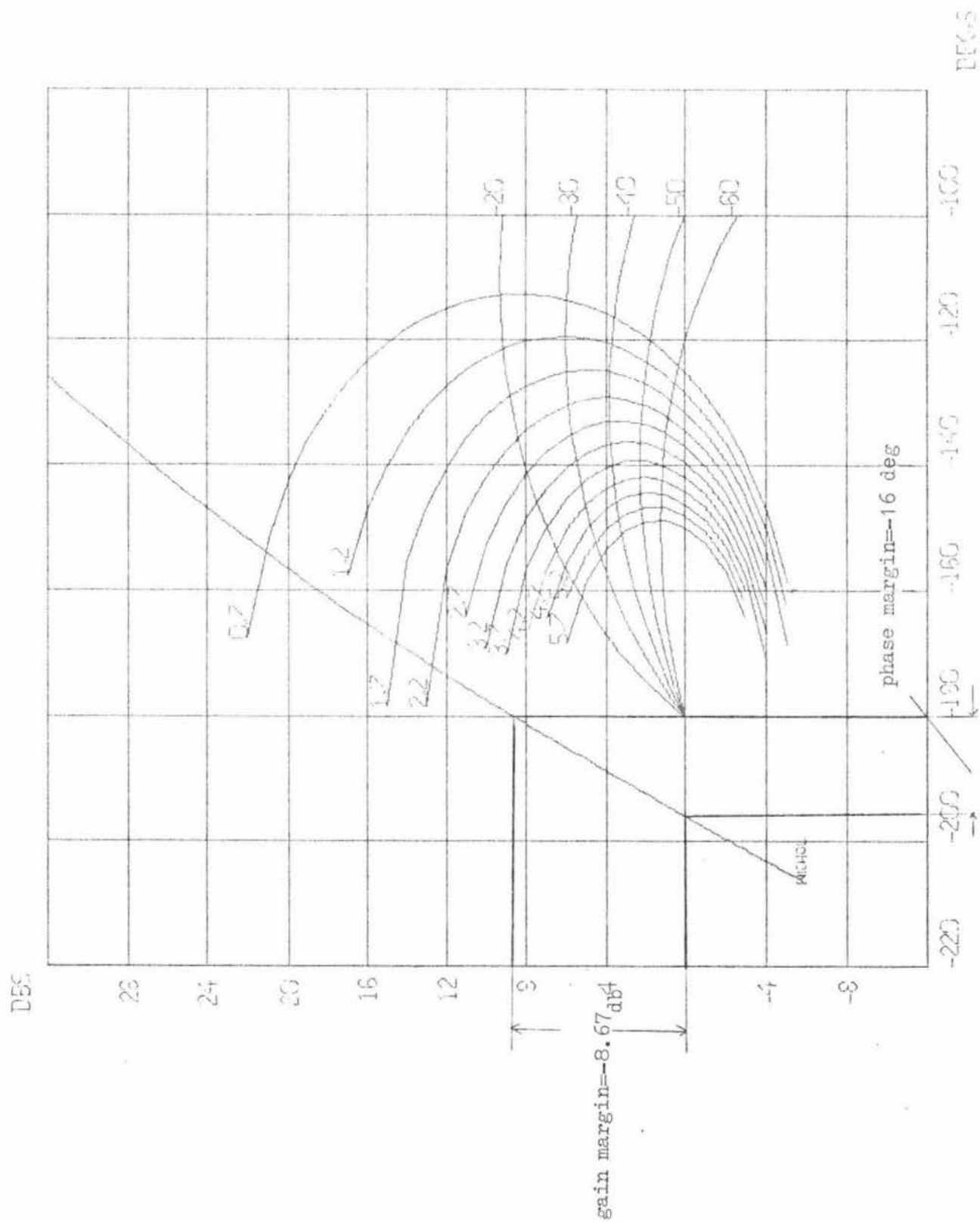


FIG. 4

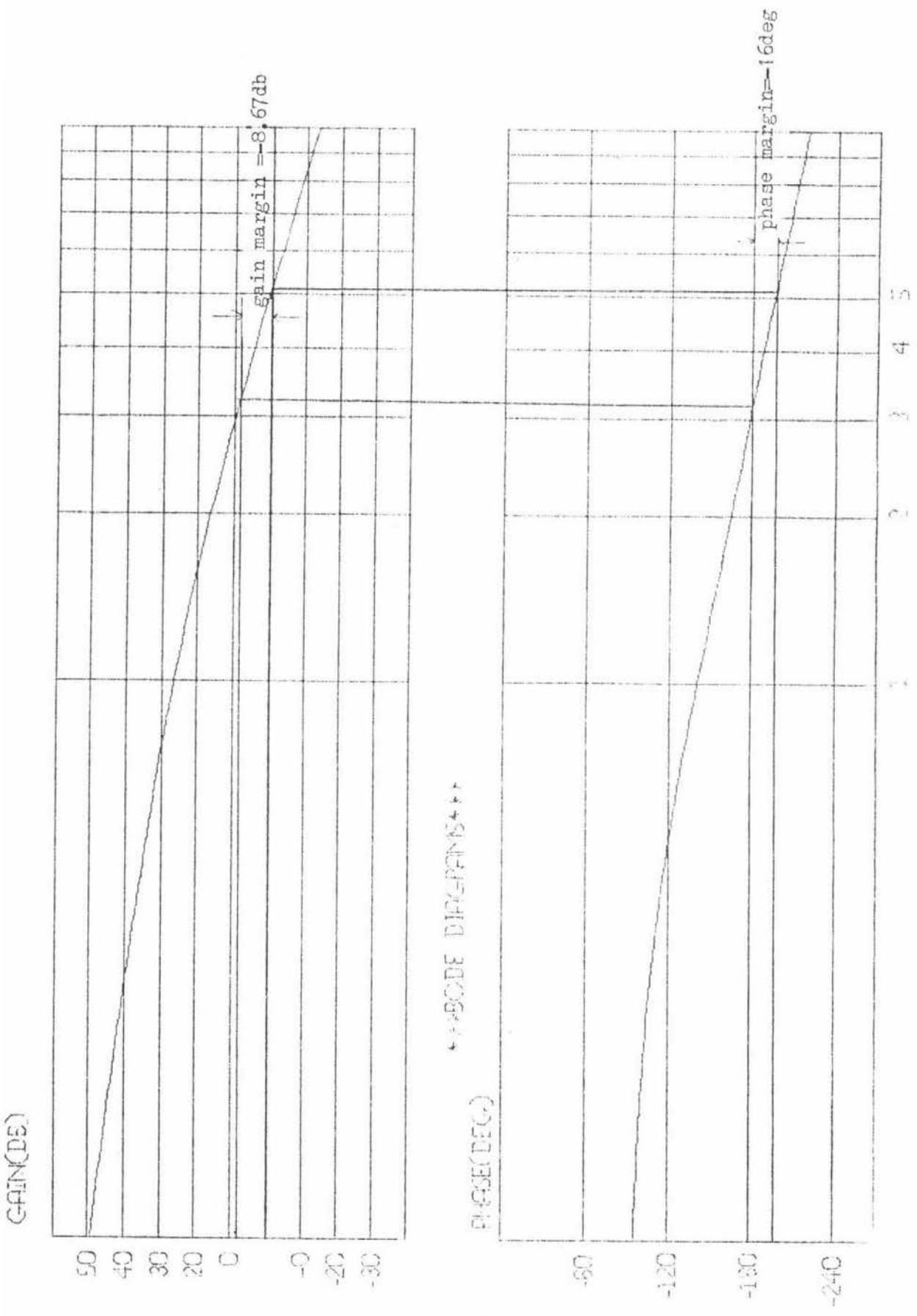
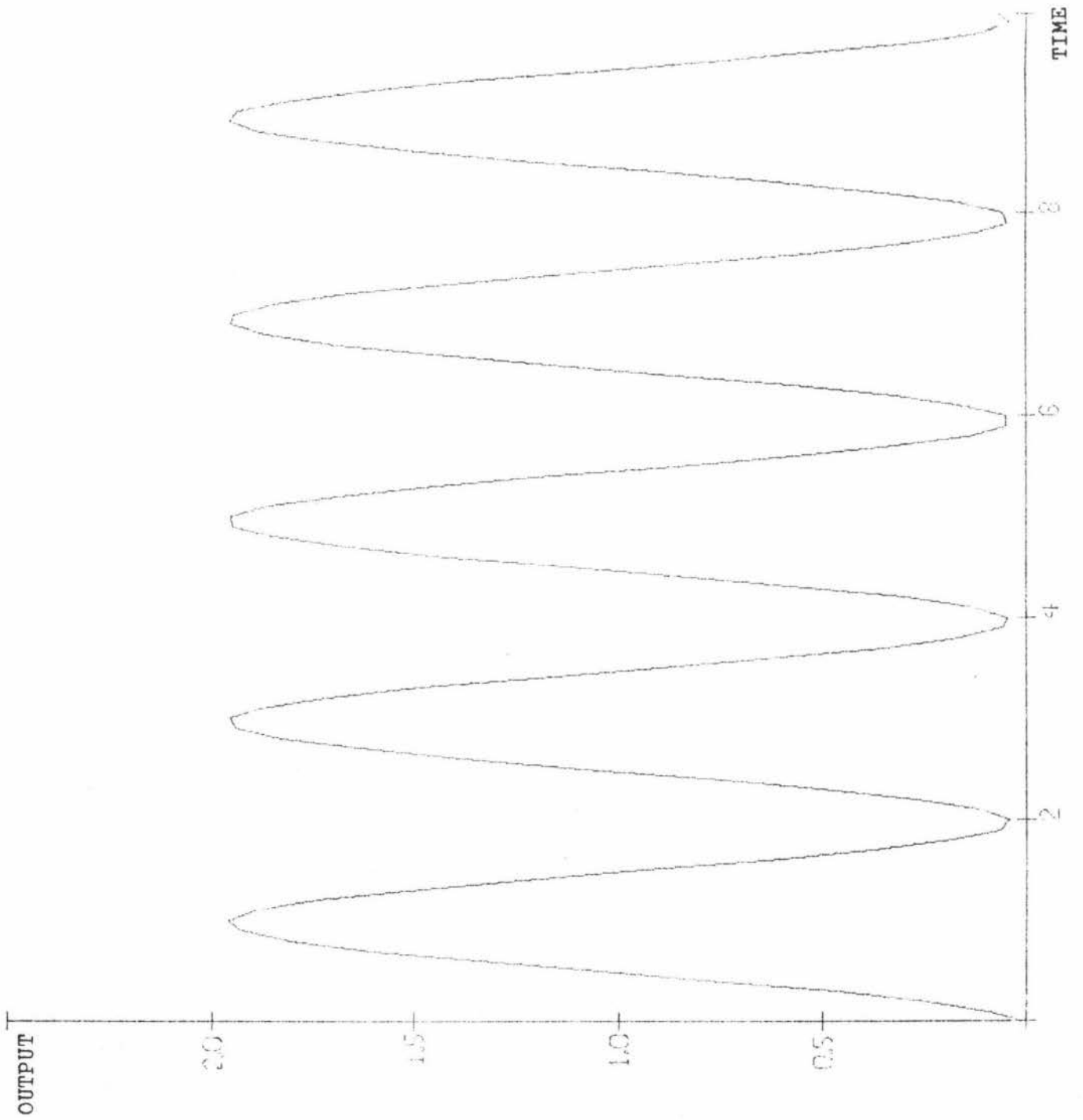


FIG. 5

FREQUENCY(HTZ)

FIG. 6



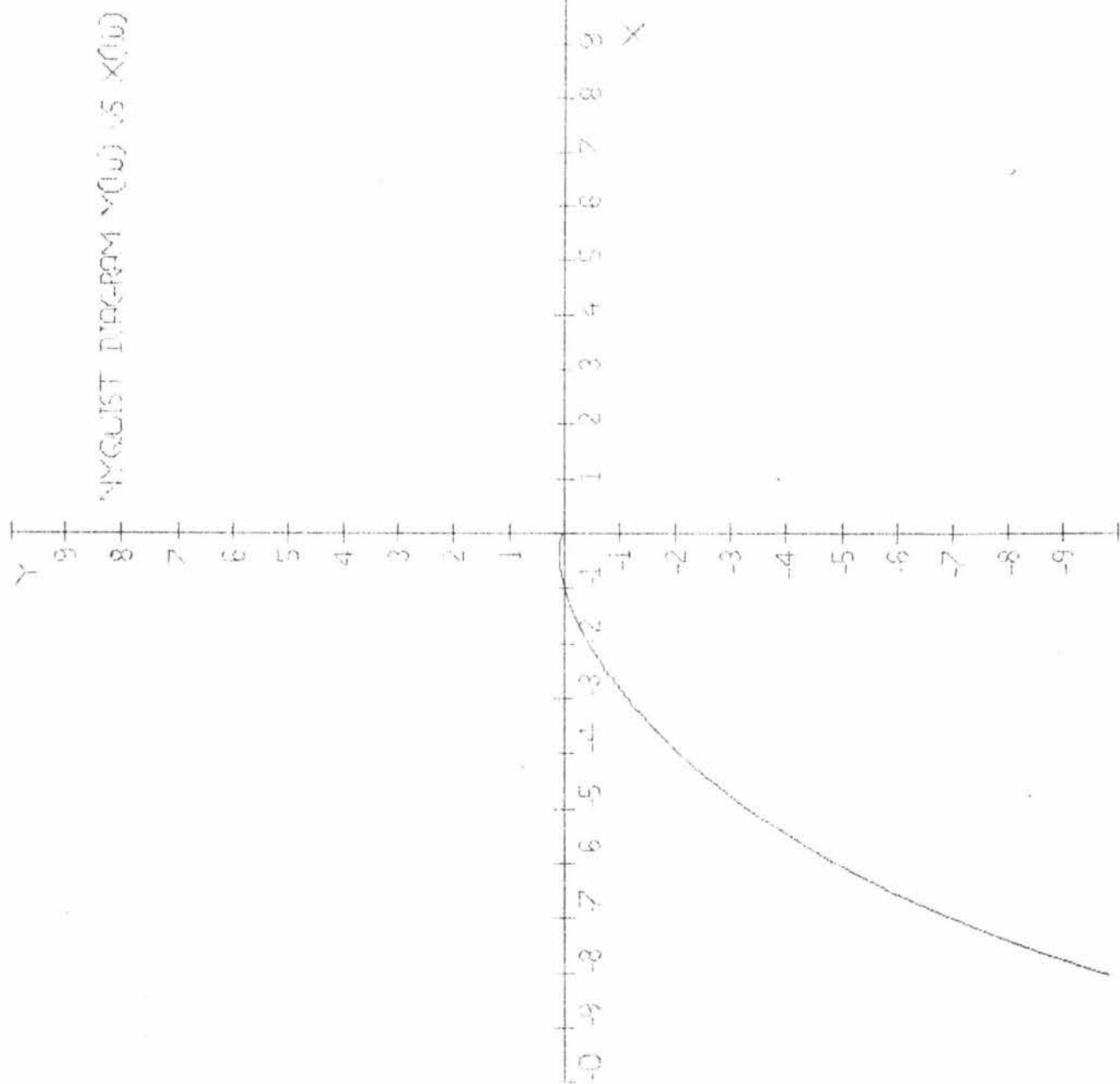


FIG. 7

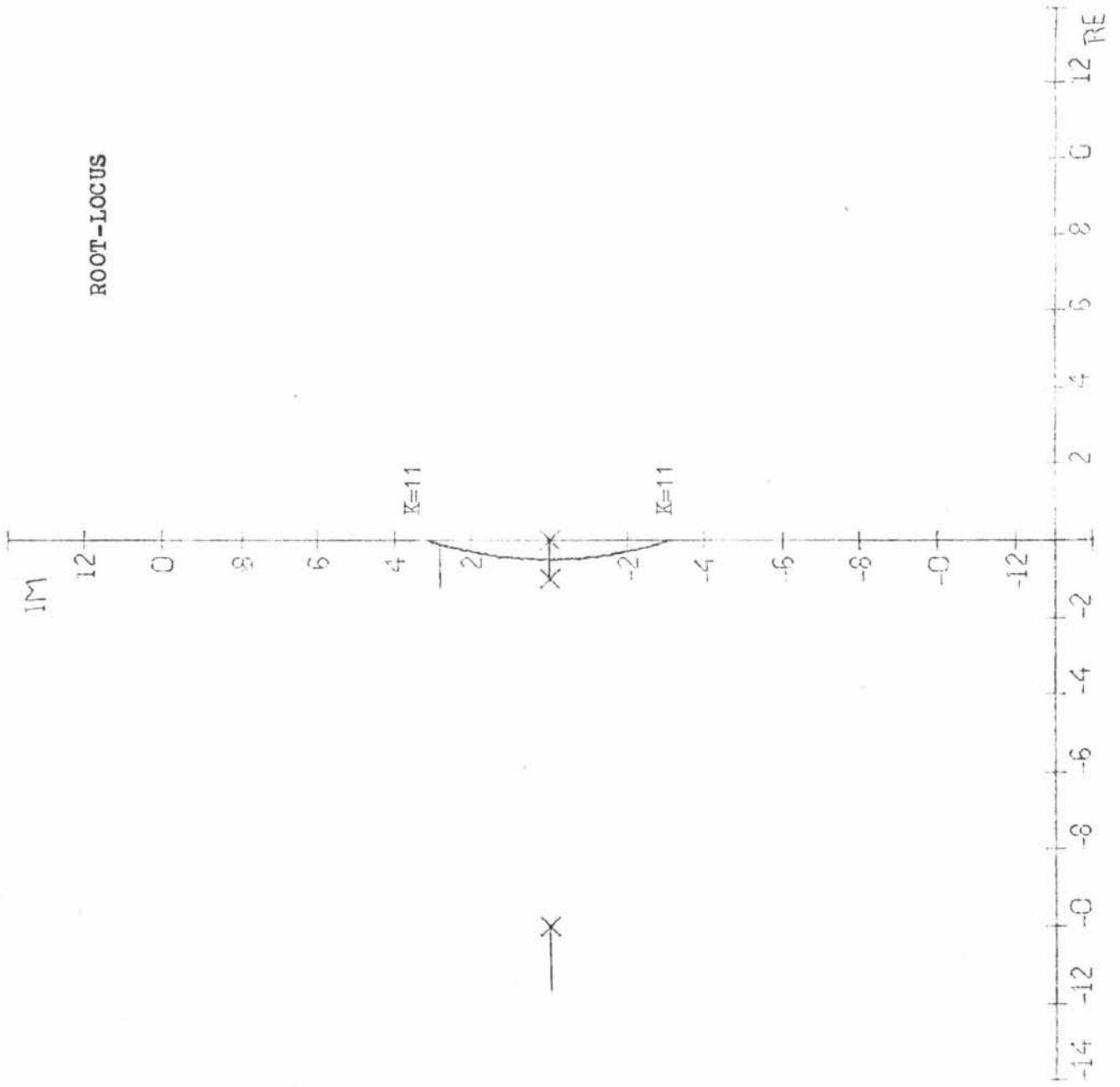
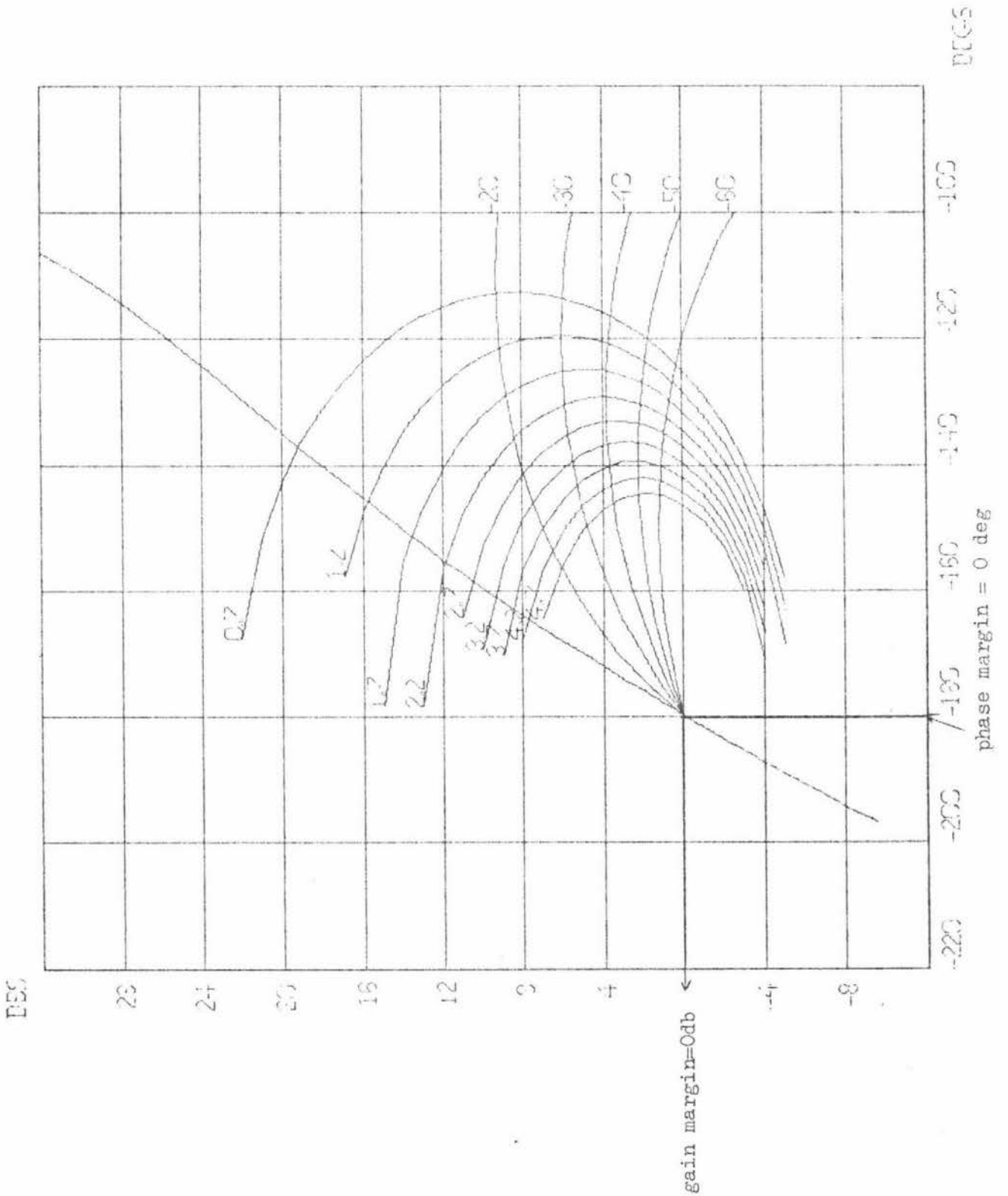


FIG. 8

FIG. 9



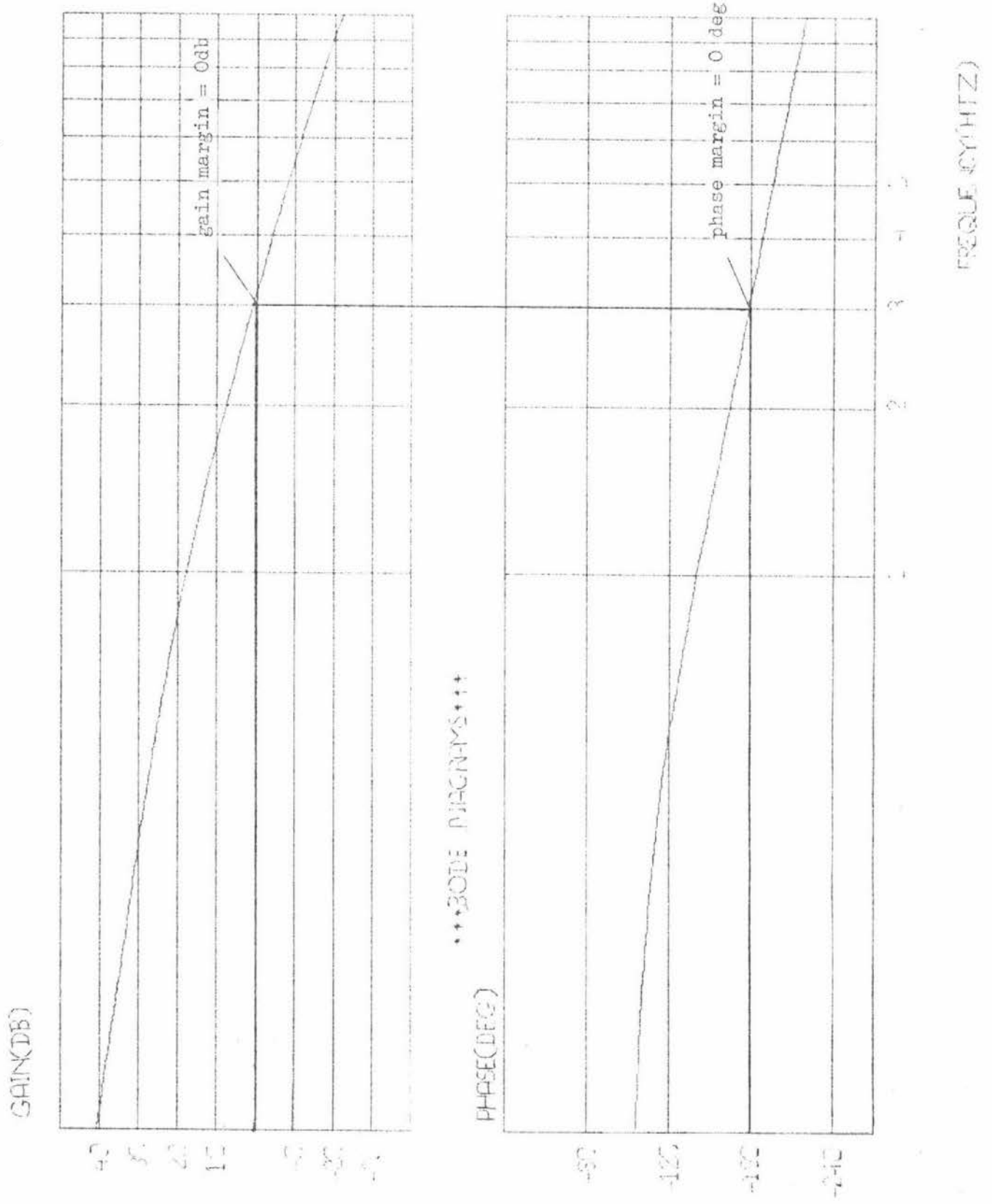


FIG. 10

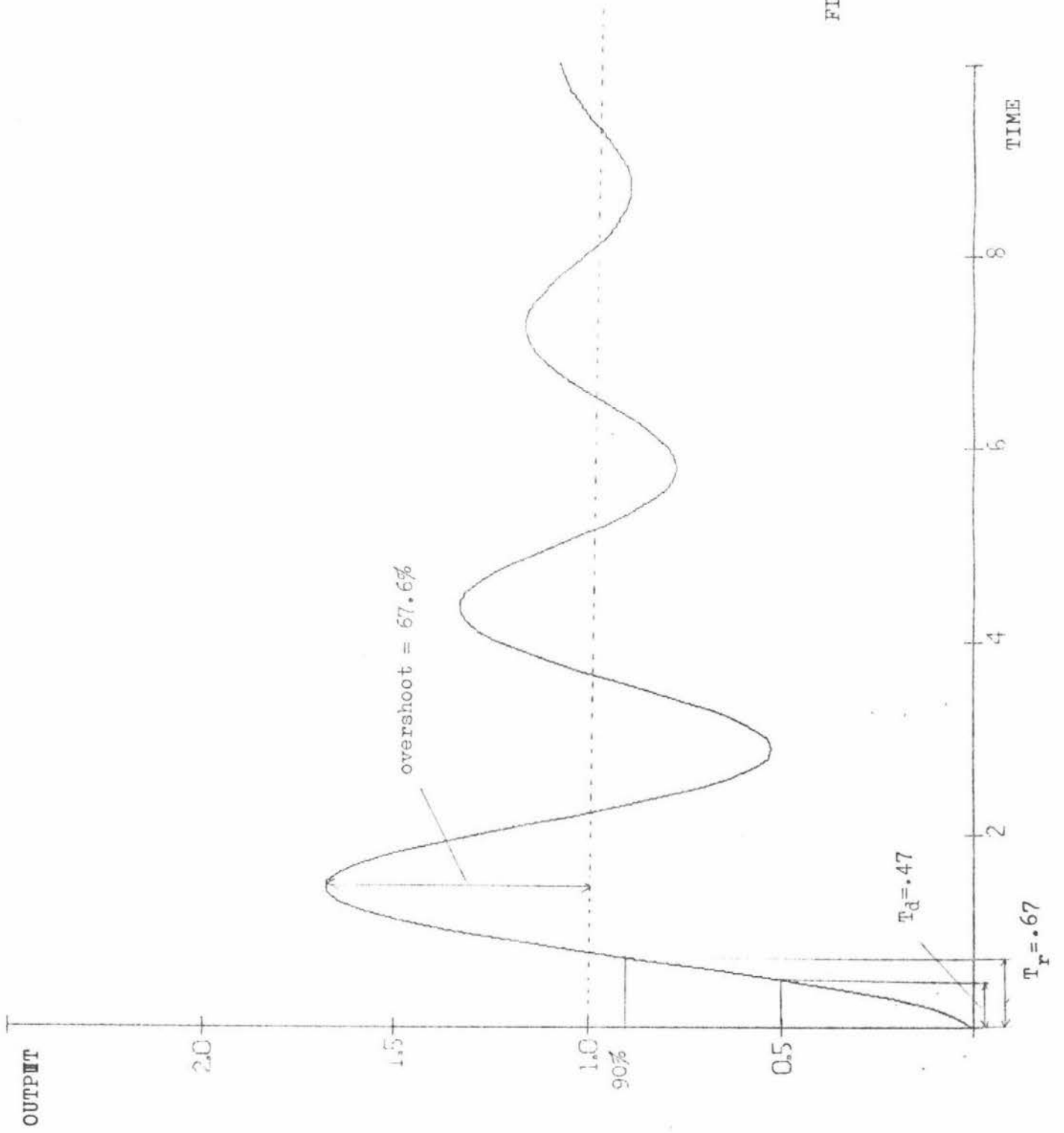


FIG. 11

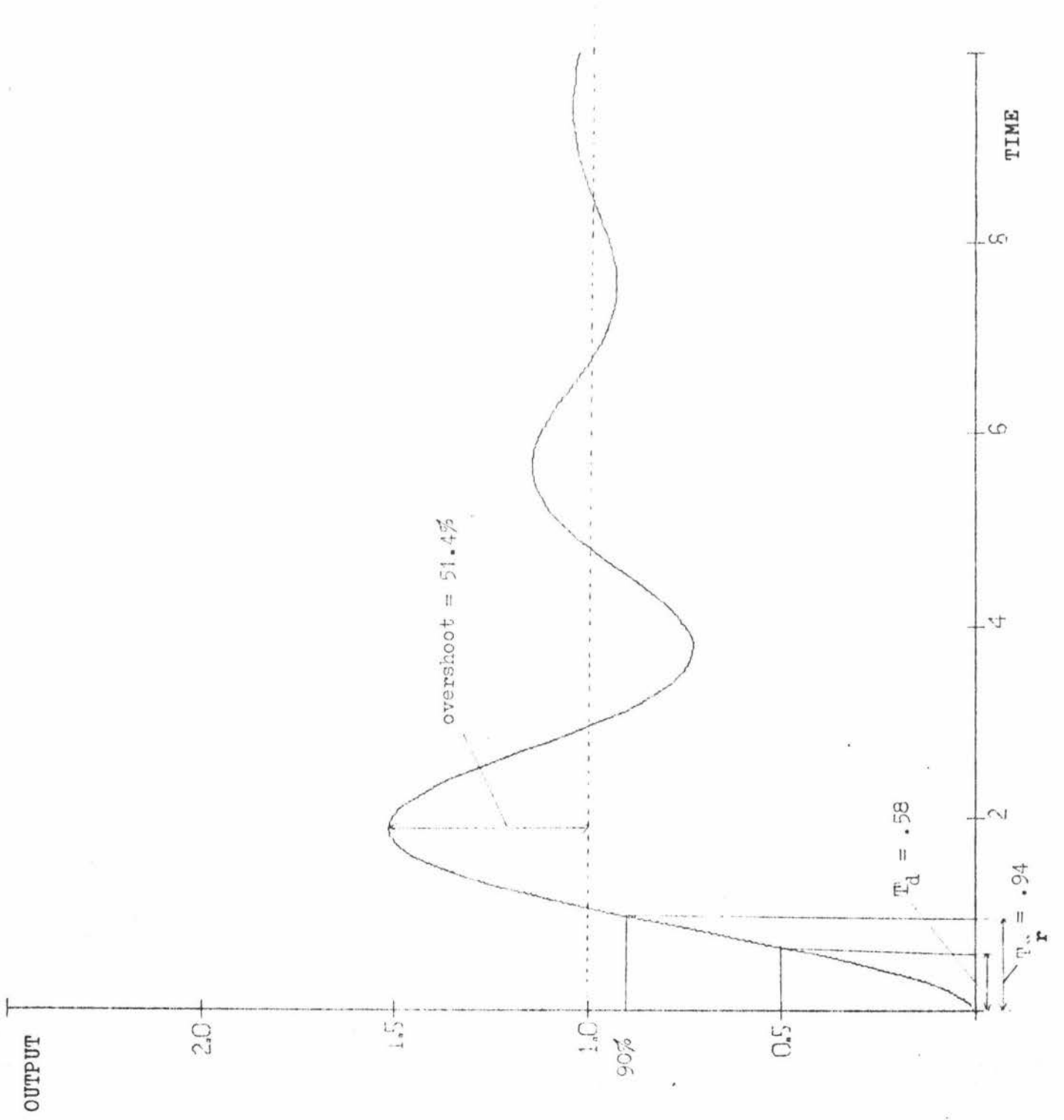


FIG. 12

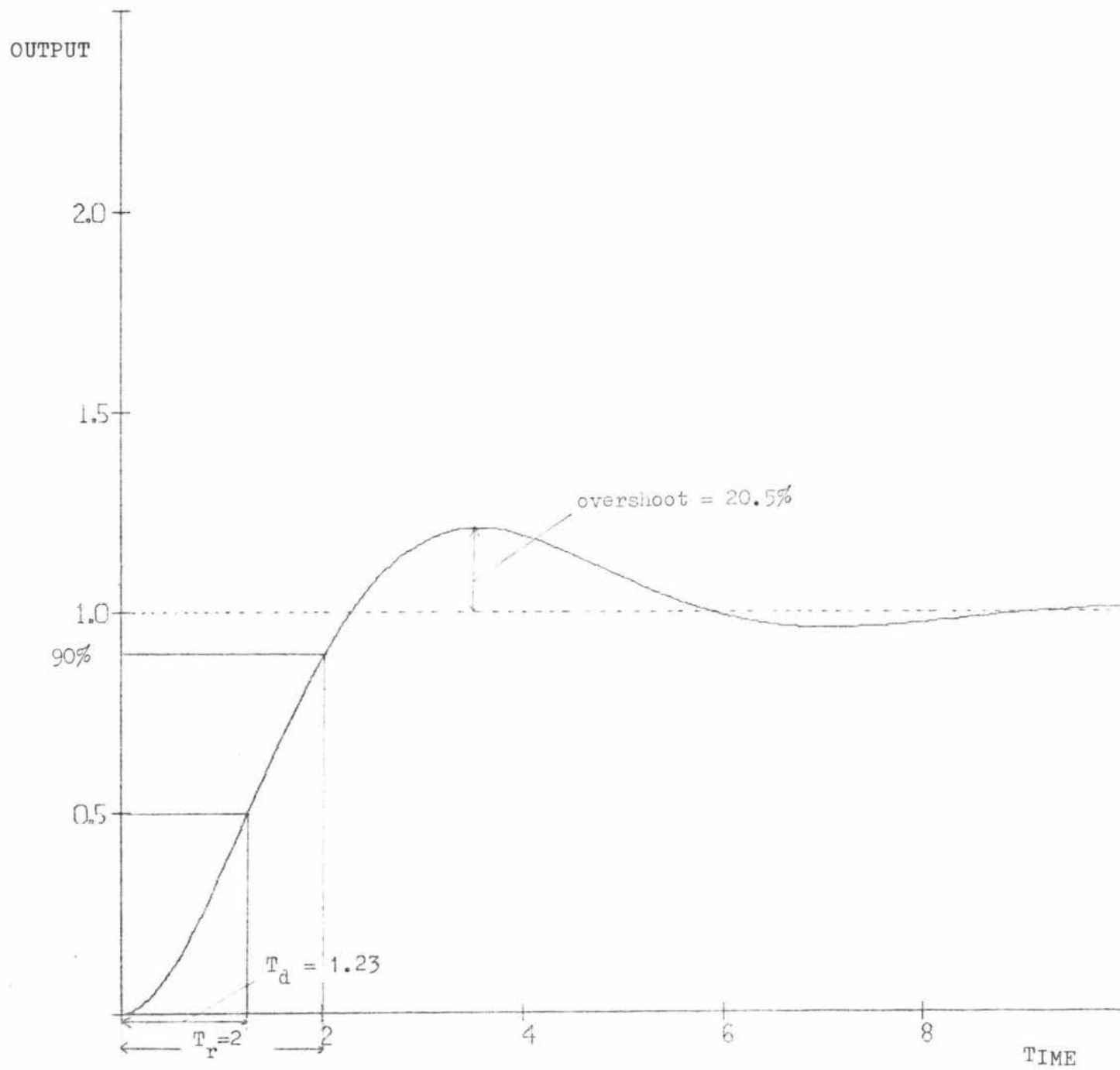
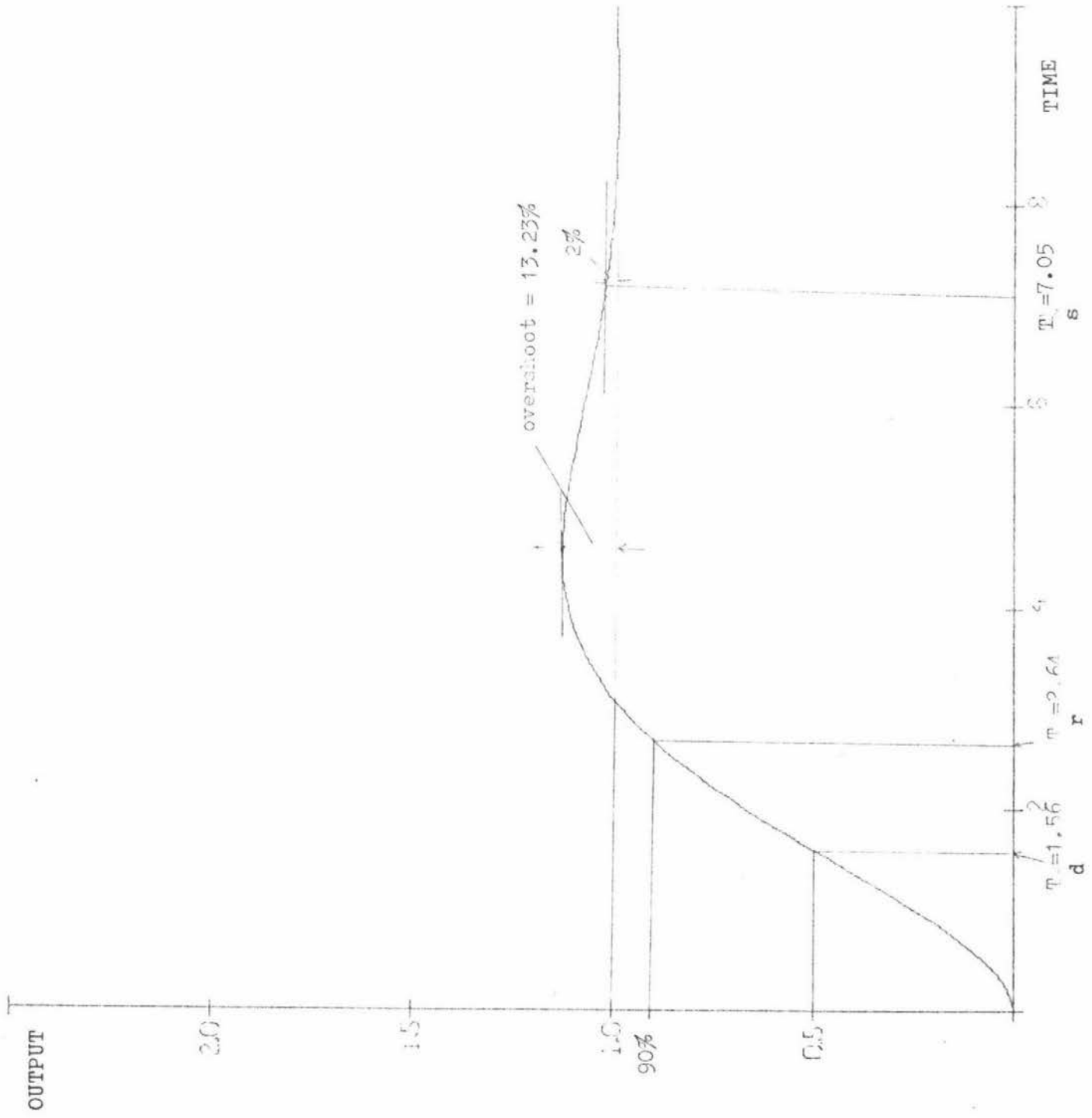


FIG. 13

FIG. 14



OUTPUT

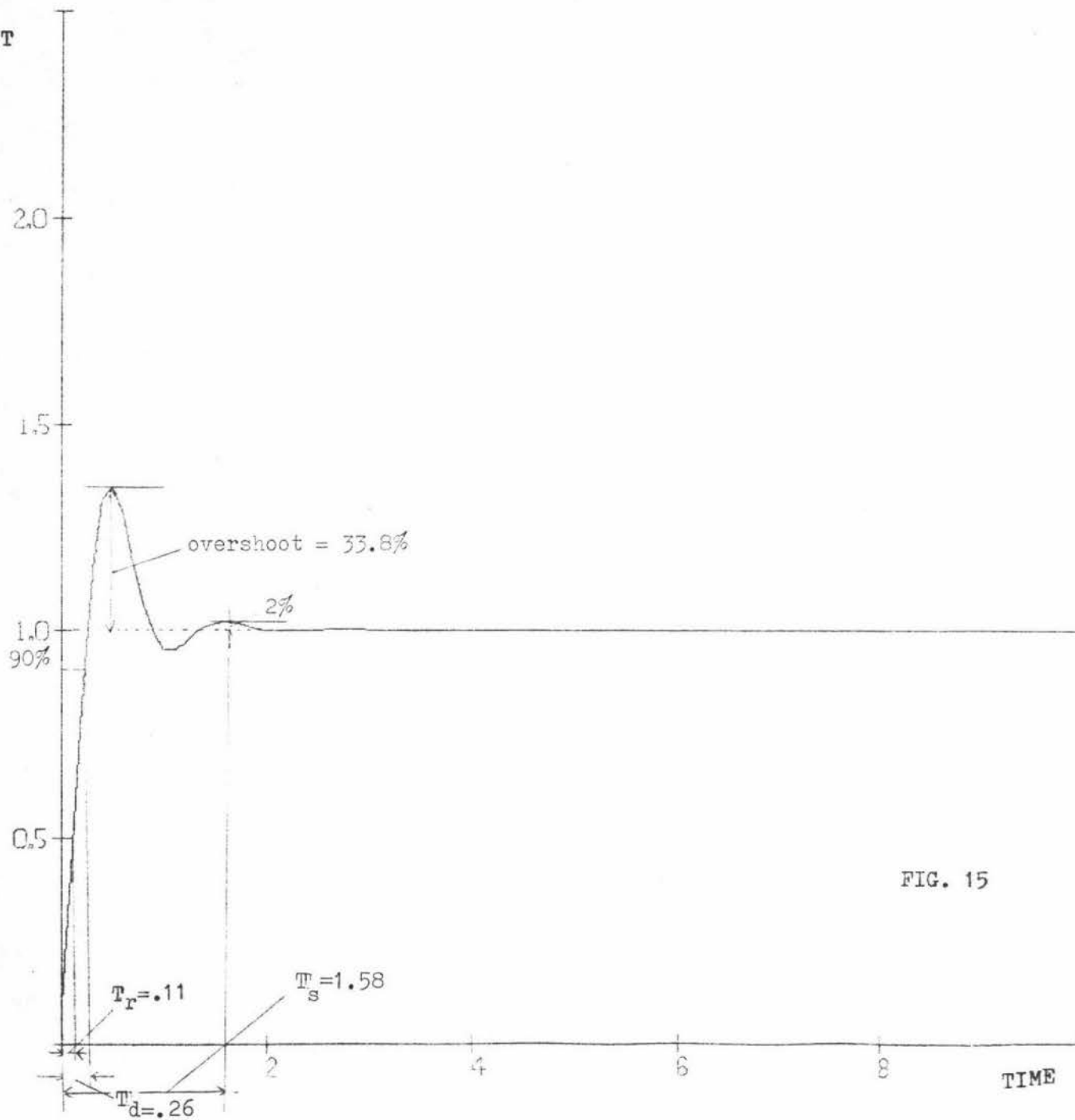


FIG. 15

loop system with different values of K are shown in Table 4.1.

Therefore the reduction of the O/P-loop gain alone may not provide sufficient alteration to produce a satisfactory performance. It is then necessary to redesign the system by incorporating additional compensation in order to improve the overall system behaviour.

The best possible design is a compromise between good stability and as high a loop gain as possible. A lead compensation network is added to the system with the O/P-loop gain $K = 11$, the transfer function of the lead compensator is

$$G_c(s) = \frac{1}{\beta} \left(\frac{1 + sT}{1 + \frac{sT}{\beta}} \right)$$

where β and T are chosen arbitrarily: $\beta = 10$ is a sensible practical value and T is chosen as .5. The compensated system is shown in Figure 4.2. System characteristics can be seen in Fig. 15, Fig. 16, Fig. 17, Fig. 18 and Fig. 19.

From Fig. 15 representing the closed-loop time response to a unit-step function input, the following quantities

- overshoot = 33.8%
- delay time = .26secs
- rise time = 11 secs
- settling time = 1.58secs

indicating a satisfactory system performance.

From Fig.16 representing the Nyquist diagram of the O/P-loop system, the locus does not enclose the point $(-1,0)$ as ω is varied from .01 to 30rad/sec,

From Fig.17 representing the Root-locus of the O/P-loop system, the loci do not pass into the R.H half of the s -plane (i.e. there are no poles with +ve real parts) as K is increased from 0 to 11,

From Fig.18 representing the Nichols chart plot of the O/P-loop system, the gain and phase margins are found 13.7db and 38.1° .

Table 4.1

		Overshoot	Rise Time	Delay Time	Settling Time	State of Output
Case 1	K = 5	67.6%	.67 secs	.47 secs	$T_1 \gg 10$ secs	- poor stability - more oscillation
Case 2	K = 3	51.4%	.94 secs	.58 secs	$10 < T_2 < T_1$	- better stability - less oscillation - slower response
Case 3	K = 1	20.5%	2 secs	1.23 secs	$T_3 < T_2$	- good stability - sluggish response
Case 4	K = .7	13.23%	2.64 secs	1.56 secs	$T_4 = 7.05$ secs	- good stability - more sluggish response

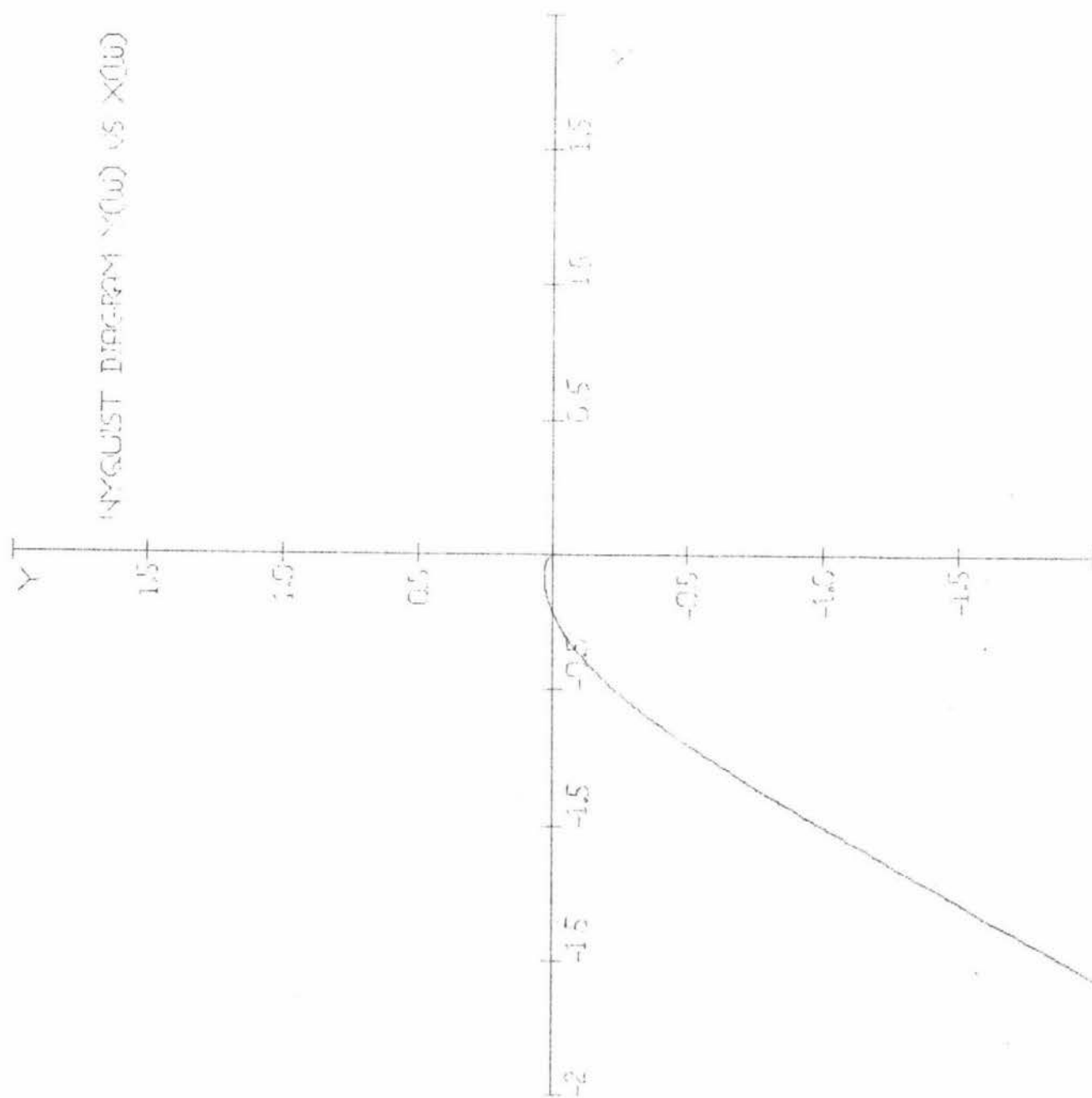
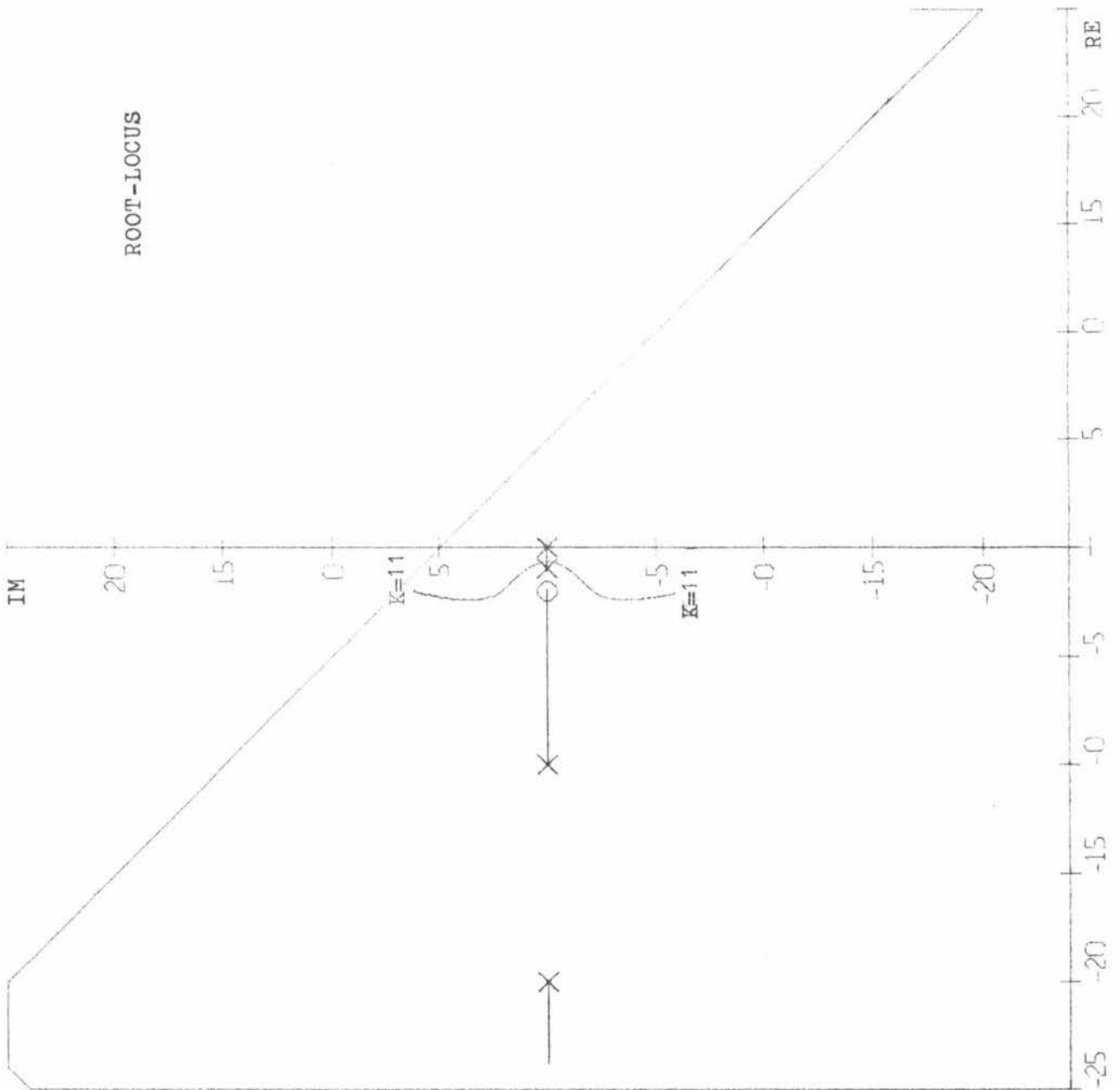


FIG. 16

FIG. 17



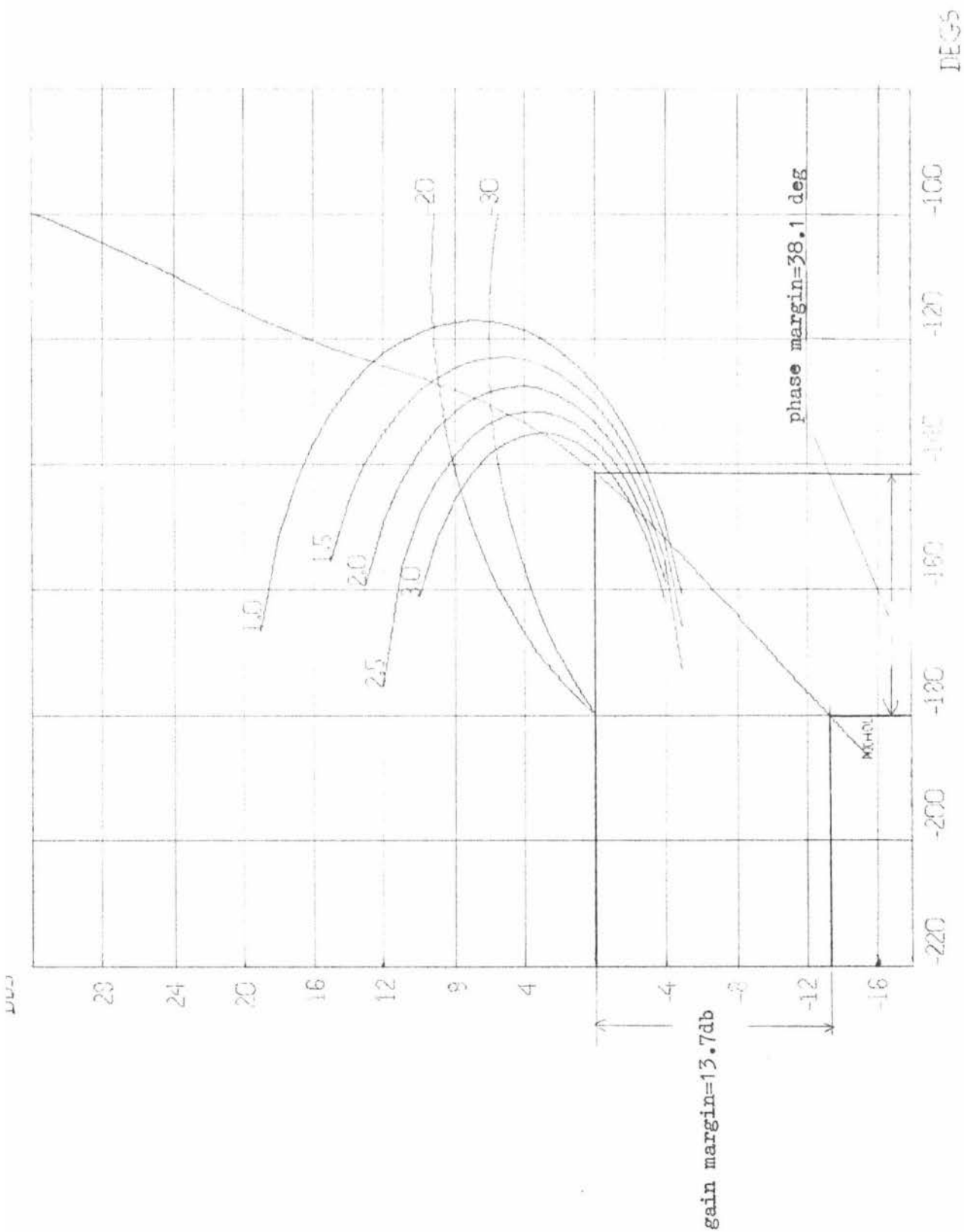
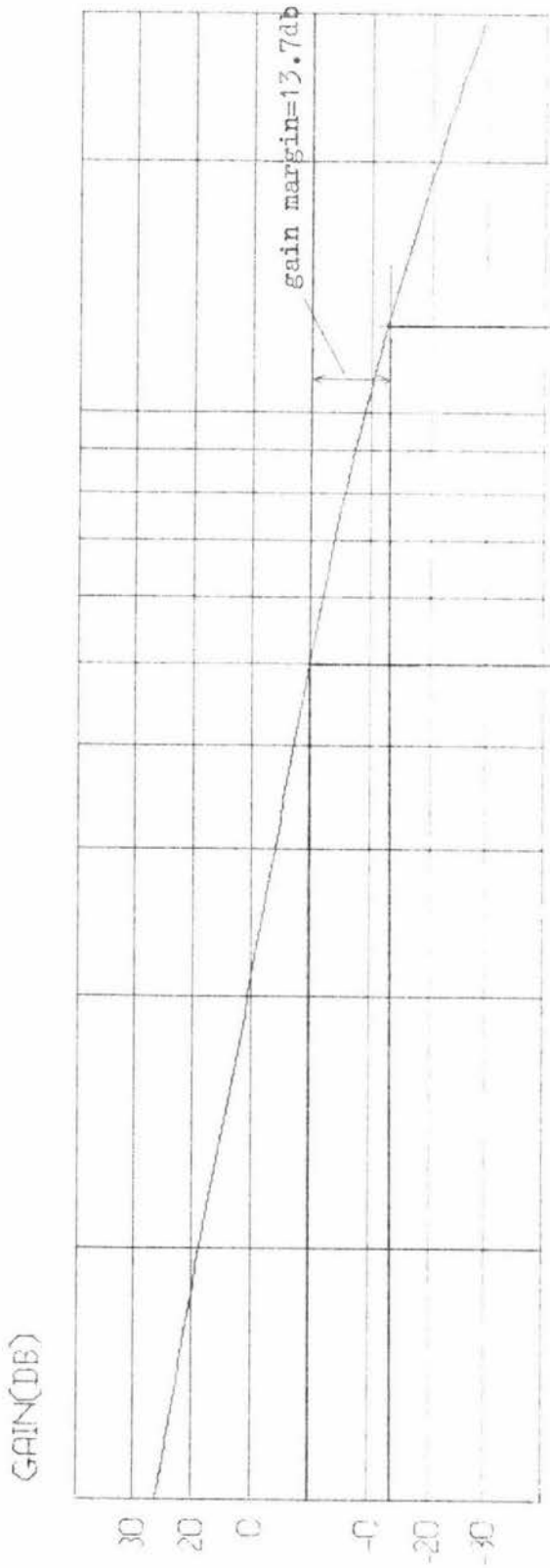


FIG. 18



BODE DIAGRAMS

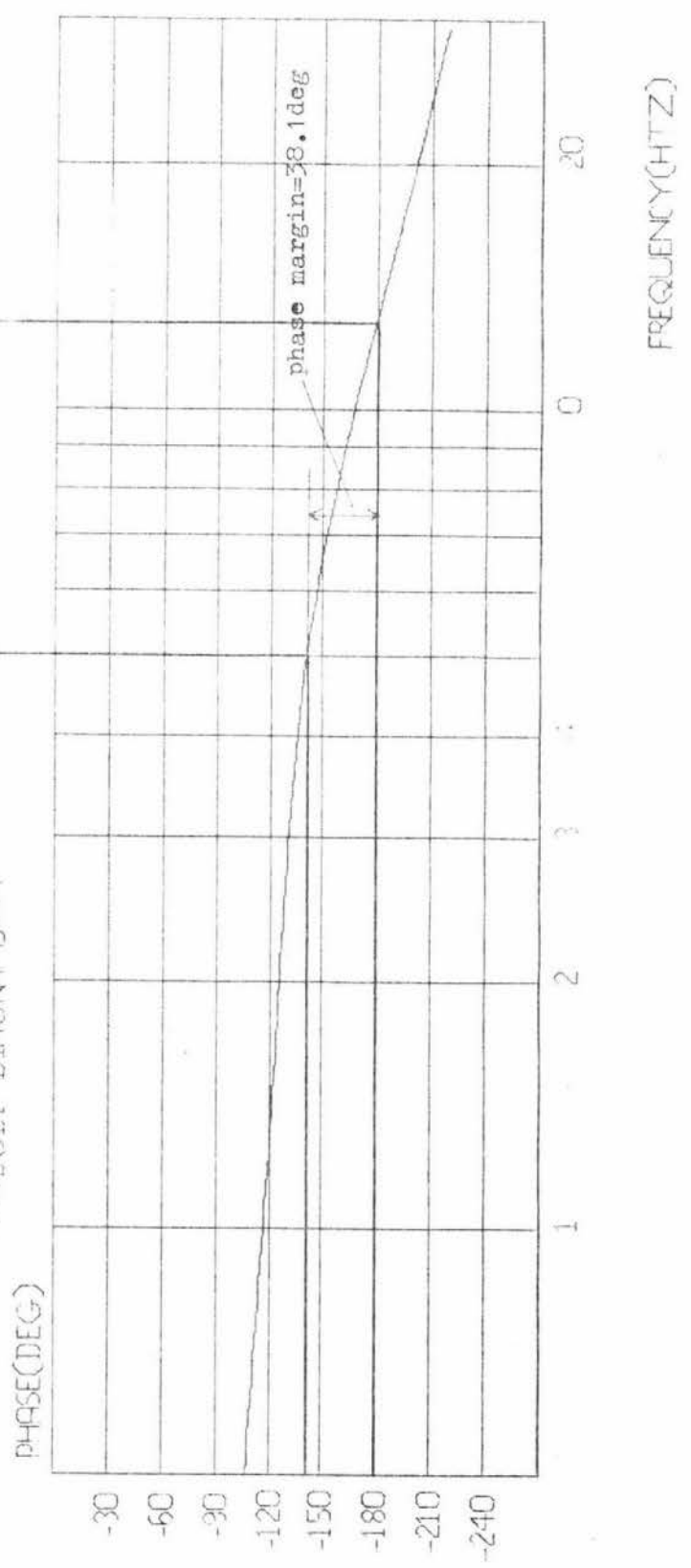


FIG. 19

From Fig.19 representing the Bode diagrams of the O/P-loop system, gain and phase margins are found to be the same as calculated from Fig.18.

The above information justifies that the compensation network has yielded a satisfactory control system with fast response, no oscillation, negligible delay time, negligible rise time and virtually no offset.

Often, better results can be achieved by trial-and-error adjustment of T and β . Further phase lead can be introduced by using more than one network if necessary.

4.2 DESIGN EXAMPLE USING ROOT-LOCUS METHOD

Consider the control system shown in Fig. 4.3. The feed forward transfer function is

$$G(s) = \frac{4}{s(s + .5)}$$

The root-locus plot for this system is shown in Fig.20. The closed-loop transfer function becomes

$$\begin{aligned} \frac{C(s)}{R(s)} &= \frac{4}{s^2 + .5s + 4} \\ &= \frac{4}{(s + .25 + j 1.98)(s + .25 - j 1.98)} \end{aligned}$$

The closed-loop poles are located at

$$s = - 0.25 \pm j 1.98$$

The damping ratio is .125, the undamped natural frequency is 2 rad/sec. It is desired to make the damping ratio equal to .5 and to increase the undamped natural frequency to 5 rad/sec. An appropriate compensation will be designed to meet these performance specifications.

Since the specifications are given in terms of time-domain quantities, the root-locus approach to design is used.

The closed-loop time response to a unit-step input, shown in Fig.22, indicates that the system has an undesirable transient-response characteristic and it therefore needs some form of compensation in order to produce a more satisfactory system which

In the present example, the desired locations of the closed-loop poles are

$$s = - 2.5 \pm j 4.2$$

In some cases, after the root-loci of the original system have been obtained, the dominant closed-loop poles may be moved to the desired location by simple gain adjustment. This is, however, not the case for the present system. Therefore a lead compensator is inserted in the feed forward path.

A general procedure for determining the lead compensator is as follows:

First, find the sum of the angles at the desired location of one of the dominant closed-loop poles with the O/P-loop poles and zeros of the original system, and determine the necessary angle ϕ to be added so that the total sum of the angles is equal to $\pm 180^\circ(2K+1)$. The lead network must contribute this angle. (If the angle is quite large, then two or more lead networks may be needed.)

If the original system has the open-loop transfer function $G(s)$, then the compensated system will have the open-loop transfer function

$$G_1(s) = \alpha \frac{Ts + 1}{\alpha Ts + 1} \cdot K_c G(s)$$

where the first term on the right-hand side corresponds to the lead network, the second term K_c is the gain of the amplifier, and the last term $G(s)$ is the original open-loop transfer function. Note that there are many possible values for T that will yield the necessary angle contribution at the desired closed-loop poles.

The next step is to determine the locations of the pole and zero of the lead network, in other words the value of T . In choosing the value of T , a procedure is introduced to obtain the largest possible value for α so that the additional gain required for the amplifier is as small as possible. First, a horizontal line is drawn passing through point P , the desired location for

one of the dominant closed-loop poles. This is shown as line PA in Fig. 4.4. Also a line is drawn connecting point P and the origin. The angle between the lines PA and PO is dissected as shown in Fig.4.4.

The intersection of PC and PD with the negative real axis give the necessary location for the pole and zero of the lead network. The compensator thus designed will make point P on the root-locus of the compensated system. The open-loop gain is determined by means of the magnitude condition.

In the present system, the angle of $G(s)$ at the desired closed-loop pole is

$$\angle \frac{4}{s(s + .5)} \Big|_{s = -2.5 + j 4.2} = 235^\circ$$

Thus for the root-locus to go through the desired closed-loop pole, the lead network must contribute $\phi = 55^\circ$ at this point. By following the foregoing procedure, the pole and zero of the lead network, as shown in Fig.20 are determined to be

$$\text{pole at } s = -9.4, \quad \text{zero at } s = -2.6$$

The lead compensator consisting of the lead network and an amplifier has the transfer function $G_c(s) = \frac{(s + 2.6)}{(s + 9.4)} \cdot K_c$

Thus the open-loop transfer function of the compensated system becomes

$$\begin{aligned} G_c(s) G(s) &= \frac{(s + 2.6)}{(s + 9.4)} K_c \cdot \frac{4}{s(s + .5)} \\ &= \frac{K^1 (s + 2.6)}{s(s + .5)(s + 9.4)} \quad \text{where } K^1 = 4 K_c \end{aligned}$$

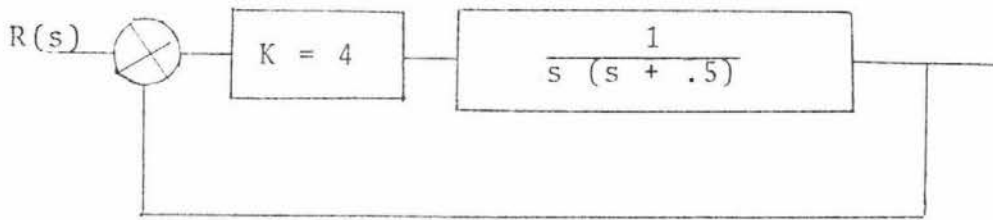


Fig. 4.3 Control system

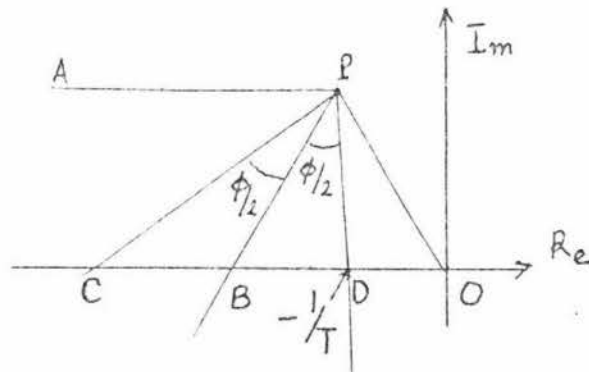


Fig. 4.4 Determination of the pole and zero of a lead network.

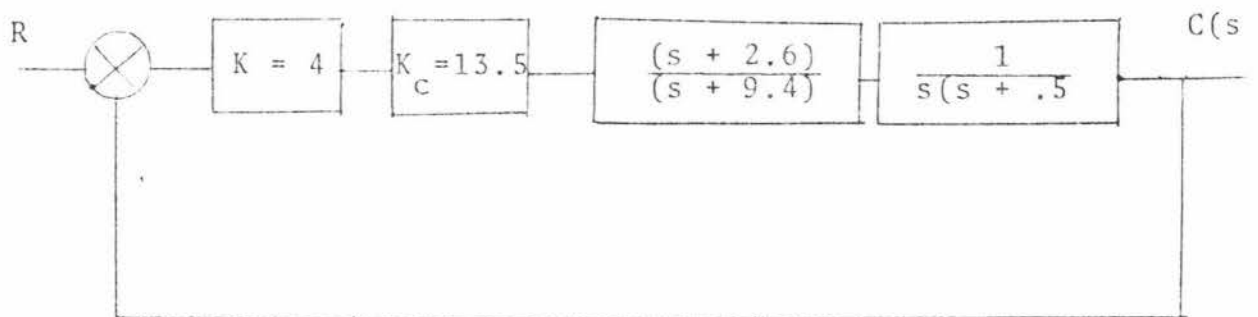


Fig. 4.5 The Compensated System.

ROOT-LOCUS

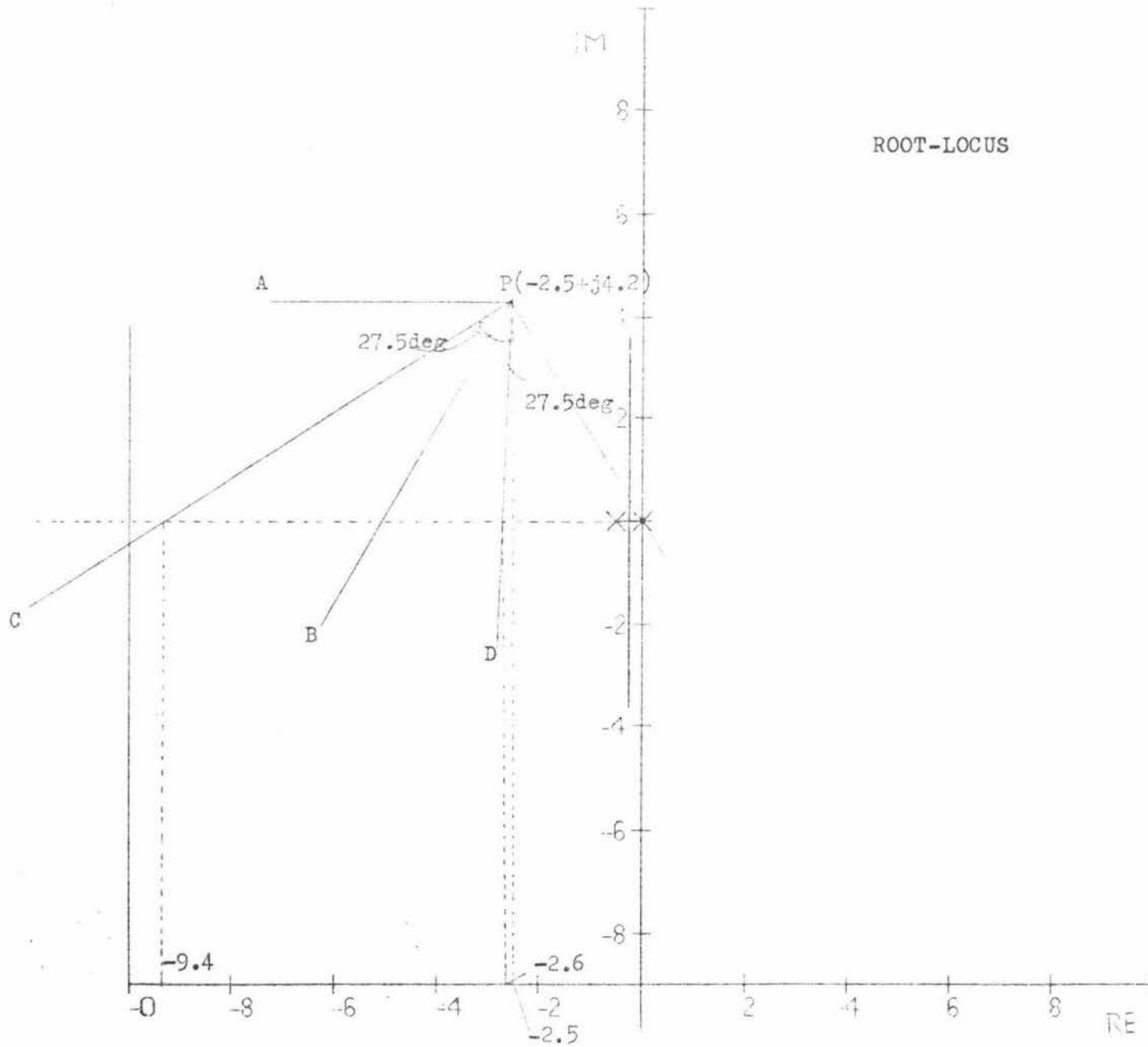


FIG. 20

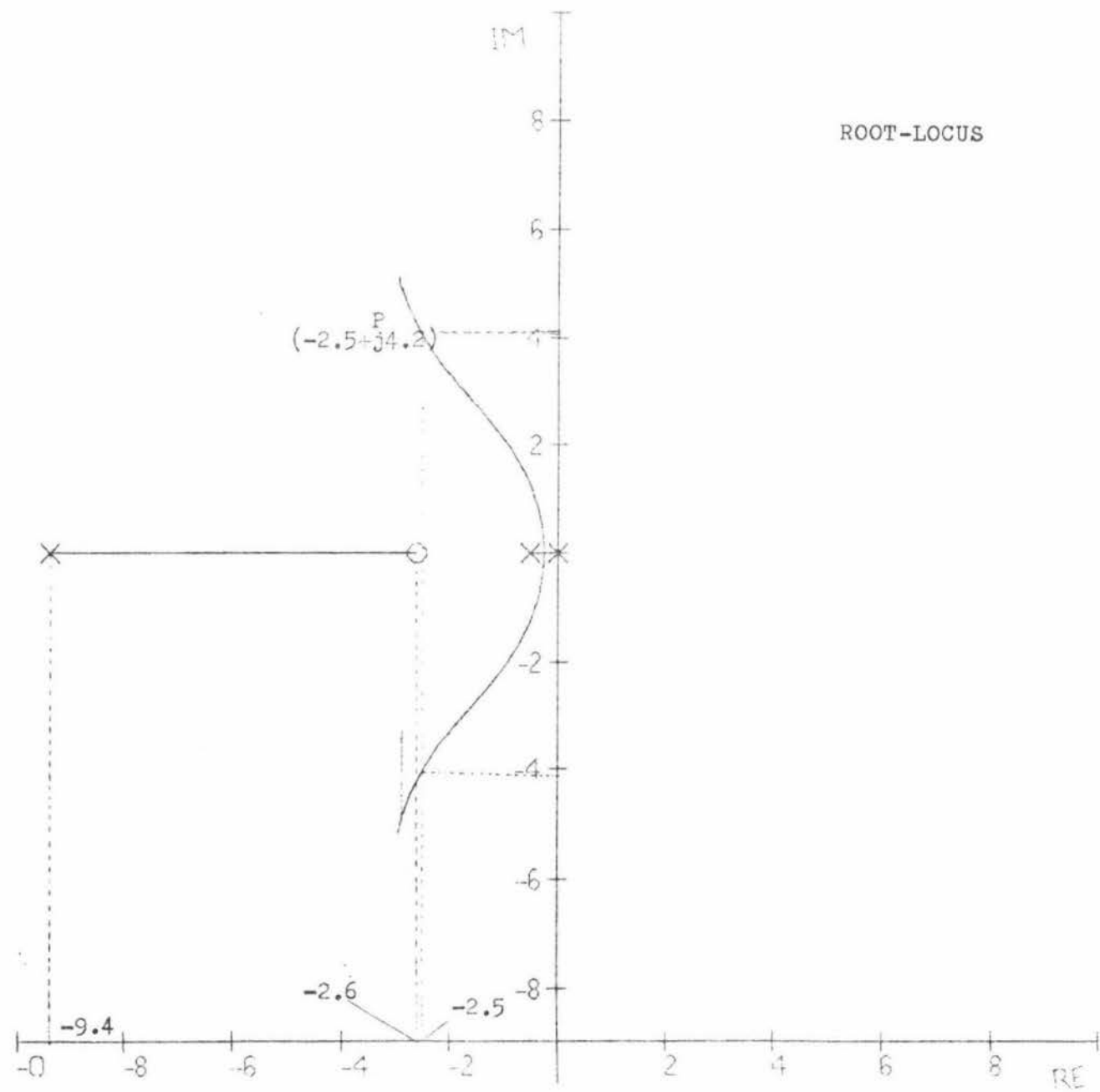


FIG. 21

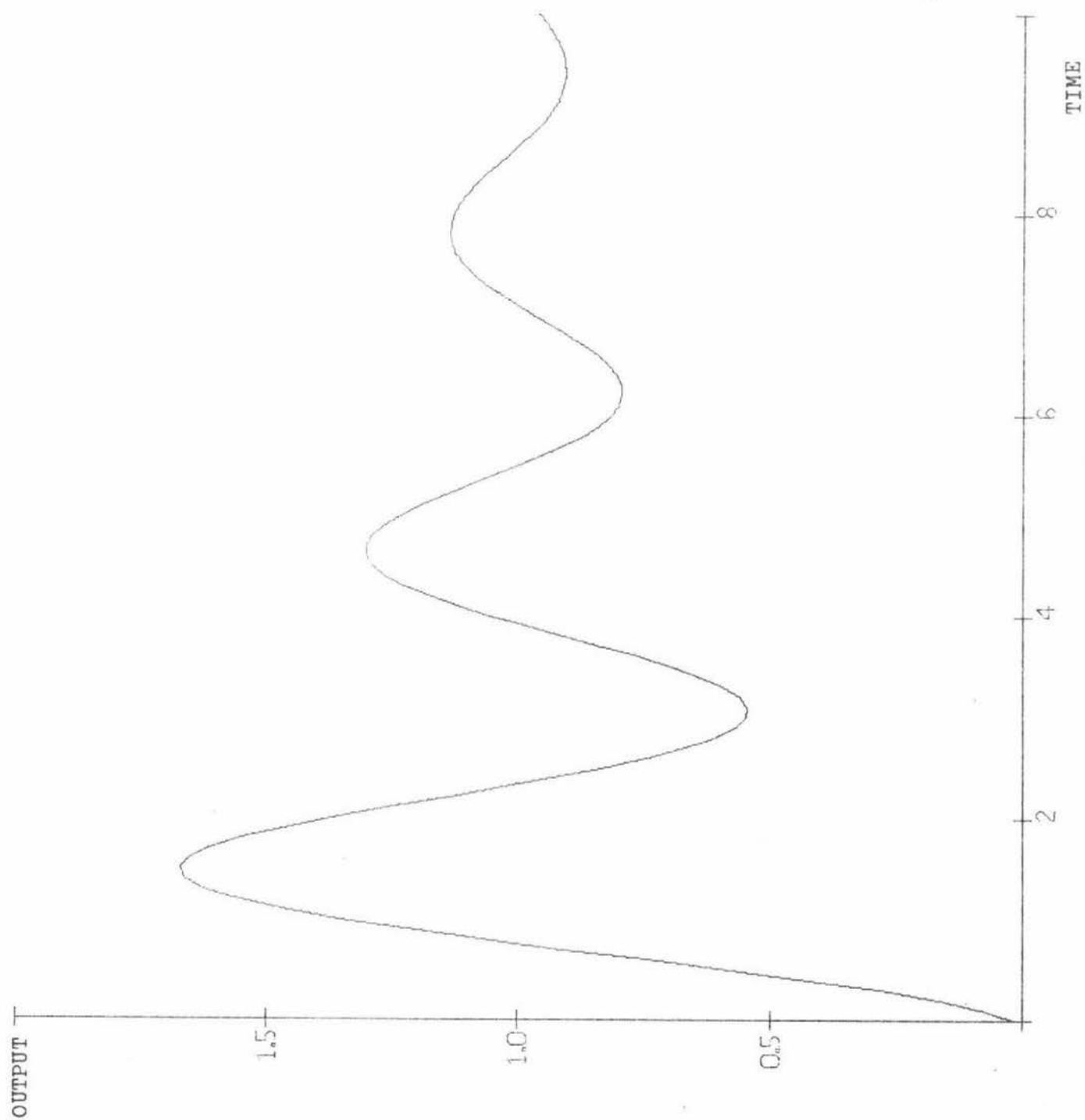


FIG. 22

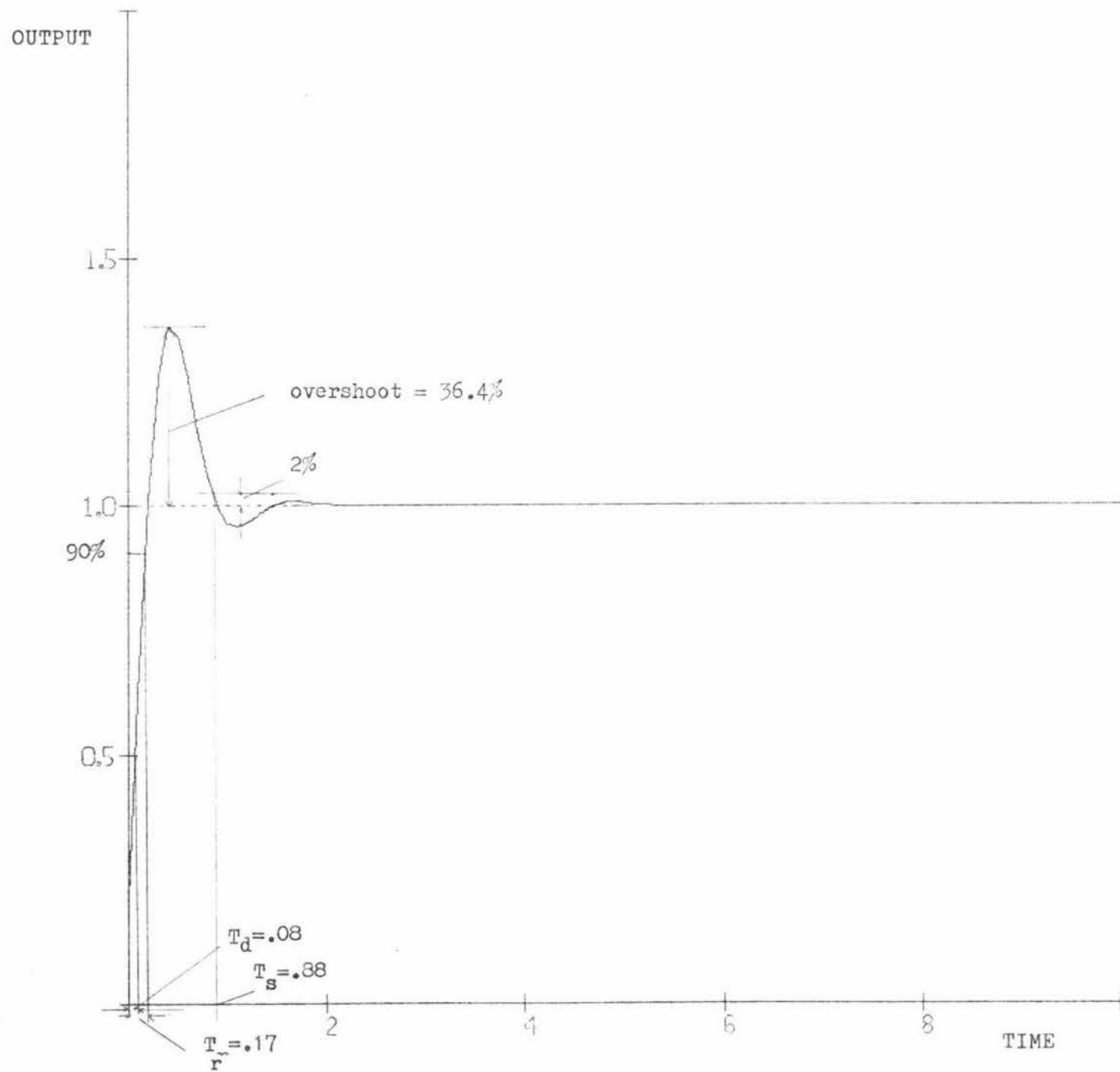


FIG. 23

The gain K is evaluated from the magnitude condition as follows

$$\left| \frac{K^1 (s + 2.6)}{s(s + .5)(s + 9.4)} \right|_{s = -2.5 + j 4.2} = 1$$

or $K \# 54$

It follows that

$$G_c(s) \cdot G(s) = \frac{54 (s + 2.6)}{s(s + .5)(s + 9.4)}$$

The gain constant K_c of the amplifier is

$$K_c = \frac{K^1}{4} = 13.5$$

The compensated system is shown in Fig. 4.5. Fig. 21 shows that the root-locus of the compensated system passes through the desired closed-loop poles $s = -2.5 \pm j4.2$.

The time response of the closed-loop system to a unit-step input is shown in Fig. 23 where the quantities measuring the performance of the system may be obtained, as follows:

- overshoot = 36.4%
- delay time = .008secs
- rise time = .17secs
- settling time = .88secs

It is concluded that the present design is satisfactory since it yields a very fast, almost immediate, response, negligible delay, rise and settling time and virtually no offset. The compensated system is shown in Fig. 4.5.

4.3 DESIGN EXAMPLE USING FREQUENCY-RESPONSE APPROACH

Consider the system shown in Fig. 4.6. The open-loop transfer function is

$$G(s) = \frac{4K}{s(s + 2)}$$

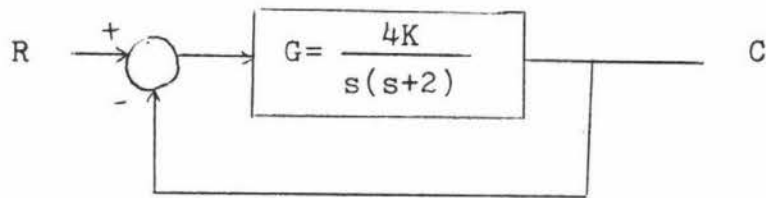


Fig. 4.6 Uncompensated system

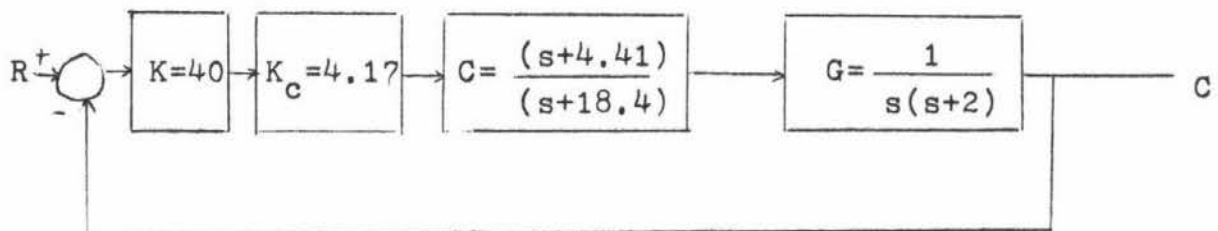


Fig. 4.7 Compensated system

It is desired to find a compensator for the system so that the static velocity error coefficient K_1 is 20sec^{-1} , the phase margin is at least 50° , and the gain margin is at least 10db. In the present example, the phase and gain margins have been specified. Bode diagrams will therefore be employed.

The first step in the design is to adjust the gain K to meet the steady-state performance specification or to provide the required static velocity error coefficient.

Since this coefficient is given as 20sec^{-1} ,

$$K_U = \lim_{s \rightarrow 0} sG(s) = \lim_{s \rightarrow 0} \frac{s4K}{s(s+2)} = 2K = 20$$

$$\text{or } K = 10$$

With $K = 10$, the system of Fig. 4.8 satisfies the steady-state requirement.

Next plot the diagrams of

$$G(j\omega) = \frac{40}{j\omega(j\omega + 2)}$$

Fig 24 shows the magnitude and phase angle curves of $G(j\omega)$. From this plot, the phase and gain margins of the system are found to 17° and db respectively. A phase margin of 17° implies that the system is quite oscillatory, as seen in Fig.26. Thus, satisfying the specification on the steady-state yields a poor transient-response performance. The specification calls for a phase margin of at least 50° . It is thus found that the additional phase lead necessary to satisfy the relative stability requirement is 33° .

In order to achieve a phase margin of 50° without decreasing the value of K , it is necessary to insert a suitable lead compensator into the system.

Noting that the addition of a lead compensator modifies the magnitude curve in the Bode diagrams, the gain crossover frequency will be shifted to the right. The increased phase lag of $G(j\omega)$,

due to this increase in the gain crossover frequency must be offset. Considering the shift of the gain crossover frequency, assume that ϕ_m , the maximum phase lead required, is approximately 38° . (This means that 5° has been added to compensate for the shift in the gain crossover frequency.)

Since

$$\sin \phi_m = \frac{1 - \alpha}{1 + \alpha}$$

$\phi_m = 38^\circ$ corresponds to $\alpha = .24$. Once the attenuation factor α has been determined on the basis of the required phase lead angle, the next step is to determine the corner frequencies $\omega = \frac{1}{T}$ & $\omega = \frac{1}{\alpha T}$ of the lead network. To do so, first note that the maximum phase lead angle ϕ_n occurs at the geometric mean of the two corner frequencies, or $\omega = \frac{1}{\sqrt{\alpha \cdot T}}$.

The amount of the modification in the magnitude curve at $\omega = \frac{1}{\sqrt{\alpha \cdot T}}$ is

$$\left| \frac{1 + j\omega\alpha T}{1 + j\omega T} \right|_{\omega = \frac{1}{\sqrt{\alpha \cdot T}}} = \left| \frac{1 + j \frac{1}{\sqrt{\alpha}}}{1 + j \alpha \frac{1}{\sqrt{\alpha}}} \right| = \sqrt{\alpha}$$

Note that

$$\frac{1}{\sqrt{\alpha}} = \frac{1}{\sqrt{.24}} = \frac{1}{.49} = 6.2 \text{ db}$$

and $G(j\omega) = 6.2 \text{ db}$ corresponds to $\omega = 9 \text{ rad/sec}$, as seen in Fig. 24. This frequency is selected to be the new gain crossover frequency ω_c . Noting that this frequency corresponds to $\frac{1}{\sqrt{\alpha \cdot T}}$, or $\omega_c = \frac{1}{\sqrt{\alpha \cdot T}}$,

$$\frac{1}{T} = \sqrt{\alpha} \omega_c = 4.41$$

and $\frac{1}{\alpha T} = \frac{\omega_c}{\alpha} = 18.4$ are obtained.

The lead network thus determined is

$$\frac{s + 4.41}{s + 18.4}$$

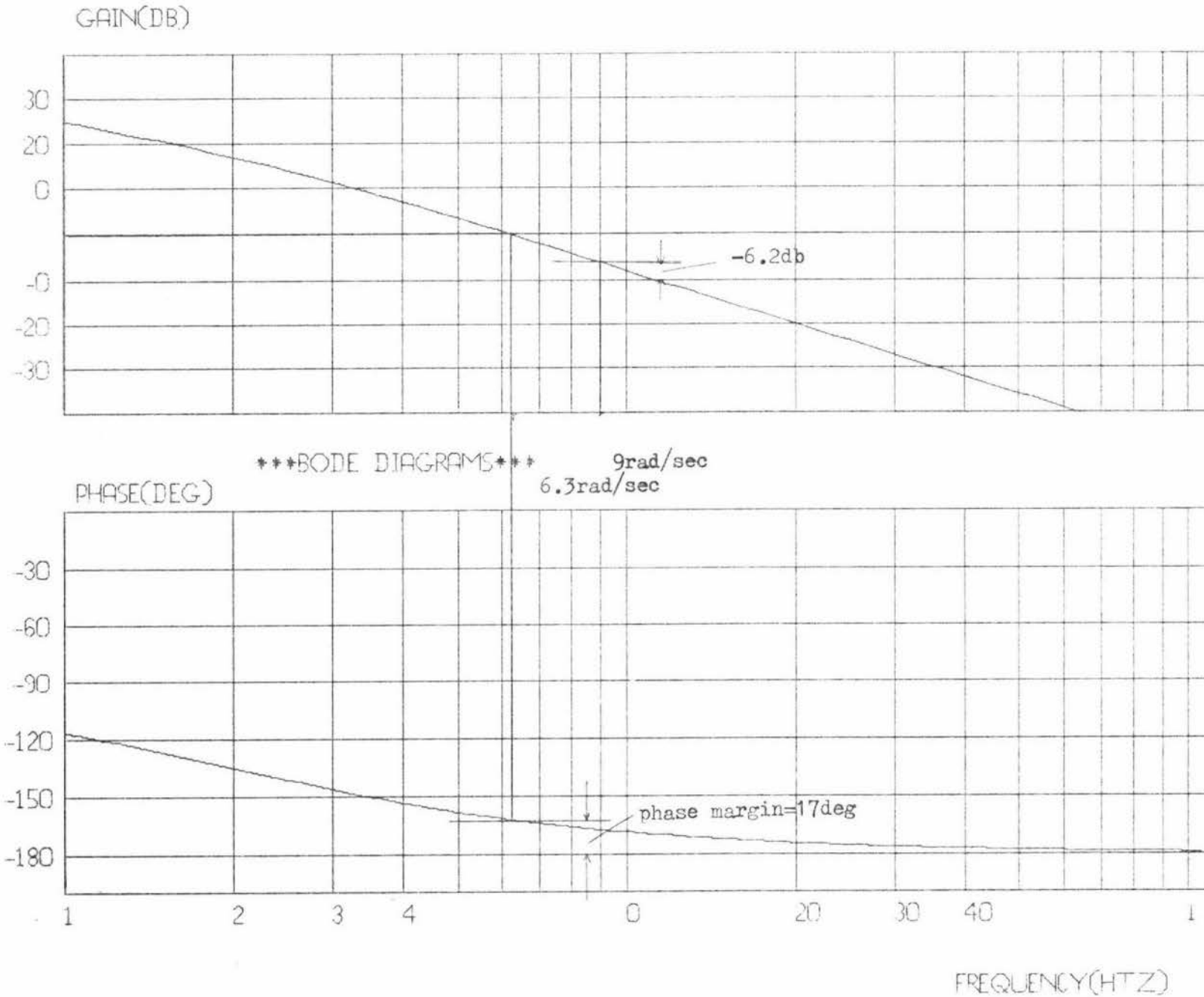


FIG. 24

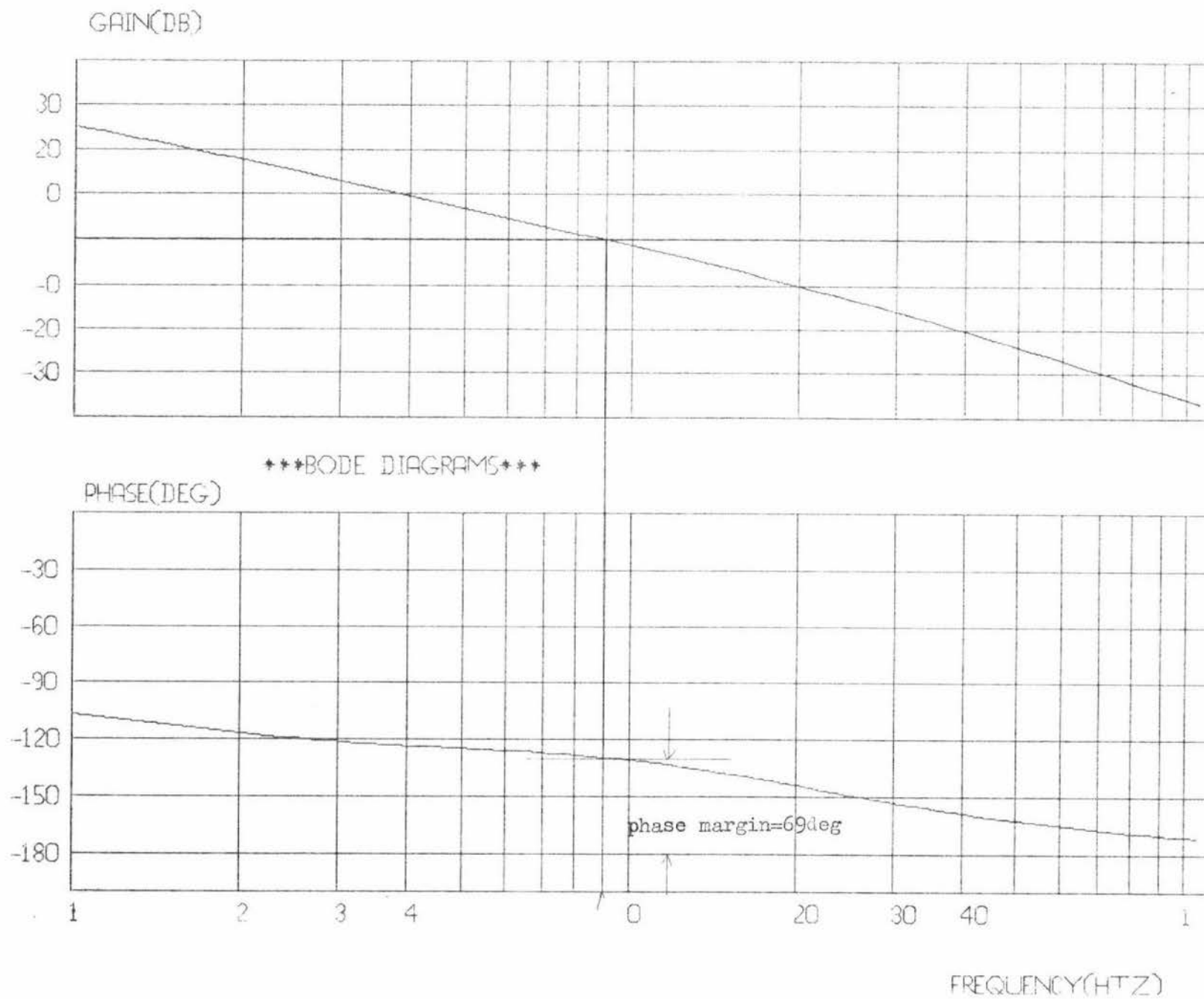


FIG. 25

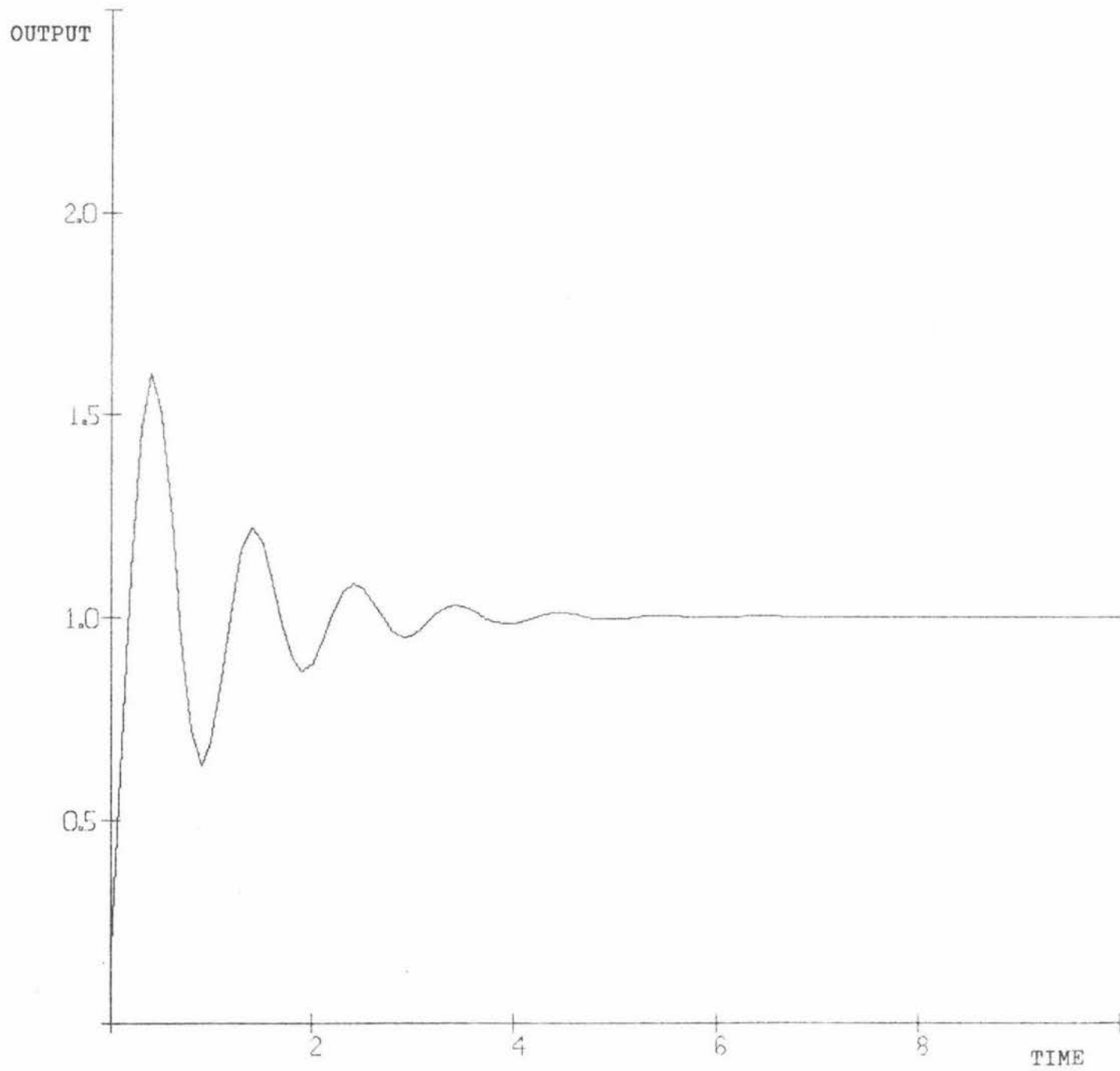


FIG. 26

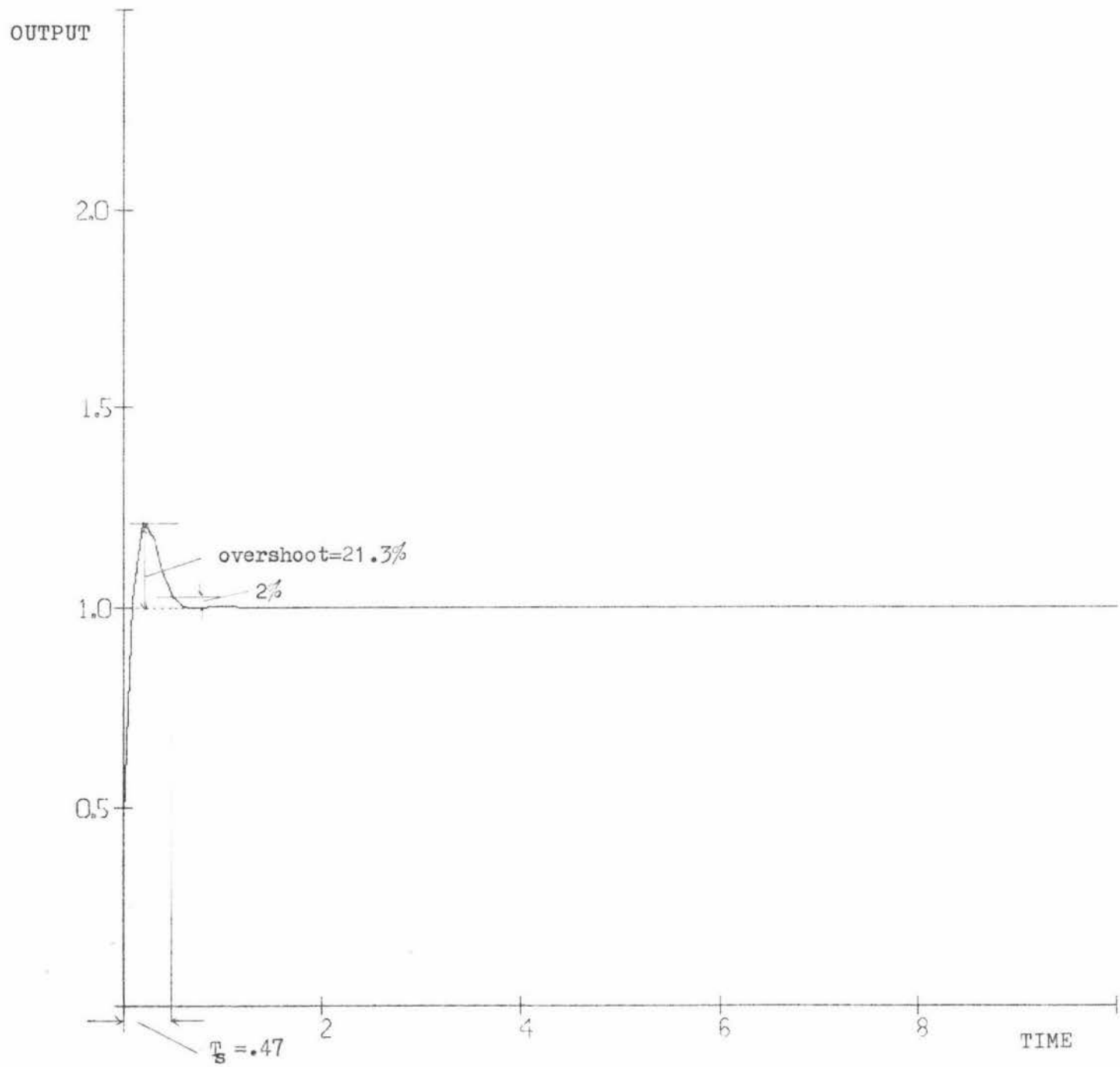


FIG.27

To compensate for the attenuation due to the lead network, the amplifier gain is increased by a factor of $\frac{1}{.24} = 4.17$. (If this were not done, the required static velocity error coefficient could not be realized.) Then the transfer function of the compensator which consists of the lead network and the amplifier becomes

$$G_c(s) = (4.17) \frac{s + 4.41}{s + 18.4}$$

The compensated system has the following open-loop transfer function

$$G_c(s) G(s) = (4.17) \left(\frac{s + 4.41}{s + 18.4} \right) \left(\frac{40}{s(s + 2)} \right)$$

Fig.25 shows the magnitude and phase-angle curves for the compensated system. The lead compensator causes the gain crossover frequency to increase from 6.3rad/s (Fig.24) to ω rad/sec. The increase in this frequency means an increase in bandwidth. This implies an increase in the speed of response which can be seen from Fig.27 showing the compensated system closed-loop time response to a unit-step input.

With an overshoot of 21.3%, zero delay time, negligible rise time and .47secs settling time, the system can be seen to have a much more satisfactory performance in comparison with the original system, shown in graph 26.

From Fig.25, the phase and gain margins are seen to be 69° and + ∞ db, respectively. The compensated system shown in Fig.4.7 therefore meets both the steady-state and the relative-stability requirements.

4.4 DISCUSSION

The design examples in this chapter illustrate the use of the computer-aided system design techniques. The **system** software has been developed which, having specified the control system configurations $G(s)$ and $H(s)$, permits varied

choice of $G_c(s)$ using trial-and-error analysis approaches.

The results from this chapter show how the typical design techniques described in chapter 2 are implemented in the control system design with the assistance of a digital computer. Different graphs representing the system performance in either time- or frequency-domain are exactly generated and analysed to obtain information essential for the compensation of the system deficiencies.

The trial-and-error method, although it is seen less direct and involving many trials, is useful in designing systems with no set performance specifications.

The root-locus approach is seen very convenient in designing systems with performance specifications given in terms of time-domain quantities, such as damping ratio, natural frequency, etc.

The Bode technique is seen very effective in producing systems with performance specifications given in terms of frequency domain quantities, such as gain and phase, bandwidth, etc...

Whatever design technique may be used, it is important to note that a unique system will not be yielded. In fact many systems may satisfy the given specifications. An optimal choice among the many possibilities may be made from such considerations as projected overall performance, cost, space and weight.

The computer system developed contributes a major part in the control system design. It helps the designer to avoid much of the numerical drudgery necessary for the checking of the system performance after each adjustment of the compensator parameters. The information about the overall-system performance is manually calculated from the computer-generated graphs. The present computer system therefore may be further developed to provide complete knowledge of the system performance, i.e. the gain and phase margins, overshoot, delay time, rise time, etc... may be obtained directly from the computer. Automatic design of the

compensation networks may also be incorporated to enhance the facility.

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CHAPTER FIVE

CONCLUSION AND SUGGESTIONS FOR FURTHER STUDY

The software for computer aided control system design facilitates the design via analysis or trial-and-error approaches. The control designer, at the Tektronix console, is able to specify certain designs and evaluate their effectiveness in terms of the various graphical analyses produced by the computer and displayed on the screen of the Tektronix. If performance specifications are given in terms of time-domain quantities, such as overshoot, rise time, etc..., the root-locus approach to design is very powerful to use. If performance specifications are given in terms of frequency-domain quantities, such as gain and phase margins, the Bode method is the most popular to use. However, if a certain value of the resonant peak is required, the Nichols chart technique is more convenient to use. The Nyquist technique provides rapid evaluation of stability of the control system, it is therefore used to supplement other methods. The time-response method is normally used for checking to see whether or not the designed system satisfies the requirements in the time-domain, though a satisfactory performance may be reached using this method.

Hence, for complete quantitative information of a system, both the differential-equation and the transfer-function representations are desirable. The transfer-function for ease in design, the differential equation for exact performance data on several important characteristics of the system.

The software development exercise has not yet been completed because of the limited project time. Many other features can be incorporated to further facilitate the design procedures and to make them more automatic. (Section 3.7.)

Although the classical approaches to control system design

have many outstanding advantages, they have numerous shortcomings. Empiricism, trial-and-error design, and a lack of a fundamental theory combine to make conventional control more an art than a science.⁵

The root-locus and frequency-response methods, which essentially consist of the loop-gain adjustment and the design of appropriate compensators, are quite useful but are limited to idealized and relatively simple control systems, such as single-input-single-output linear time-invariant systems. Such classical design approaches suffer from severe limitations and difficulties when applied to the design of multiple-input-multiple-output and time-varying systems. In other words, for the exact requirements of the more complex automatic control problems in modern technology, classical methods are entirely inadequate. This is where "modern control" enters the picture. Modern control theory utilizes time-domain system description (state variables) rather than transfer-function methods. Extensive use is made of digital computing. The methods of modern control have been developed in many instances by applied and pure mathematics, with the result that much of the written presentation is very formal and quite inaccessible to most control engineers. By applying modern control theory, the designer is able to start from a performance index, together with constraints imposed on the system, and to proceed to design a stable system by a completely analytical procedure. The advantage of design based upon such control theory is that it enables the designer to produce a control system which is optional with respect to the performance index considered.

In addition, the concept of adaptive control systems has also been introduced to cope with the problem where the plant is normally exposed to varying environments so that plant

parameters change from time to time.

However, it is important to note that if performance specifications are given in terms of the time-domain or frequency-domain quantities considered in this thesis, modern design techniques are not convenient, and the classical techniques are useful for the control system design.

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APPENDIX A

ERROR COEFFICIENTS

2.3.1. *Static error coefficients*i. Steady-state errors

Consider system shown in Fig. 2.13. The closed-loop T.F. is

$$\frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

The transfer-function between the actuating error signal and the input signal $r(f)$ is

$$\frac{E(s)}{R(s)} = 1 - \frac{C(s)H(s)}{R(s)} = \frac{1}{1 + G(s)H(s)}$$

where the actuating error $e(t)$ is the difference between the input signal and the feedback signal. The final value theorem provides a convenient way to find the steady-state performance of a stable system. Since $E(s)$ is

$$E(s) = \frac{1}{1 + G(s)H(s)} R(s)$$

The steady-state actuating error is

$$e_s = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} \frac{s R(s)}{1 + G(s)H(s)}$$

The static error coefficients defined in the following are figures of merit of control systems.

ii. Static position error coefficient k_p

The steady-state error of the system for a unit-step input is

$$\begin{aligned} e_{s_s} &= \lim_{s \rightarrow 0} \frac{s}{1 + G(s)H(s)} \quad \frac{1}{s} \\ &= \frac{1}{1 + G(0)H(0)} \end{aligned}$$

The static position error coefficient k_p is defined by

$$k_p = \lim_{s \rightarrow 0} G(s)H(s) = G(0)H(0)$$

Thus, the steady-state error in terms of the static position error coefficient k_p is given by

$$e_{s_s} = \frac{1}{1 + k_p}$$

iii. Static velocity error coefficient k_v

The steady-state error of the system with a unit-ramp input is given by

$$\begin{aligned} e_{s_s} &= \lim_{s \rightarrow 0} \frac{s}{1 + G(s)H(s)} \quad \frac{1}{s^2} \\ &= \lim_{s \rightarrow 0} \frac{1}{s G(s)H(s)} \end{aligned}$$

The static velocity error coefficient k_v is defined by

$$k_v = \lim_{s \rightarrow 0} sG(s)H(s)$$

Thus the steady-state error is

$$e_{s_s} = \frac{1}{k_v}$$

iv. Static acceleration error coefficient k_a

The steady-state error of the system with a unit-parabol input (acceleration input) which is defined by

$$r(t) = \frac{t^2}{2} \quad \text{for } t \geq 0$$

$$= 0 \quad \text{for } t < 0$$

is given by $e_{s_s} = \lim_{s \rightarrow 0} \frac{s}{1 + G(s)H(s)} \frac{1}{s^3}$

$$= \lim_{s \rightarrow 0} \frac{1}{s^2 G(s)H(s)}$$

The static acceleration error coefficient k_c is defined by the equation

$$k_a = \lim_{s \rightarrow 0} s^2 G(s)H(s)$$

The steady-state error is then

$$e_{s_s} = \frac{1}{k_a}$$

The error coefficients k_p , k_u , and k_a describe the ability of a system to reduce or eliminate steady-state error. Therefore they are indicative of the steady-state performance. The higher the coefficients, the smaller the steady-state error. Therefore, it is generally desirable to increase the error coefficients, while maintaining the transient response within an acceptable range.

2.3.2 *Dynamic error coefficients*

The dynamic errors provide some information about how the error varies with time. We shall now introduce dynamic error coefficients to describe the dynamic error and we shall limit our systems to unity-feedback ones. By dividing the numerator polynomial of $E(s)/R(s)$ by its denominator polynomial, $E(s)/R(s)$ can be expanded into a series in ascending powers of s as

follows $\frac{E(s)}{R(s)} = \frac{1}{1 + G(s)} = \frac{1}{k_1} + \frac{1}{k_2} s + \frac{1}{k_3} s^2 + \dots$

The coefficients k_1, k_2, k_3, \dots , are defined to be the dynamic error coefficients. Namely,

k_1 = dynamic position error coefficient

k_2 = dynamic velocity error coefficient

k_3 = dynamic acceleration error coefficient

An advantage of the dynamic error coefficients becomes clear when $E(s)$ is written on the following form:

$$E(s) = \frac{1}{k_1} R(s) + \frac{1}{k_2} sR(s) + \frac{1}{k_3} s^2 R(s) + \dots$$

The region of convergence of this series is the neighbourhood of $s = 0$. This corresponds to $t = \infty$ in the time domain. The corresponding time solution or the steady-state error is given, assuming all initial conditions are zero and neglecting impulses at $t = 0$, as follows

$$\lim_{t \rightarrow \infty} e(t) = \lim_{t \rightarrow \infty} \left[\frac{1}{k_1} r(t) + \frac{1}{k_2} \dot{r}(t) + \frac{1}{k_3} \ddot{r}(t) + \dots \right]$$

The steady-state error due to the input function and its derivatives can thus be given in terms of the dynamic error coefficients.

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APPENDIX B

PERFORMANCES SPECIFICATIONS

1. *Frequency-domain specifications*

Frequency-domain specifications are usually stated in the following terms:

a. gain-margin GM, a measure of relative stability, is defined as the log modulus (db) of the system when the phase shift is 180° , denoted positive for negative db.

On the Nyquist diagram, the gain margin is given directly by $-20\log_{10}M$ where M is the value of $G GH(j\omega)$ when the curve cuts the negative axis.

b. phase-margin PM, a measure of relative stability, is defined as 180° minus the phase angle at the frequency where the gain is unity (odb).

On the Nyquist diagram, the phase margin is found by drawing a circle of unity radius about the origin and drawing the radius from the centre to the point where the curve $GH(j\omega)$ cuts the unit circle.

The phase margin is then the angle between this radius and the negative $X(\omega)$ axis, positive below the axis and negative above. Gain and phase margins Bode diagrams and Nyquist diagrams are shown in Fig. B-1.

c. bandwidth BW, a measure of the speed of response of a system, is defined as that range of frequencies over which the magnitude ratio does not differ by more than -3db from its value at a specified frequency. For many feedback control systems this frequency is zero.

d. resonant peak M_p , a measure of relative stability, is the maximum value of the magnitude of the closed-loop frequency

response,

- e. resonant frequency ω_p , is the frequency at which M_p occurs. Bandwidth, resonant peak and resonant frequency are illustrated in Fig. B.2,

2. *Time-response specifications*

Time-response specifications are usually stated in terms of the unit-step function response. Typical specifications are:

- a. Overshoot is the maximum difference between the transient and steady-state solutions for a unit-step function input. It is a measure of relative stability and is often represented as a percentage of the final value of the output. The following four specifications are measures of the speed of response.
- b. Delay time T_d is often defined as the time required for the response to a unit-step function input to reach 50 percent of its final value.
- c. Rise-time T_r is customarily defined as the time required for the response to a unit-step function input to rise from 10 to 90 percent of its final value.
- d. Settling time T_s is defined as the time required for the response to a unit-step function input to reach and remain within a specified percentage (frequently 2 or 5 percent) of its final value.
- e. Subsidence ratio is the ratio of the amplitudes of successive cycles in a decaying oscillation.

The above specifications are illustrated in Fig. B.3.

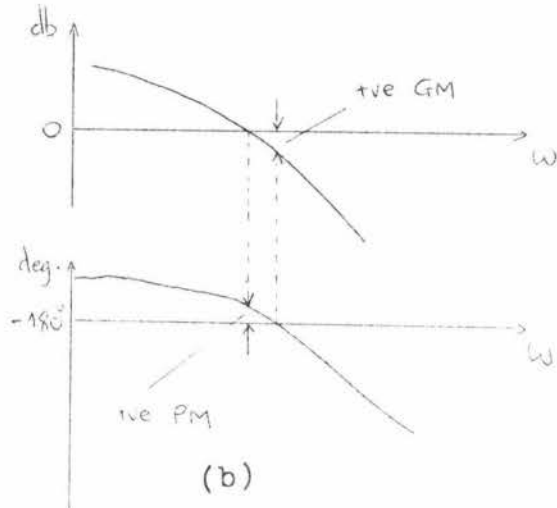
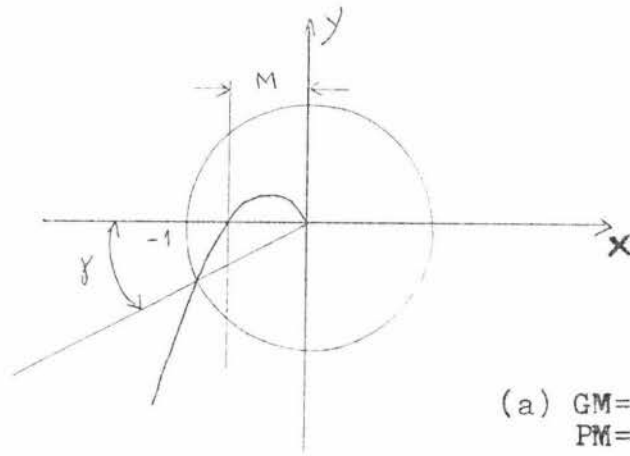


Fig. B.1 (a) Gain and phase margin on Nyquist diagram
 (b) Gain and phase margin on Bode diagram

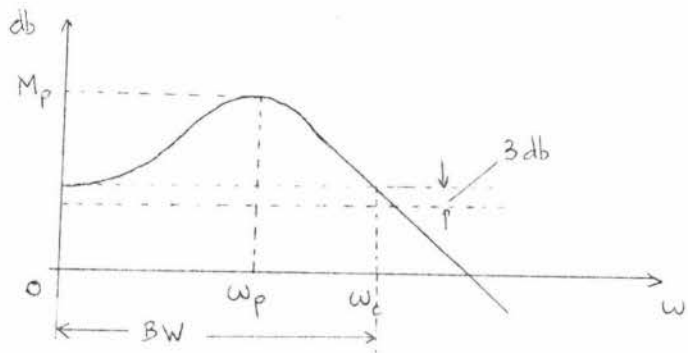


Fig. B.2 Bandwidth, resonant peak & resonant frequency

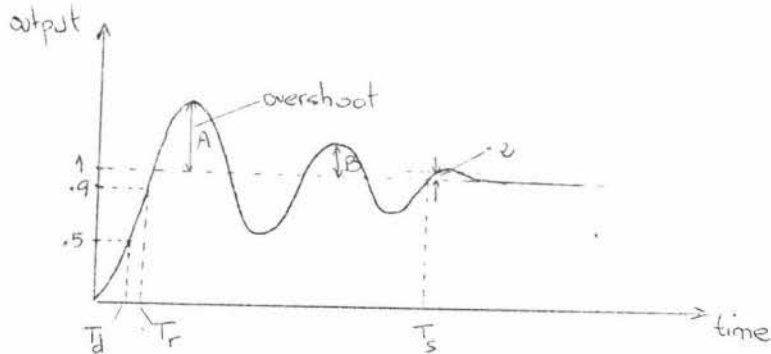


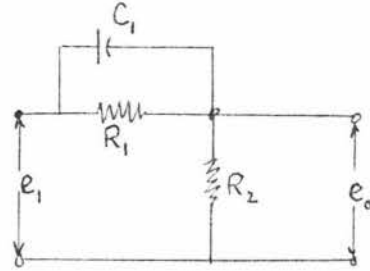
Fig. B.3 Illustration of performance specifications

APPENDIX C

COMPENSATION NETWORKS

1. *Lead compensation*

a. Lead network A schematic diagram of an electrical lead network is shown below



Define $R_1 C_1 = T$, $\frac{R_2}{R_1 + R_2} = \alpha < 1$

Then the transfer function for the above network may be derived as

$$\frac{E_o(s)}{E_i(s)} = \alpha \frac{Ts + 1}{\alpha Ts + 1} = \frac{s + \frac{1}{T}}{s + \frac{1}{\alpha T}}$$

b. Characteristics of lead network

Fig. C₁ shows the Nyquist diagram of

$$\alpha \frac{j\omega T + 1}{j\omega \alpha T + 1} \quad (0 < \alpha < 1)$$

For a given value of α , the angle between the positive real axis and the tangent line drawn from the origin to the semicircle gives the maximum phase lead angle, ϕ_m . Define the frequency at the tangent point to be ω_m . From Fig. C₁, the phase angle at $\omega = \omega_m$ is

$$\sin \phi_m = \frac{1 - \alpha}{1 + \alpha} \quad (C_1)$$

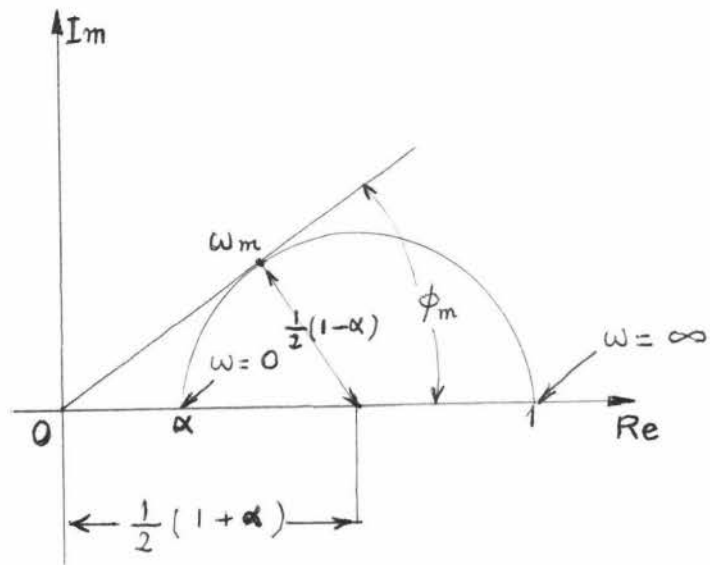


Fig.C1 Nyquist plot of $\alpha \cdot \frac{jT\omega + 1}{j\omega\alpha T + 1}$
 ($0 < \alpha < 1$)

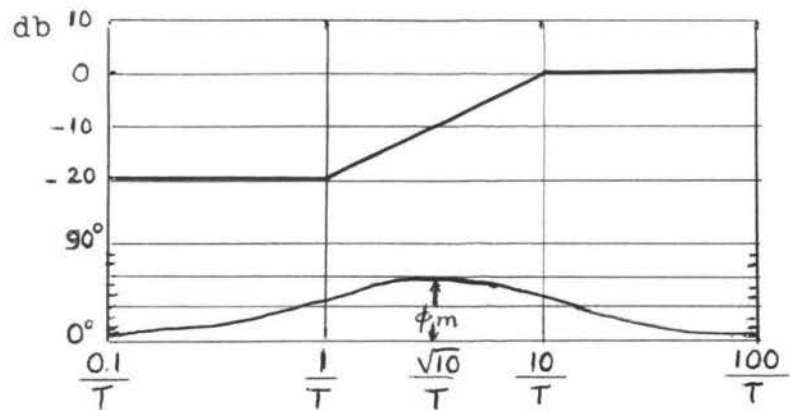


Fig.C2 Bode diagram of a lead network

Fig. C₂ shows the Bode diagram of a lead network when $\alpha = .1$. The corner frequencies for the lead network are $\omega = \frac{1}{T}$ and $\omega = \frac{1}{\alpha T}$. By examining Fig. C₂, it is seen that ω_m is the geometric mean of the two corner frequencies, or

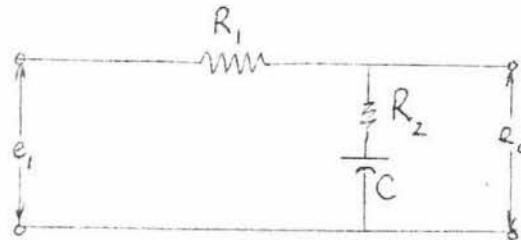
$$\log \omega_m = \frac{1}{2} \left[\log \frac{1}{T} + \log \frac{1}{\alpha T} \right]$$

Hence
$$\omega_m = \frac{1}{\alpha T}$$

As seen from Fig. C₂, the lead network is basically a high-pass filter, therefore an additional gain elsewhere is needed to increase the low frequency gain.

2. Lag compensation

a. Lag network A schematic diagram of an electrical lag network is shown below



Define $R_2 C = T$, $\frac{R_1 + R_2}{R_2} = \beta$ 1

Then the transfer function for the network may be derived as

$$\frac{E_o}{E_i}(s) = \frac{1}{\beta} \left(\frac{s + \frac{1}{T}}{s + \frac{1}{\alpha T}} \right)$$

b. Characteristics of lag network

Fig. C₃ shows the Bode diagram of a lag network when $\beta = 10$. The corner frequencies of the lag network are $\omega = \frac{1}{T}$ & $\omega = \frac{1}{\beta T}$. As seen from Fig. C₃, the lag network is essentially a

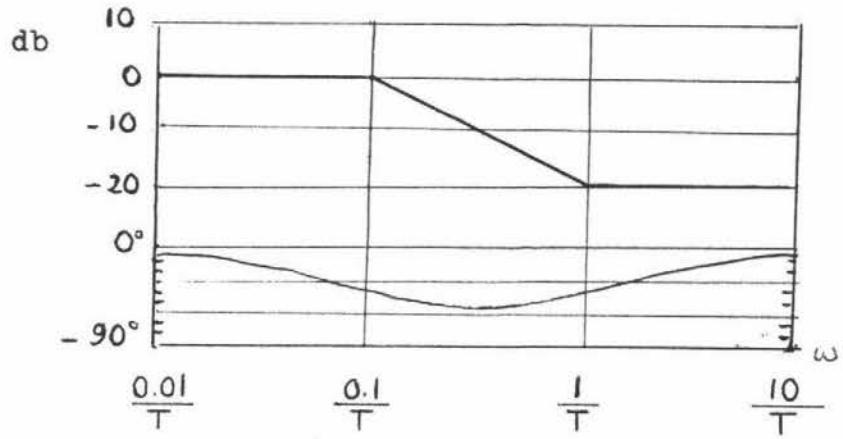


Fig.C3 Bode diagram of a lag network

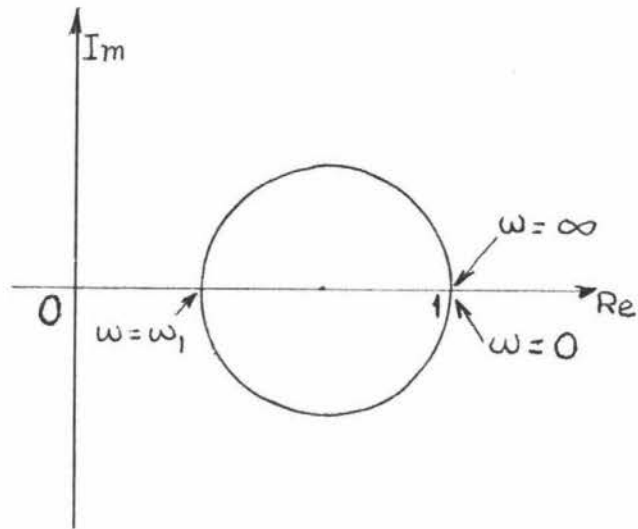


Fig.C4 Nyquist diagram of a lead-lag network

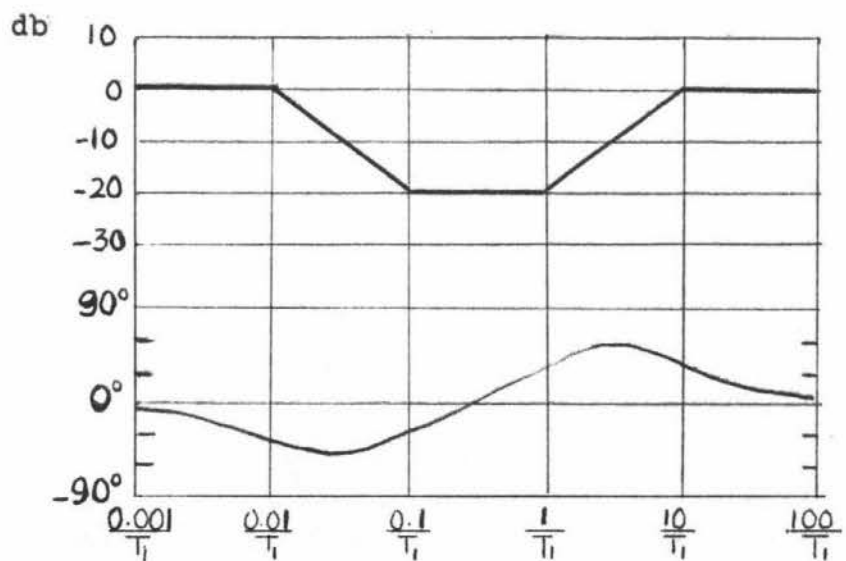


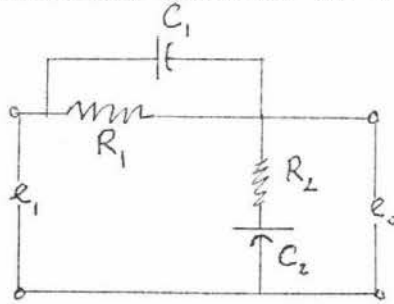
Fig.C5 Bode diagram of a lead-lag network

low-pass filter,

3. Lead-lag compensation

a. Lead-lag network

The electrical network of a lead-lag network is shown below:



The transfer function of this network is

$$\frac{E_o(s)}{E_i(s)} = \frac{(s + \frac{1}{T_1})(s + \frac{1}{T_2})}{(s + \frac{\beta}{T_1})(s + \frac{1}{\beta T_2})}$$

with $R_1, C_1, = T_1, \quad R_2, C_2, = T_2$

$$R_1, C_1, + R_2, C_2, + R_1, C_2 = \frac{T_1}{\beta} + \beta T_2 \quad (\beta > 1)$$

b. Characteristics of lead-lag network

Fig. C₄ shows the Nyquist diagram of a lead-lag network. It can be seen that for $0 < \omega < \omega_1$, the network acts as a lag network, while for $\omega_1 < \omega < \alpha$ it acts as a lead network.

The frequency at which the phase angle is zero is

$$\omega_1 = \frac{1}{\sqrt{T_1 T_2}}$$

Fig. C₅ shows the Bode diagram of a lead-lag network when $\beta = 10$ and $T_2 = 10T_1$.

APPENDIX D

CONSTITUENT PROCEDURES

The program consists of the following constituent procedures:

1. POLYADD which adds two polynomials
2. POLYMUL which multiplies two polynomials
3. REALPOLYZEROFINDE which finds all the zeros of a real polynomial
4. DIFEQNS which solves a system of first order differential equations. (The above procedures are system procedures⁸.)
5. I_1 which calculates the imaginary part of polynomial with order less than three, as ω varies.
6. R_1 which calculates the real part of polynomial with order less than three as ω varies.
7. I_2 which calculates the imaginary part of polynomial with order greater than or equal to three, as ω varies.
8. R_2 which calculates the real part of polynomial with order greater than or equal to three, as ω varies.
9. TIMEDOMAIN which generates the time-response of the system, either open- or closed-loop, to a unit-step input. This procedure provides flexible features, such as the interactive changeability of scales and ranges of the display.
10. BODE which generates the Bode diagrams of the system, either open- or closed-loop, to a unit-step input. Associated features such as scales and ranges of the display may be changed and grid lines also may be drawn.
11. NYQUIST which generates the Nyquist diagram of the system, either in open- or closed-loop form. Similar flexible

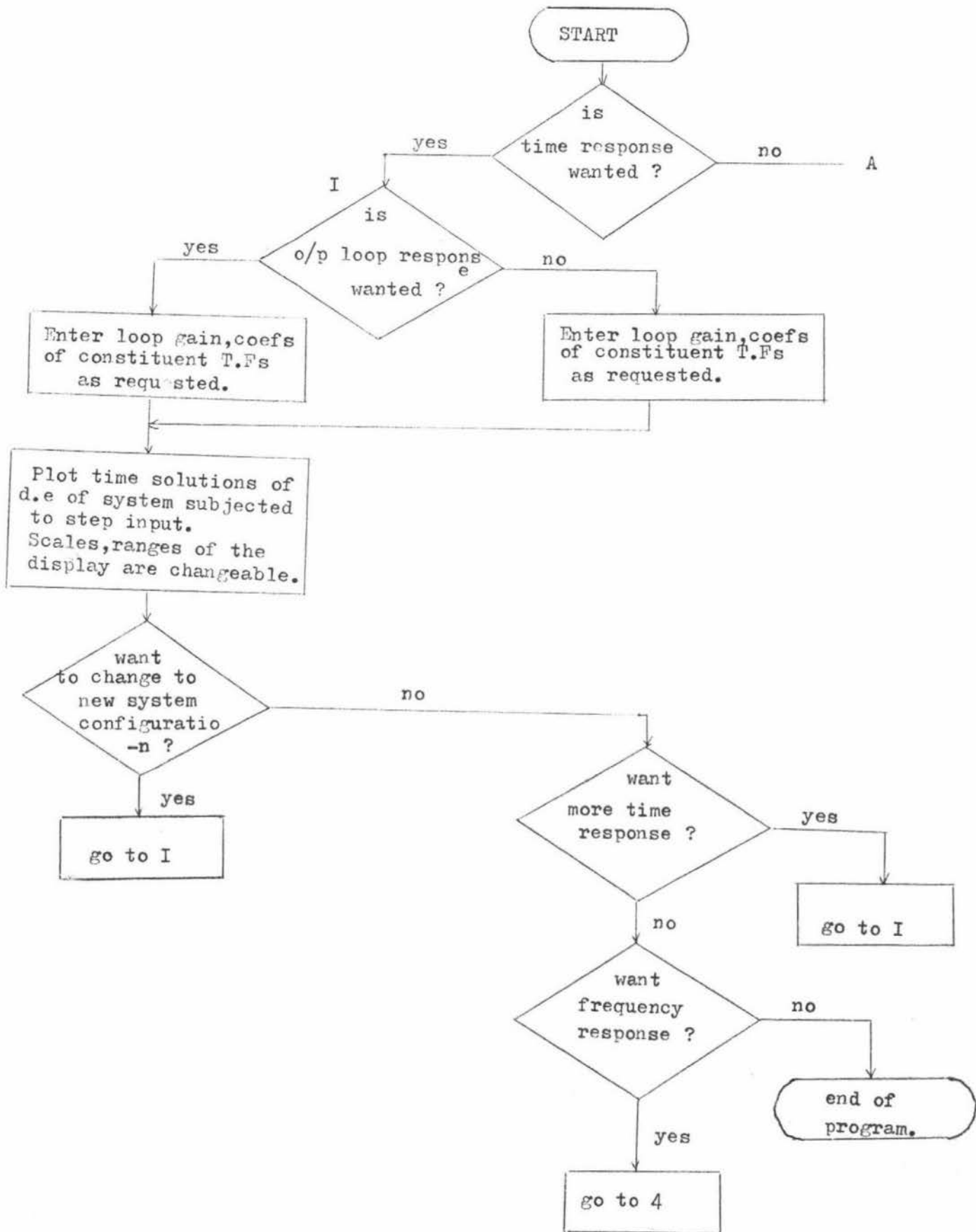
- features provided by BODE procedure are also available here.
-]2. NICHOLS which generates the Nichols chart plot of the open-loop system superimposed on the Nichols chart. This procedure provides similar features as in BODE procedure.
 -]3. RLOCUS which generates the root-locus of the system. Scales and ranges of the display and final value of the open-loop gain k are changeable.
 -]4. READCGH1 which manipulates the constituent transfer functions $C_1(s)$, $C_2(s)$, $G_1(s)$, $G_2(s)$, $H_1(s)$, $H_2(s)$ and the open-loop gain to produce the system open-loop transfer function in a general form $\frac{A(s)}{B(s)}$. The coefficients of the constituent transfer functions and the open-loop gain are entered from the keyboard as requested.
 -]5. READCGH2 which manipulates the constituent transfer functions $C_1(s)$, $C_2(s)$, $G_1(s)$, $G_2(s)$, $H_1(s)$, $H_2(s)$ and the open-loop gain to produce the system closed-loop transfer function in a general form $\frac{A(s)}{B(s)}$.

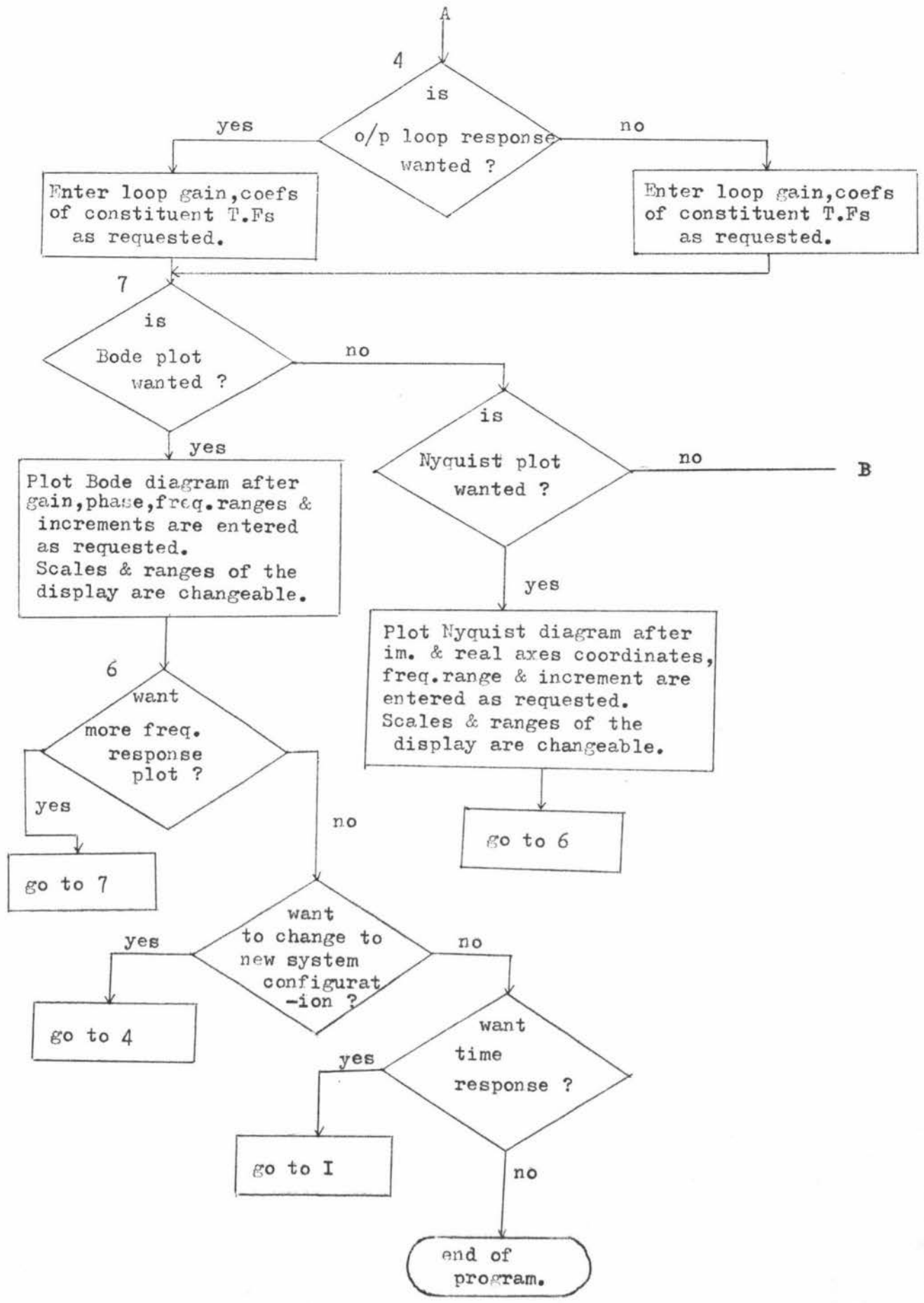
In addition, the plotting procedures used in the program are STARTP, DEVICE, YRANGE, XRANGE, EDGESR, FRAME, AT, LINETO, SYMSSEL, SYMSIZ, PUTSYM, WRTARY, TXT, FIXEDN, TXTSIZ, TXTDIR, SUBFRM, ORGRM, XAXIS, YAXIS, MODE, CHARMD and ENP.

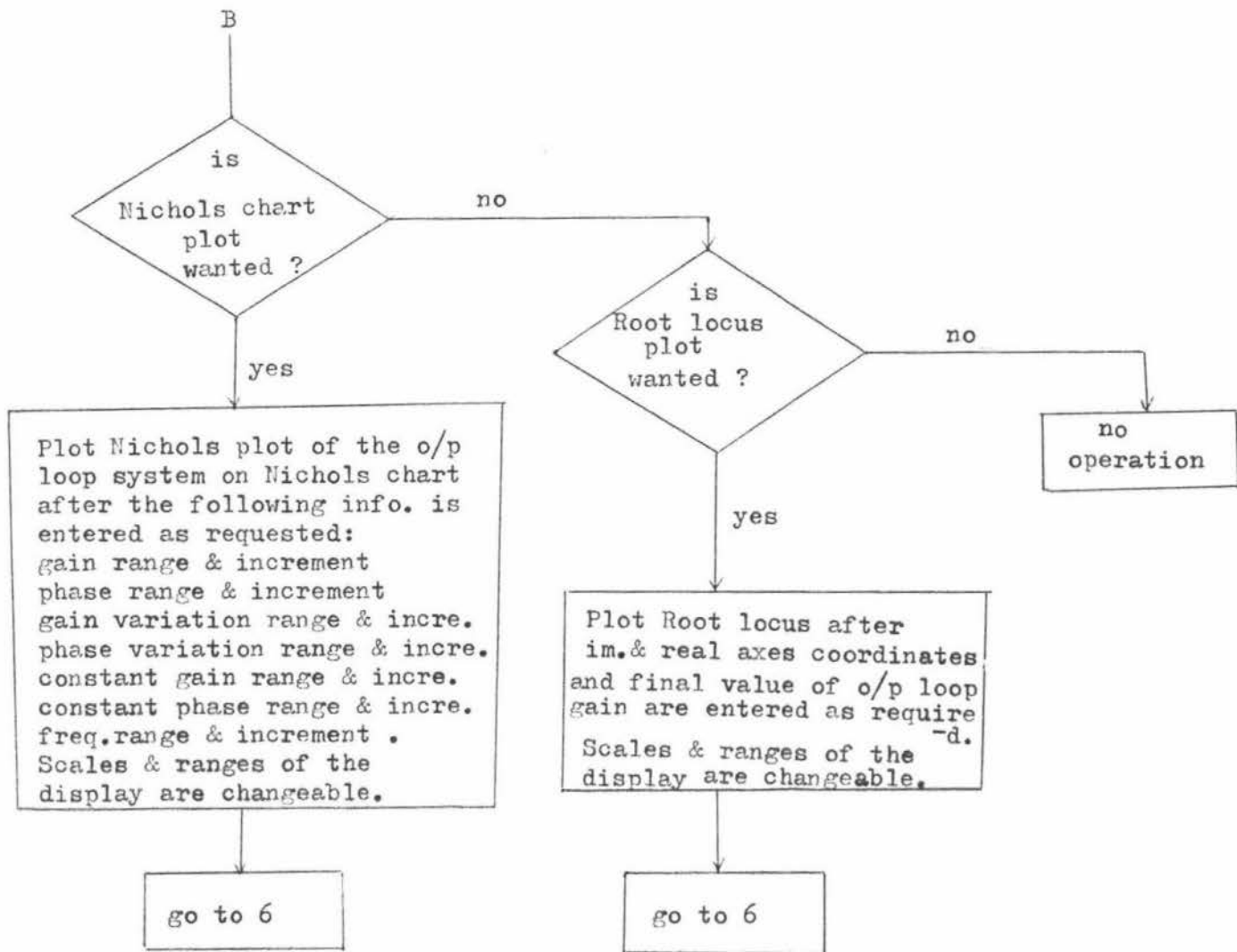
The functions and other details of these procedures may be obtained from the Massey plotter package manual⁸.

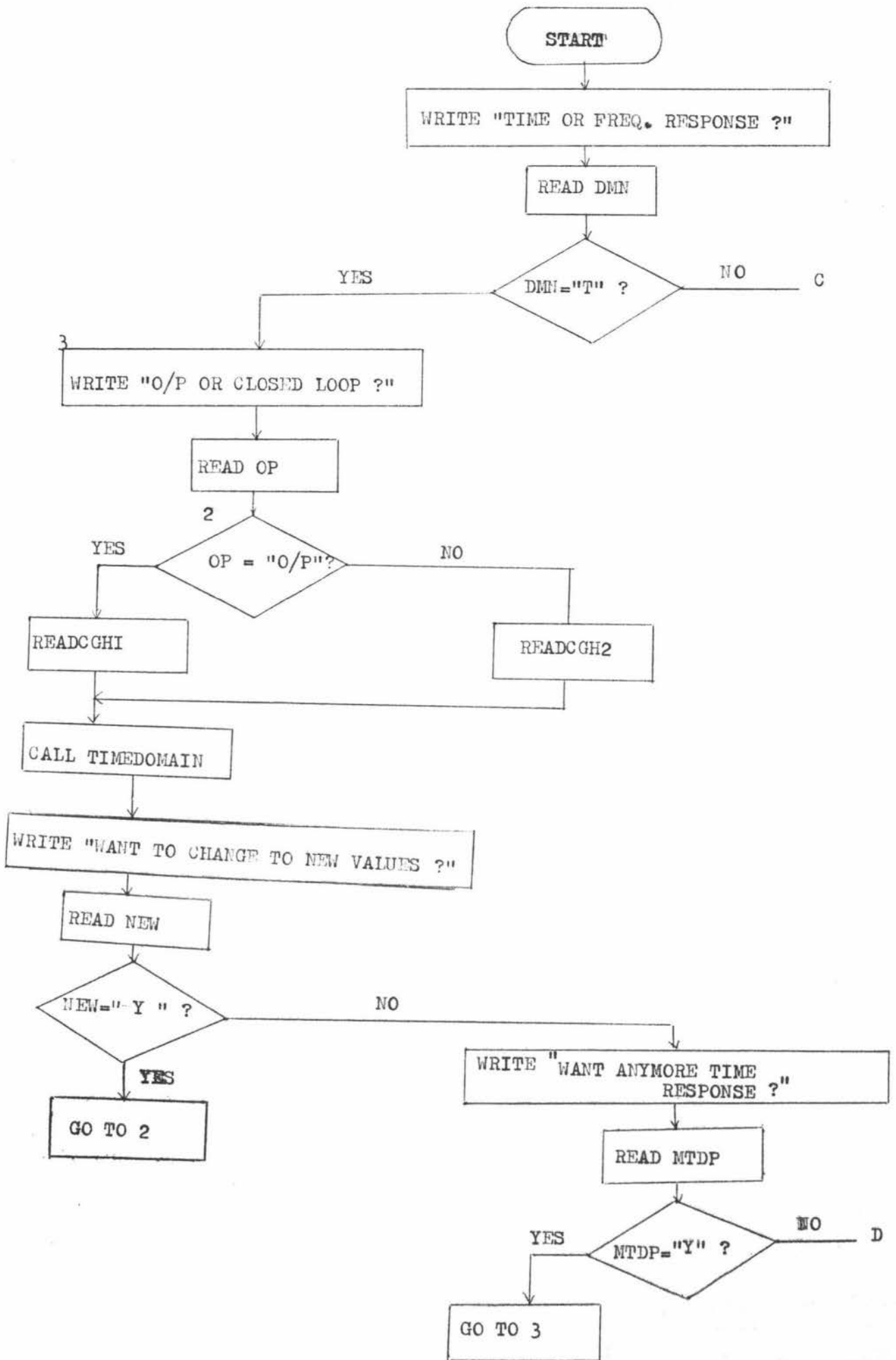
APPENDIX E

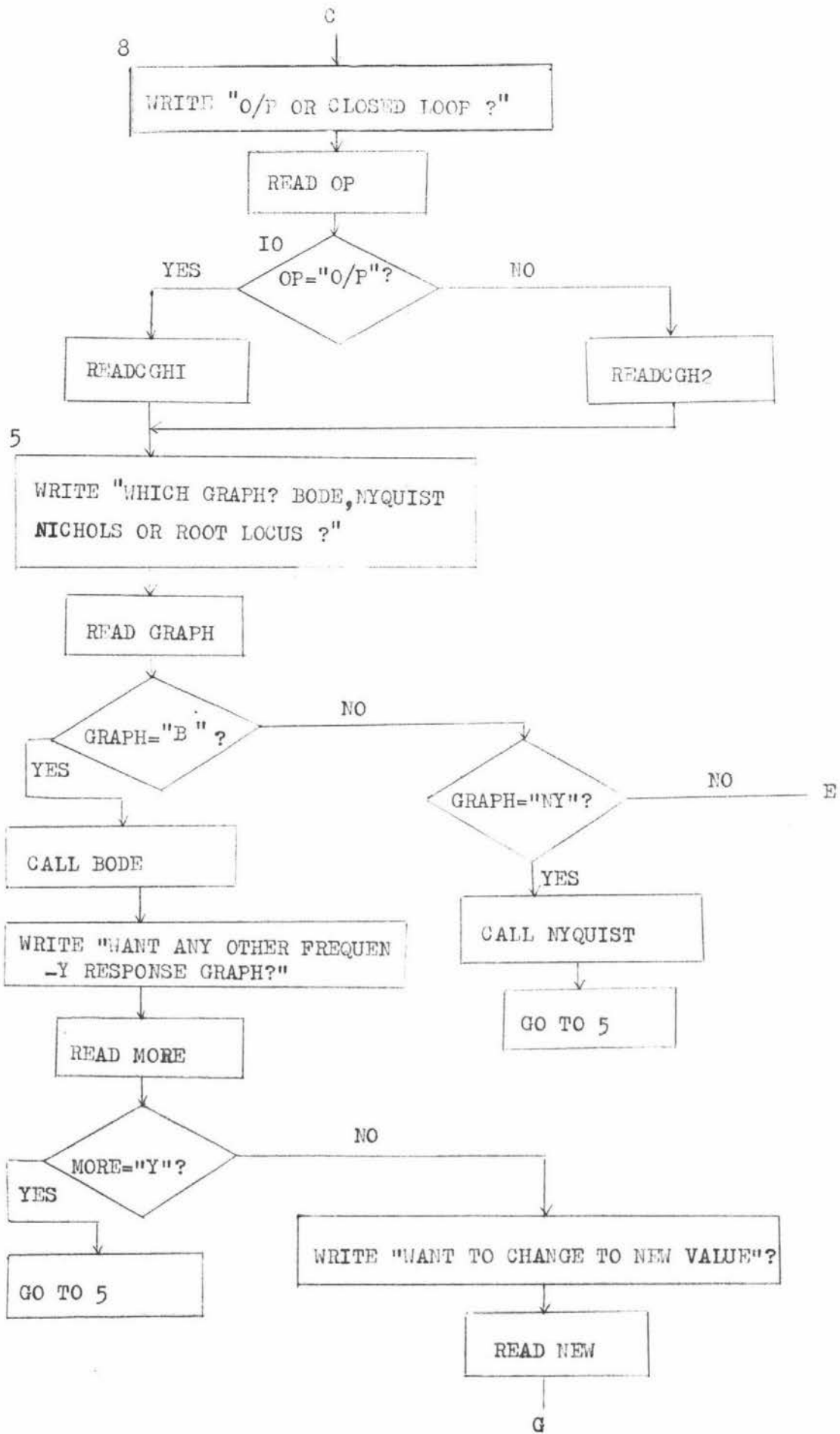
FLOWCHART

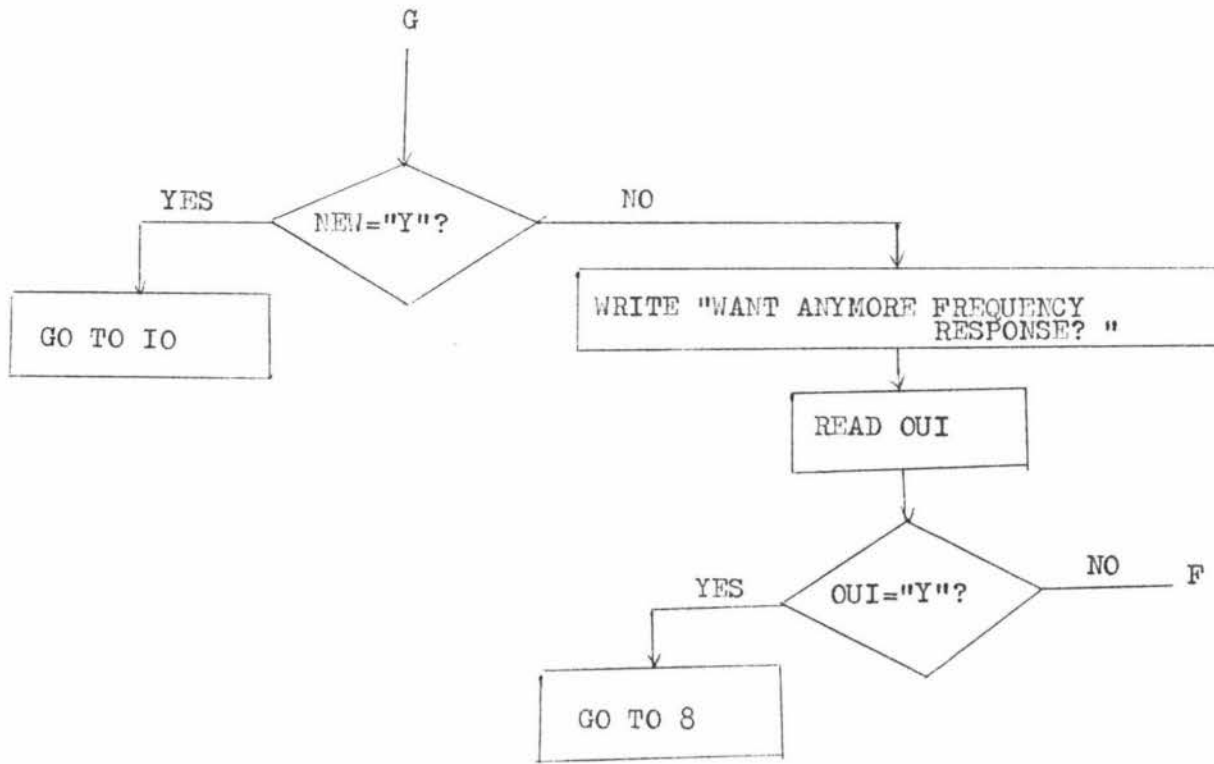


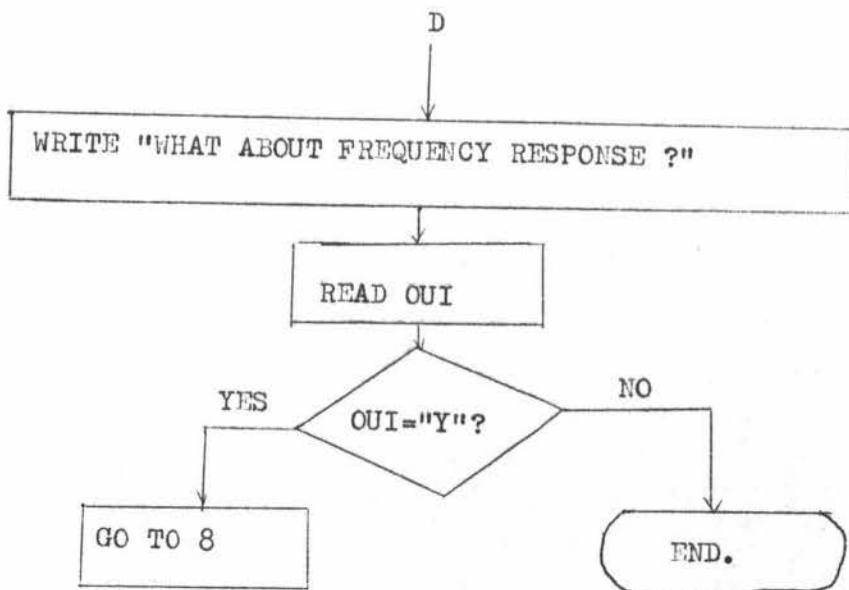
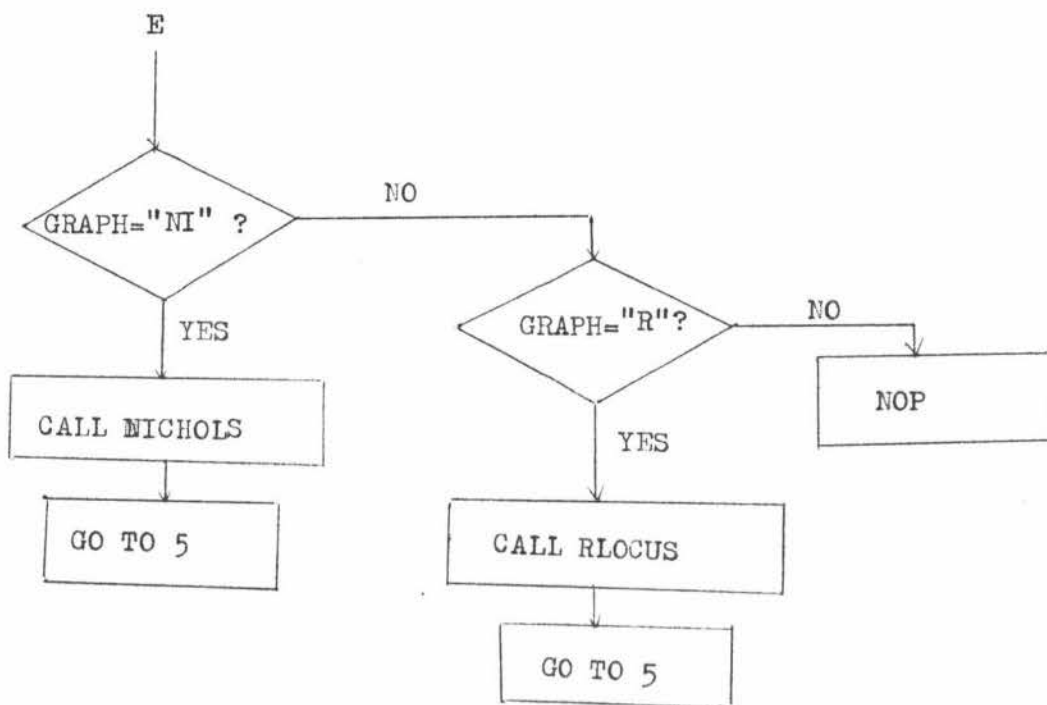
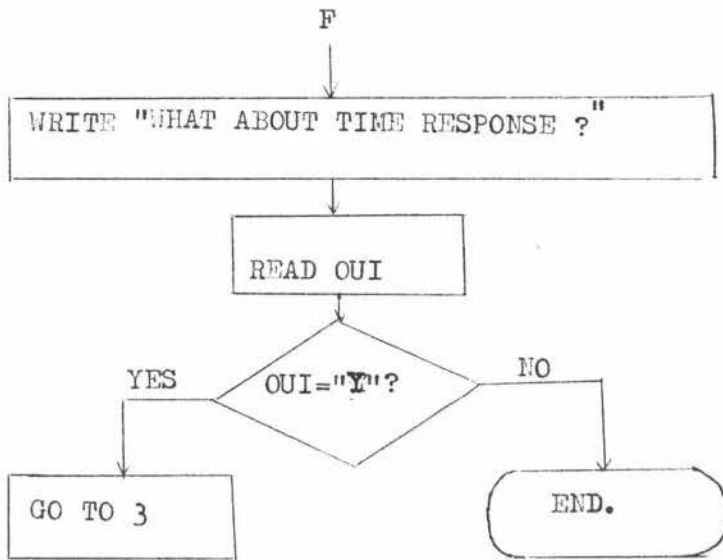












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100 BEGIN
150 $SET LIST
200 $RESET SINGLE
300 $INCLUDE "PLOT/IN/ALGOL"
400 $BIND POLYADD,POLYMUL FROM MATHLIB/=
500 REAL DC1,DC2,DG1,DG2,DH1,DP1,DP2,DA,DB,DPOLY1,DPOLY2;
600 REAL DCG1,DCG2,DPUL1,DPOL2,J,DH2;
700 REAL DMM,MTDP,QI,DKCG1;
800 ARRAY U,INB[0:0];
900 REAL GRAPH,OP,MORE,ANY,NEW,OUI,W;
1000 LABEL LAB5,LAB6,LAB7,LAB8,LAB9;
1100 PROCEDURE POLYMUL(N1DEG,CUEF1,N2DEG,CUEF2,NRDEG,COEFR);
1200 VALUE N1DEG,N2DEG;
1300 REAL N1DEG,N2DEG,NRDEG;
1400 ARRAY COEF1,COEF2,COEFR[0];
1500 EXTERNAL;
1600 PROCEDURE POLYADD(N1DEG,CUEF1,N2DEG,CUEF2,NRDEG,COEFR);
1700 VALUE N1DEG,N2DEG;
1800 REAL N1DEG,N2DEG,NRDEG;
1900 ARRAY COEF1,COEF2,COEFR[0];
2000 EXTERNAL;
2100 REAL PROCEDURE R1(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];FORWARD;
2200 REAL PROCEDURE I1(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];FORWARD;
2300 REAL PROCEDURE IC(W,Q,POLY);VALUE W,Q;REAL Q;REAL W;
2400 ARRAY POLY[0];FORWARD;
2500 REAL PROCEDURE RC(W,Q,POLY);VALUE W,Q;REAL Q;REAL W;
2600 ARRAY POLY[0];FORWARD;
2700 REAL PROCEDURE R2(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];FORWARD;
2800 REAL PROCEDURE I2(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];FORWARD;
2900 REAL PROCEDURE RI(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];
3000 CASE T OF
3100 BEGIN
3200 R1:=A[0];
3300 R1:=A[1];
3400 R1:=A[2]-A[0]*W**2;
3700 END;
3800 REAL PROCEDURE I1(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];
3900 CASE T OF
4000 BEGIN
4100 I1:=0;
4200 I1:=A[0]*W;
4300 I1:=A[1]*W;
4600 END;
4700 REAL PROCEDURE R(W,Q,POLY);VALUE W,Q;REAL Q;REAL W;ARRAY POLY[0];
4800 BEGIN
4900 REP:=REN;
5000 REP:=REN:=0;
5100 FOR J:=0 STEP 4 UNTIL Q DO REP:=REP+POLY[J]*W**(Q-J);
5200 FOR J:=2 STEP 4 UNTIL Q DO REN:=REN+POLY[J]*W**(Q-J);
5300 R:=REP-REN;
5400 END;
5500 REAL PROCEDURE IC(W,Q,POLY);VALUE W,Q;REAL Q;REAL W;ARRAY POLY[0];
5600 BEGIN
5700 REAL IMP,IMN;
5800 IMP:=IMN:=0;
5900 FOR J:=3 STEP 4 UNTIL Q DO IMP:=IMP+POLY[J]*W**(Q-J);
6000 FOR J:=1 STEP 4 UNTIL Q DO IMN:=IMN+POLY[J]*W**(Q-J);
6100 I:=IMP-IMN;

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6200      END;
6300 REAL PROCEDURE R2(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];
6400 CASE P MOD 4 OF
6500     BEGIN
6600     R2:=R(W,P,A);
6700     R2:=-I(W,P,A);
6800     R2:=-R(W,P,A);
6900     R2:=I(W,P,A);
7000     END;
7100 REAL PROCEDURE I2(W,P,A);VALUE W,P;REAL P;REAL W;ARRAY A[0];
7200 CASE P MOD 4 OF
7300     BEGIN
7400     I2:=I(W,P,A);
7500     I2:=R(W,P,A);
7600     I2:=-I(W,P,A);
7700     I2:=-R(W,P,A);
7800     END;
7900 PROCEDURE TIMEDOMAIN(DA,DB,AA,BB);VALUE DA,DB;REAL DA,DB;ARRAY AA,BB[0];
8000     BEGIN
8100     VALUE ARRAY T1("OUTPUT"),T2("TIME");
8200     INTEGER I;
8300     REAL XI2, XFX,XI1,XF2,X1,X2,XI,Y1,Y2,YI,HI1,SC,QI;
8400     ARRAY YSTART[0:10];
8500     LABEL FAIL;
8600     PROCEDURE DIFEQNS(SP,XFX,N,XI,XF,Y,HI,EPS,ERR,ERRLBL);
8700     VALUE SP,N,XI,XF,HI,EPS,ERR;PROCEDURE XFX;
8800     INTEGER N;REAL SP,XI,XF,HI,EPS,ERR;
8900     LABEL ERRLBL;ARRAY Y[0];FORWARD;
9000     PROCEDURE FUNCT(X,Z,DZ);
9100     VALUE X;REAL X;ARRAY Z,DZ[0];
9200     BEGIN
9300     DZ[0]:=0;
9400     FOR I:=1 STEP 1 UNTIL DB DO
9500     BEGIN
9600     IF (DB-I)<=DA THEN
9700     BEGIN
9800     IF I=DB THEN DZ[I]:=-Z[I]*BB[I]+AA[I-(DB-DA)]*QI;
9900     ELSE DZ[I]:=-Z[I]*BB[I]+Z[I+1]+AA[I-(DB-DA)]*QI;
10000    END
10100    ELSE DZ[I]:=-Z[I]*BB[I]+Z[I+1];
10200    END;
10300 END;
10400 PROCEDURE DIFEQNS(SP,XFX,N,XI,XF,Y,HI,EPS,ERR,ERRLBL);
10500 VALUE SP,N,XI,XF,HI,EPS,ERR;
10600 PROCEDURE XFX;
10700 INTEGER N;
10800 REAL SP,XI,XF,HI,EPS,ERR;
10900 LABEL ERRLBL;
11000 ARRAY Y[0];
11100 COMMENT*XXXX;
11200 COMMENT PURPOSE
11300 SOLVE A SYSTEM OF FIRST ORDER ORDINARY DIFFERENTIAL
11400 EQUATIONS
11500 COMMENT*YYYY
11600 DESCRIPTION OF PARAMETERS
11700 SP - A POSITIVE VALUE FOR SP WILL CAUSE OUTPUT AT
11800 INCREMENTS OF SP IN ADDITION TO PROGRAM
11900 CONTROLLED OUTPUT. IF SP = 0 ONLY PROGRAM
12000 CONTROLLED OUTPUT WILL OCCUR
12100 XFX - PROCEDURE TO GENERATE THE RIGHT-HAND-SIDE OF
12200 THE DIFFERENTIAL EQUATION. IT MUST HAVE A

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12300          HEADING OF THE FORM                                00012300
12400          *PROCEDURE NAME(X,Y,DY)                          00012400
12500          VALUE X REAL X                                    00012500
12600          ARRAY Y,DY(0)                                     00012600
12700          WHERE GIVEN THE VALUE OF THE INDEPENDENT          00012700
12800          VARIABLE X AND THE VALUES OF THE DEPENDENT        00012800
12900          VARIABLES AT X, STORED IN Y(1:N), NAME RETURNS    00012900
13000          THE VALUES OF THE DERIVATIVES AT X IN DY(1:N)    00013000
13100          N - NUMBER OF DIFFERENTIAL EQUATIONS IN THE SYSTEM 00013100
13200          N SHOULD BE LESS THAN OR EQUAL TO 10              00013200
13300          XI - INITIAL VALUE OF THE INDEPENDENT VARIABLE     00013300
13400          XF - FINAL VALUE OF THE INDEPENDENT VARIABLE.     00013400
13500          XF MAY BE LESS THAN XI IN WHICH CASE              00013500
13600          HI MUST BE NEGATIVE                                00013600
13700          Y - ARRAY CONTAINING THE INITIAL VALUES OF THE   00013700
13800          DEPENDENT VARIABLES, WHICH MUST BE STORED IN       00013800
13900          Y(1) THRU Y(N). THESE ARE NOT PRESERVED.          00013900
14000          Y MUST BE DIMENSIONED (0:N)                        00014000
14100          HI - INITIAL STEPSIZE. HI MAY BE NEGATIVE, AND    00014100
14200          MUST SATISFY  $ABS(XF-XI)/(2*15) \geq ABS(H)$       00014200
14300          EPS - ERROR TOLERANCE  $(1.0@-10 \text{ LEQ } EPS \text{ LEQ } 1.0@-2)$  00014300
14400          ERR - PARAMETER TO SPECIFY WHICH CONVERGENCE       00014400
14500          CRITERION TO USE                                    00014500
14600          ERR = "SE" - STANDARD ERROR                        00014600
14700          ERR = "RE" - RELATIVE ERROR                       00014700
14800          ERR = "AE" - ABSOLUTE ERROR                       00014800
14900          ERRLBL - A LABEL IN THE CALLING PROGRAM TO WHICH A 00014900
15000          RETURN WILL BE MADE IN CASE OF AN INPUT ERROR    00015000
15100          OR IN CASE THE ITERATION FAILS TO CONVERGE        00015100
15200          PARAMETERS RETURNED                                00015200
15300          Y - CONTAINS THE SOLUTION VALUES AT THE FINAL    00015300
15400          PROGRAM SOLUTION POINT.                            00015400
15500          METHOD                                              00015500
15600          RATIONAL EXTRAPOLATION METHOD OF R.BULIRSCH AND J.STOER 00015600
15700          WHICH HAS BEEN FOUND TO BE SUPERIOR TO ALL OTHER   00015700
15800          METHODS BOTH IN SPEED AND ACCURACY FOR SMALL SYSTEMS OF 00015800
15900          DIFFERENTIAL EQUATIONS, SAY N LESS THAN 5.        00015900
16000          REFERENCE                                          00016000
16100          DIFEQNS IS AN ADAPTION OF THE FORTRAN PACKAGE DESUB 00016100
16200          WHICH IS VOL 2, ISSUE 1, OF THE NUMERICAL MATHEMATICS 00016200
16300          PROGRAM LIBRARY PROJECT, BELL TELEPHONE LABORATORIES, 00016300
16400          MURRAY HILL, NEW JERSEY ;                          00016400
16500          COMMENT*ZZZZ;                                       00016500
16600          BEGIN                                             00016600
16700          COMMENT INITIALIZATION OF MACHINE DEPENDENT CONSTANTS 00016700
16800          ZOT - REAL NUMBERS ARE CONSIDERED TO BE EQUAL IF   00016800
16900          THE DIFFERENCE BETWEEN THEM IS LESS THAN          00016900
17000           $ZOT = 2*(-33)$                                     00017000
17100          ZOTUP - THE CONSTANT USED IN THE STEP HALVING DECISION 00017100
17200          IN DESUB BASED ON COMPARING  $ZOTUP/DT$  WITH C.      00017200
17300           $ZOTUP = 2*33$                                        00017300
17400          COMMENT PARAMETERS FOR NUMERICAL IMPLEMENTATION OF THE ALGORITHM 00017400
17500          HDIV - THE CONSTANT USED IN FINDING HMIN, THE MINIMUM 00017500
17600          VALUE THE STEPSIZE IS ALLOWED TO ATTAIN SUCH      00017600
17700          THAT  $HMIN = H/HDIV$                                 00017700
17800           $HDIV = 1024$                                        00017800
17900          EMIN - THE LOWER BOUND ON EPS                       00017900
18000           $EMIN = 1.0@-10$                                     00018000
18100          EMAX - THE UPPER BOUND ON EPS                       00018100
18200           $EMAX = 1.0@-2$  ;                                       00018200
18300          COMMENT (XEND - XSTART) / P2 IS USED AS LOWER BOUND 00018300

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18400 FOR INITIAL STEPSIZE HI. P2 IS SET AT 2 * 15. ;
18500 COMMENT THE ORDER OF EXTRAPOLATION JM IS SET AT 6. ;
18600 COMMENT THE MAXIMUM NO. OF DIFFERENTIAL EQUATIONS IN THE SYSTEM
18700 IS SET AT 10. THIS LIMIT MAY BE INCREASED BY THE USER
18800 BUT THE PROGRAM MAY NOT BE EFFICIENT FOR LARGER SYSTEMS ;
18900 DEFINE ZOI = 1.16415321827E-10 #
19000 ZOTUP = 8589934592.0 #
19100 HDIV = 1024.0 #
19200 EMIN = 1.0E-10 #
19300 EMAX = 1.0E-2 #
19400 P2 = 32768.0 #
19500 JM = 6 #
19600 NMAX = 10 #
19700 COMMENT ERROR INFORMATION PARAMETERS ;
19800 REAL EX,ER,EH;
19900 INTEGER NE;
20000 COMMENT OUTPUT PARAMETERS ;
20100 REAL X,XR,T1;
20200 INTEGER NP;
20300 COMMENT VARIOUS LOGICAL VARIABLES ;
20400 BOOLEAN STYPE,KONVF,RE,SE,AE;
20500 COMMENT OTHER VARIABLES ;
20600 REAL TTL,H,HMIN,HMAX,HP,HQ,HR,XP,XPMX,XT,FH ;
20700 INTEGER KOUNT,I,J ;
20800 COMMENT WORKING ARRAYS ;
20900 ARRAY S,R,YA,YR,YL,YM,DY,DZ,SA[0:N], DL[0:7],
21000 DT[0:N,0:7], YG,YH,SG[0:8,0:N];
21100 LABEL L1,L2,L3,L4,L5,EXIT ;
21200
21300 FILE PRINTER(KIND=PRINTER, BUFFERS=2, MAXRECSIZE=22);
21400 FORMAT TYME(7)"ELAPSED PROCESSOR TIME IN SECONDS = ",F10.4);
21500 PROCEDURE ERROR(XFX,XCURNT,ERRNUM,PRNT,ERRLBL);
21600 VALUE XCURNT,ERRNUM;
21700 PROCEDURE XFX;
21800 REAL XCURNT ;
21900 INTEGER ERRNUM ;
22000 BOOLEAN PRNT ;
22100 LABEL ERRLBL;
22200 COMMENT PURPOSE
22300 PRINT THE CURRENT VALUES OF X AND ARRAY Y AND THE
22400 APPROPRIATE ERROR MESSAGE, WHENEVER AN ERROR OCCURS
22500 DESCRIPTION OF PARAMETERS
22600 XCURNT - CURRENT VALUE OF X
22700 ERRNUM - NUMBER INDICATING THE ERROR OCCURRED
22800 PRNT - BOOLEAN VARIABLE WHOSE VALUE IS TRUE IF
22900 VALUES OF X,Y ETC.
23000 OTHER PROCEDURES REQUIRED
23100 XFX ;
23200
23300 LABEL BEGIN L1 ;
23400 IF NOT PRNT THEN GO TO L1 ;
23500 XFX(XCURNT,Y,DY) ;
23600
23700 L1: BEGIN
23800 SWITCH FORMAT ERRTLE(/X30"ERROR"/X30"*****"/) ;
23900 (/ "***** ND CONVERGENCE IN ABOVE STEP TO",X19,"*****"/
24000 "***** X = ",R12.6," WITH H = ",R12.6," *****"/
24100 "***** THE LIMITING ERROR IS ",R12.6," IN EQUATION ",I2,"*****"),
24200 (/ "*****",X17,"N LSS 0 OR N GTR 10",X18,"*****"),
24300 (/ "*****",X15,"EPS LSS @-10 OR EPS GTR @-2",X15,"*****"),
24400 (/ "*****",X15,"H I (XEND - XSTART) LSS 0",X15,"*****"),

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24500 (/ "*****",X15,"H = 0 OR XEND = XSTART",X16,"*****"),
24600 (/ "*****",X1,"H,LSS (XEND - XSTART)/2*15 OR H GTR (XEND - XSTART)",
24700 X1,"*****")
24800 WRITE(PRINTER,ERRTL);
24900 IF ERRNUM = 0 THEN
25000 WRITE(PRINTER,SWFMT[ERRNUM],EX,EH,ER,N)
25100 ELSE
25200 WRITE(PRINTER,SWFMT[ERRNUM]);
25300 GO TO ERRLBL ;
25400 END
25500 END ERROR ;
25600 PROCEDURE END DESUB(X,Y,H,HMIN,REDIF);
25700 VALUE HMIN,REDIF;
25800 REAL X,H,HMIN;
25900 ARRAY Y(0);
26000 BOOLEAN REDIF;
26100 BEGIN
26200 COMMENT PERFORMS A SINGLE INTEGRATION STEP. WHEN THE PROCEDURE
26300 IS EXITED,X AND THE ARRAY Y WILL CONTAIN THE MOST RECENTLY COMPUTED
26400 VALUES FOR THE INDEPENDENT AND DEPENDENT VARIABLES ;
26500 COMMENT DESUB IS BASED ON THE ALGOL PROCEDURE DIFFSYS OF
26600 R. BULIRSCH AND J. STOER, NUMERISCHE MATHEMATIK 8,1-13(1966) AND
26700 INCORPORATES THE MODIFICATIONS GIVEN IN SUBROUTINE DESUB OF THE
26800 BELL TELEPHONE LABORATORIES DESUB PACKAGE ;
26900 REAL A,B,B1,C,G,U,V,T,A,FC;
27000 INTEGER I,J,K,KK,JJ,L,M,SR,JMAX,JR,JS,N1;
27100 LABEL RESETH,RESTART1,RESTART2,EXIT;
27200 BOOLEAN KONV,BO,BH;
27300 COMMENT ALTERNATE INITIALIZATION IN CASE A SPECIFIED POINT HAS
27400 BEEN OVERSTEPPED SO THAT A RETURN MUST BE MADE TO THE PREVIOUS POINT;
27500 N1 := N+1;
27600 IF REDIF THEN
27700 BEGIN
27800 JMAX := JM+3;
27900 WRITE(Y[*],N1,YA[*]);
28000 GO TO RESTART1;
28100 END;
28200 COMMENT INITIALIZATION ;
28300 COMMENT FOR AN EXTRAPOLATION OF ORDER JM, JM+1 UNEXTRAPOLATED
28400 APPROXIMATIONS ARE REQUIRED. THREE MORE ARE ALLOWED IN ATTEMPTING
28500 TO ACHIEVE CONVERGENCE ;
28600 JMAX := JM+3;
28700 COMMENT SAVE THE INITIAL VALUES FOR THE DEPENDENT VARIABLES AND
28800 THE ERROR TEST VECTOR FOR THE STEP ;
28900 WRITE(YA[*],N1,Y[*]);
29000 WRITE(SA[*],N1,SL[*]);
29100 COMMENT USE THE FUNCTION ROUTINE TO OBTAIN THE INITIAL SLOPES.
29200 DZ = DY/DX ;
29300 XFX(X,Y,DZ);
29400 COMMENT THE LOGICAL VARIABLE BH DETERMINES WHETHER THE STEP SIZE
29500 HAS BEEN HALVED. INITIALLY FALSE.
29600 LATER BH IS FALSE IF THE STEPSIZ IS CUT BY A FACTOR NOT 2 ;
29700 RESTART1: BH := FALSE;
29800 COMMENT PRESET THE CONVERGENCE SUCCESS FLAG TRUE ;
29900 KONVF := TRUE;
30000 COMMENT ADVANCE THE INDEPENDENT VARIABLE BY THE STEPSIZE H ;
30100 RESTART2: A := H+X;
30200 COMMENT SET THE SWITCH FOR THE FIRST SET OF D COEFFICIENTS ;
30300 BO := FALSE ;
30400 COMMENT INITIALIZE THE H SEQUENCE: H/M, H/JR, H/JS ;
30500 M := 1; JR := 2; JS := 3;

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30600 COMMENT JJ IS THE INDEX FOR THE ARRAY OF VALUES SAVED IN CASE 00030600
30700 THE INTERVAL MUST BE HALVED ; 00030700
30800 JJ := -1 ; 00030800
30900 COMMENT INTEGRATION STEP: MID-POINT RULE + EXTRAPOLATION ; 00030900
31000 FOR J := 0 STEP 1 UNTIL JMAX DO 00031000
31100 BEGIN 00031100
31200 COMMENT SET THE VALUES OF THE EXTRAPOLATION COEFFICIENTS TO 00031200
31300 THEIR CORRECT VALUES FOR THIS EXTRAPOLATION STEP: 00031300
31400 D[1]:=16/9, D[3]:=64/9, D[5]:=256/9 OR D[1]:=9/4, D[3]:=9, D[5]:=36 ; 00031400
31500 IF BH THEN 00031500
31600 BEGIN 00031600
31700 D[1] := 1.777777777778; 00031700
31800 D[3] := 7.111111111111; 00031800
31900 D[5] := 28.444444444444; 00031900
32000 END 00032000
32100 ELSE 00032100
32200 BEGIN 00032200
32300 D[1] := 2.25; 00032300
32400 D[3] := 9.0; 00032400
32500 D[5] := 36.0; 00032500
32600 END; 00032600
32700 COMMENT IF THE ORDER OF THE EXTRAPOLATION STEP BEING COMPUTED 00032700
32800 IS LESS THAN JM/2, SET KONV FALSE ; 00032800
32900 IF J LSS JM/2 THEN KONV := FALSE ELSE KONV := TRUE ; 00032900
33000 COMMENT IF J IS GREATER THAN JM RESTRICT THE ORDER OF 00033000
33100 EXTRAPOLATION TO JM AND ADJUST THE LAST EXTRAPOLATION COEFFICIENT ; 00033100
33200 IF J GTR JM THEN 00033200
33300 BEGIN 00033300
33400 L := JM; D[L] := 4.0 * D[L-2]; 00033400
33500 COMMENT DISCOURAGE THE STEP-INCREASING FACTOR FC BY A FACTOR OF 00033500
33600 SQRT(2) SINCE CONVERGENCE WAS NOT OBTAINED IN JM EXTRAPOLATIONS ; 00033600
33700 FC := .707106781187 * FC; 00033700
33800 END 00033800
33900 ELSE 00033900
34000 COMMENT THE NUMBER, J, OF EXTRAPOLATIONS HAS NOT EXCEEDED JM. 00034000
34100 FIND D(J) = (H DIVIDED BY H/M)*2. 00034100
34200 ADJUST THE FACTOR, FC, USED TO ADJUST THE STEPSIZE FOR THE NEXT STEP 00034200
34300 TO BE TAKEN ; 00034300
34400 BEGIN 00034400
34500 L := J; D[L] := M*M; 00034500
34600 FC := (JM-J)/6.0 + 1.0; 00034600
34700 END; 00034700
34800 COMMENT MODIFIED MIDPOINT RULE USED TO FIND FIRST VALUE FOR THIS 00034800
34900 EXTRAPOLATION STEP ; 00034900
35000 M := M+M; G := H/M; B := G+G; 00035000
35100 COMMENT IF THE STEPSIZE HAS BEEN HALVED AND IF THE ORDER OF THE 00035100
35200 EXTRAPOLATION STEP DOES NOT EXCEED THAT FOR WHICH PREVIOUSLY COMPUTED 00035200
35300 VALUES WERE SAVED, THEY CAN BE RESTORED ; 00035300
35400 IF BH AND J LSS JMAX=1 THEN 00035400
35500 BEGIN 00035500
35600 WRITE(YM[*],N1,YH[J,*]); 00035600
35700 WRITE(YL[*],N1,YG[J,*]); 00035700
35800 WRITE(ST[*],N1,SG[J,*]); 00035800
35900 END 00035900
36000 ELSE 00036000
36100 BEGIN 00036100
36200 COMMENT OTHERWISE THE STARTING VALUES FOR THE MIDPOINT RULE 00036200
36300 MUST BE COMPUTED ; 00036300
36400 KK := .5 * (M-2); 00036400
36500 WRITE(YL[*],N1,YA[*]); 00036500
36600 FOR I := 1 STEP 1 UNTIL N DO 00036600

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36700      YM[I] := YA[I] + DZ[I]*G;
36800      WRITE(S[*],N1,SAL[*]);
36900      COMMENT THE MEMBER OF THE H SEQUENCE BEING USED BY THE MIDPOINT
37000      INTEGRATION RULE IS H/M. COMPUTE TO THE END OF THE STEP CONTINUOUSLY
37100      UPDATING THE VECTOR,S, CONTAINING THE MAXIMUM VALUE COMPUTED UP TO
37200      NOW FOR EACH DEPENDENT VARIABLE ;
37300      M := M-1;
37400      FOR K := 1 STEP 1 UNTIL M DO
37500      BEGIN
37600      XFX(X + K*G,YM,DY);
37700      FOR I := 1 STEP 1 UNTIL N DO
37800      BEGIN
37900      U := YL[I] + DY[I]*B;
38000      YL[I] := YM[I];
38100      YM[I] := U;
38200      S[I] := MAX(ABS(U),S[I]);
38300      END;
38400      COMMENT IN CASE THE INTERVAL MUST BE HALVED NEXT TIME, SAVE THE
38500      VALUES AT HALFWAY ALONG (KK = M/2 - 1) THE STEP UNLESS K=2 ;
38600      IF K=KK AND K NEQ 2 THEN
38700      BEGIN
38800      JJ := JJ+1;
38900      WRITE(YH[JJ,*],N1,YM[*]);
39000      WRITE(YG[JJ,*],N1,YL[*]);
39100      WRITE(SG[JJ,*],N1,S[*]);
39200      END;
39300      END KLOOP;
39400      END;
39500      XFX(A,YM,DY);
39600      FOR I := 1 STEP 1 UNTIL N DO
39700      BEGIN
39800      COMMENT V IS USED TO SAVE THE VALUE OBTAINED BY THE MIDPOINT RULE
39900      USING THE PREVIOUS MEMBER OF THE H SEQUENCE (THE FIRST TIME THROUGH
40000      THIS VALUE IS UNDEFINED, BUT IT IS NOT USED SINCE L IS LESS THAN 1);
40100      V := DT[I,U];
40200      COMMENT COMPUTE THE FINAL VALUE OBTAINED FOR THIS MEMBER OF THE
40300      H SEQUENCE BY THE MODIFIED MIDPOINT RULE ;
40400      TA := C := DT[I,0] := .5 * (DY[I]*G + YL[I] + YM[I]);
40500      COMMENT IF THE VALUE JUST COMPUTED BY THE MIDPOINT RULE SHOWS
40600      A LARGE JUMP FROM THE PREVIOUS VALUE, HALVE THE INTERVAL ;
40700      IF ABS(V)*ZOTUP LSS ABS(C) AND ABS(H) NEQ HMIN
40800      AND J GTR .5*JM THEN GO TO RESETH;
40900      COMMENT PERFORM THE L STEPS FOR THE CURRENT LTH ORDER
41000      EXTRAPOLATION STEP. IF THE DENOMINATIONATOR OF THE RATIONAL FUNCTION
41100      GOES TO ZERO AT ANY STEP,SET DT AT THAT STEP TO ITS VALUE JUST BEFOR;
41200      FOR K:= 1 STEP 1 UNTIL L DO
41300      BEGIN
41400      B1 := D[K]*V;
41500      B := B1-C;
41600      U := V;
41700      IF B NEQ 0 THEN
41800      BEGIN
41900      B := (C-V)/B;
42000      U := C*B;
42100      C := B1*B;
42200      END;
42300      V := DT[I,K];
42400      DT[I,K] := U;
42500      TA := U+TA;
42600      END KLOOP;
42700      COMMENT EACH DEPENDENT VARIABLE IS CHECKED SEPARATELY FOR

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42800      CONVERGENCE ;
42900          IF RE THEN S[I] := ABS(Y[I]) ELSE
43000          IF SE THEN S[I] := MAX(ABS(TA),S[I]);
43100          R[I] := ABS(Y[I]-TA);
43200          Y[I] := TA;
43300          S[I] := MAX(S[I],EPS);
43400          IF R[I] GTR EPS*S[I] THEN KONV := FALSE;
43500      END ILOOP;
43600      IF KONV THEN GO TO EXIT;
43700      COMMENT RESET THE EXTRAPOLATION COEFFICIENTS ;
43800      D[2] := 4.0; D[4] := 16.0;
43900      COMMENT FLIP THE SWITCH FOR THE NEXT SET OF COEFFICIENTS ;
44000      BO := NOT BO;
44100      COMMENT TAKE THE NEXT MEMBER OF THE H SEQUENCE ;
44200      M := JR; JR := JS; JS := M+M;
44300      COMMENT AND GO BACK FOR THE NEXT EXTRAPULATION ;
44400      END JLOOP;
44500      COMMENT IF, AFTER ALL THE EXTRAPOLATIONS ALLOWED, CONVERGENCE HAS
44600      NOT BEEN ACHIEVED, ATTEMPT TO HALVE H SO THAT THE SAVED VALUES CAN
44700      BE USED (SET BH TRUE FOR THIS PURPOSE). IF HALVING H MAKES IT LESS
44800      THAN HMIN, SET H:=HMIN. IN THIS CASE THE SAVED VALUES CANNOT BE USED.
44900      IF H HAD ALREADY BEEN AT HMIN, CONVERGENCE CANNOT BE ACHIEVED FOR
45000      THIS HMIN AND THIS EPS CRITERION. SET KONVF FALSE ;
45100      BH := NOT BH;
45200      RESETH: IF ABS(H) LEQ HMIN THEN
45300          BEGIN KONVF := FALSE; GO TO EXIT; END;
45400          H := .5*H;
45500          IF ABS(H) GEQ HMIN THEN GO TO RESTART2;
45600          H := SIGN(H) * HMIN;
45700          GO TO RESTART1;
45800      COMMENT WHETHER OF NOT CONVERGENCE HAS BEEN ACHIEVED SET A NEW
45900      SUGGESTED STEPSIZE FOR THE NEXT STEP AND ASSIGN THE END OF STEP
46000      VALUE TO THE INDEPENDENT VARIABLE ;
46100      EXIT: H := FC*H;
46200          X := A;
46300      END DESUB;
46400      COMMENT MAIN PROCEDURE BEGINS HERE ;
46500          T1 := TIME(2);
46600      COMMENT SET ERROR CRITERION PARAMETER ;
46700          IF ERR = "RE" OR ERR = "0000RE" OR ERR = " RE"
46800          OR ERR = "RE0000" OR ERR = "RE " THEN RE := TRUE
46900          ELSE
47000          IF ERR = "SE" OR ERR = "0000SE" OR ERR = " SE"
47100          OR ERR = "SE0000" OR ERR = "SE " THEN SE := TRUE
47200          ELSE
47300          BEGIN
47400              AE := TRUE;
47500              FOR I := 1 STEP 1 UNTIL N DO S[I] := 1.0;
47600          END;
47700      COMMENT SET SPECIFIED POINTS FLAG FOR OUTPUT AND CHECK SP ;
47800          NP := IF SP = 0 THEN 0 ELSE 1 ;
47900          KQUANT := 0 ;
48000          IF SP * (XF - XI) LSS 0 THEN SP := SP * SIGN(XF-XI) ;
48100      COMMENT CHECK PARAMETERS AND PERFORM INITIALIZATION ;
48200          STYPE := TRUE ;
48300          H := HI ;
48400          IF N LEQ 0 OR N GTR NMAX THEN ERROR(XFX,XI,1,TRUE,ERRLBL);
48500          IF EPS LSS EMIN OR EPS GTR EMAX THEN ERROR(XFX,XI,2,TRUE,
48600              ERLBL);
48700          TTL := XF - XI ;
48800          IF TTL * H LSS 0 THEN ERROR(XFX,XI,3,TRUE,ERRLBL) ELSE IF

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48900		TTL * H = 0 THEN ERROR(XFX,XI,4,TRUE,ERRLBL);	00048900
49000		IF (H / TTL) * P2 LSS 1 OR H / TTL GTR 1	00049000
49100		THEN ERROR(XFX,XI,5,TRUE,ERRLBL);	00049100
49200		KONVF := TRUE;	00049200
49300		FOR I := 1 STEP 1 UNTIL N DO S[I] := ABS(Y[I]);	00049300
49400		HMIN := ABS(H/HDIY);	00049400
49500		HMAX := ABS(TTL);	00049500
49600		HP := 0;	00049600
49700		XP := X := XI;	00049700
49800	COMMENT	BEGIN SOLUTION OF DIFFERENTIAL EQUATIONS	00049800
49900		THE RIGHTMOST 21 BITS OF THE STEPSIZE H ARE ZEROED OUT	00049900
50000		IN AN ATTEMPT TO REDUCE THE BUILD UP OF ROUND OFF ERROR.	00050000
50100		THIS LEAVES H WITH APPROXIMATELY 5 NON ZERO DECIMAL	00050100
50200		DIGITS;	00050200
50300	L1:	IF X NEQ XI THEN	00050300
50400		H.[20:21] := 0;	00050400
50500		IF NP = 0 AND NOT STYPE THEN GO TO L3;	00050500
50600		XPMX := XP - X;	00050600
50700		FH := XPMX/H;	00050700
50800		IF FH GTR ZOT THEN GO TO L3;	00050800
50900		J := IF ABS(FH) GTR ZOT THEN 1 ELSE 0;	00050900
51000		CASE J OF	00051000
51100		BEGIN	00051100
51200		BEGIN	00051200
51300		WRITE(YR[*],N+1,Y[*]);	00051300
51400		HQ := HP;	00051400
51500		XR := X;	00051500
51600		END;	00051600
51700		BEGIN	00051700
51800		HR := HQ := XPMX + HP;	00051800
51900		XR := XI;	00051900
52000		DESUB(XR,YR,HR,HQ,TRUE);	00052000
52100		HQ := XR - XT;	00052100
52200		END;	00052200
52300		END;	00052300
52400		STYPE := TRUE;	00052400
52500		XFX(XR,YR,DY);	00052500
52600		STYPE := FALSE;	00052600
52700		IF KONVF THEN	00052700
52800		BEGIN	00052800
52900		IF (XF - XR)/TTL LEQ 0 THEN GO TO EXIT;	00052900
53000		KOUNT := KOUNT + 1;	00053000
53100		XP := XI + KOUNT * SP;	00053100
53200		IF (XP - XF)/H GTR 0 THEN XP := XF;	00053200
53300		GO TO L1	00053300
53400		ELSE GO TO L4;	00053400
53500	COMMENT	OUTPUT RESULTS AT A NATURAL STEP;	00053500
53600	L3:	IF ABS((X-XR)/H) LEQ ZOT THEN GO TO L2;	00053600
53700		XFX(X,Y,DY);	00053700
53800	COMMENT	PERFORM A STEP IN SOLUTION OF DIFFERENTIAL EQUATIONS;	00053800
53900	L2:	IF (XF - X)/TTL LEQ 0 THEN GO TO EXIT;	00053900
54000		IF ABS(H) LSS HMIN THEN H := HMIN * SIGN(H) ELSE	00054000
54100		IF ABS(H) GTR HMAX THEN H := HMAX * SIGN(H);	00054100
54200		IF (XF - X - H)/TTL LSS 0 THEN H := XF - X;	00054200
54300		XT := X;	00054300
54400		DESUB(X,Y,H,HMIN,FALSE);	00054400
54500		HP := X - XI;	00054500
54600		IF KONVF THEN GO TO L1 ELSE	00054600
54700	BEGIN	J := 0;	00054700
54800		GO TO L5	00054800
54900			00054900

55000	END	;	00055000
55100	L4:	IF KONVF THEN GO TO EXIT ;	00055100
55200		J := 1 ;	00055200
55300	L5:	ER := 0 ;	00055300
55400		FOR I := 1 STEP 1 UNTIL N DO	00055400
55500		BEGIN	00055500
55600	LABEL	L6;	00055600
55700		IF ER * S[I] GEQ R[I] THEN GO TO L6 ;	00055700
55800		ER := R[I] / S[I] ;	00055800
55900		NE := I ;	00055900
56000	L6:	END	00056000
56100		;	00056100
56200		CASE J OF	00056200
56300		BEGIN	00056300
56400		BEGIN	00056400
56500		EH := HP ;	00056500
56600		EX := X ;	00056600
56700		ERROR(XFX,X,0,TRUE,ERRLBL);	00056700
56800		END	00056800
56900		BEGIN	00056900
57000		EH := HQ ;	00057000
57100		EX := XR ;	00057100
57200		ERROR(XFX,XR,0,FALSE,ERRLBL);	00057200
57300		END	00057300
57400		END	00057400
57500	EXIT:	BEGIN END;	00057500
57600		END DIFEQNS;	00057600
57700	STARTP;		00057700
57800		XRANGE(-250,1250);YRANGE(-250,1250);	00057800
57900		SUBFRM(0,0,1000,1000);	00057900
58000		DEVICE(3,1,0,0);	00058000
58100		WRITE(FILE97, <"INITIAL CONDITIONS?">);	00058100
58200		READ(FILE97, /, FOR I:=1 STEP 1 UNTIL DB DO YSTART[I]);	00058200
58300		WRITE(FILE97, <"FINAL TIME VALUE AND INCREMENT?">);	00058300
58400		READ(FILE97, /, XF1, XI2);	00058400
58500		WRITE(FILE97, <"VALUE OF INPUT STEP?">);	00058500
58600		READ(FILE97, /, Q1);	00058600
58700		WRITE(FILE97, <"TIME RANGE AND INCREMENT?">);	00058700
58800		READ(FILE97, /, X1, X2, XI);	00058800
58900		WRITE(FILE97, <"OUTPUT RANGE AND INCREMENT?">);	00058900
59000		READ(FILE97, /, Y1, Y2, YI);	00059000
59100		AT(90,1000);TXDIR(0);WRTARY(1,T1);	00059100
59200		AT(1000,5);TXDIR(0);WRTARY(1,T2);	00059200
59250		XRANGE(X1,X2);YRANGE(Y1,Y2);XAXIS(Y1,XI);YAXIS(X1,YI);	00059250
59300		NOLINE;	00059300
59400		HI1:=XI2/10;	00059400
59500		FOR XI1:=-XI2 STEP XI2 UNTIL (XF1-XI2) DO	00059500
59600		BEGIN	00059600
59700		XF2:=XI1+XI2;	00059700
59800	DIFEQNS	(0,FUNCT,DB,XI1,XF2,YSTART,HI1,@-3,"RE",FAIL);	00059800
59900		CHARMD;	00059900
60000		LINE TO(XF2,YSTART[1]);	00060000
60100		END;	00060100
60200		WHEN(10);	00060200
60300		WRITE(FILE97, <"WANT TO CHANGE THE SCALE?">);	00060300
60400		READ(FILE97, /, SC);	00060400
60500		IF SC="Y" THEN TIMEDOMAIN(DA,DB,AA,BB);	00060500
60600		ENDP;	00060600
60700	FAIL:	END;	00060700
60800		COMMENT *** PROCEDURE BUDE ***;	00060800
60900		PROCEDURE BUDE(W,DA,DB,A,B);VALUE W,DA,DB;REAL DA,DB;	00060900
		REAL W;ARRAY A[0],B[0];	00060900

```

61000      BEGIN
61100      FILE EE1(KIND=DISK,MAXRECSIZE=40);
61200      FILE E1(KIND=DISK,MAXRECSIZE=40);
61300      FILE E2(KIND=DISK,MAXRECSIZE=40);
61400      ARRAY YB(0:100000);
61500      INTEGER I;
61600      REAL SC,GR,X,Y,COF,S,G;
61700      REAL LF,FI,HF,LG,HG,LP,HP,GI,PI;
61800      VALUE ARRAY T1("***BODE DIAGRAMS***");
61900      VALUE ARRAY T2("GAIN(DB)");
62000      VALUE ARRAY T3("PHASE(DEG)");
62100      VALUE ARRAY T4("FREQUENCY(HZ)");
62200      PROCEDURE MGRAPH1; FORWARD;
62300      PROCEDURE MGRAPH2; FORWARD;
62400      PROCEDURE MGRAPH3; FORWARD;
62500      PROCEDURE MGRAPH4; FORWARD;
62600      PROCEDURE PGRAPH1; FORWARD;
62700      PROCEDURE PGRAPH2; FORWARD;
62800      PROCEDURE PGRAPH3; FORWARD;
62900      PROCEDURE PGRAPH4; FORWARD;
63000      PROCEDURE THREE(W,YB); VALUE W; REAL W; ARRAY YB(0); FORWARD;
63100      PROCEDURE MGRAPH1;
63200      BEGIN
63300      FOR W:=LF STEP W*1.1/20 UNTIL HF DO
63400          BEGIN
63500      G:=((R1(W,DA,A)**2+I1(W,DA,A)**2)/(R1(W,DB,B)**2+I1(W,DB,B)**2));
63600      IF G>0 THEN BEGIN
63700      Y:=10*LOG(G);
63800      X:=W;
63900      IF Y<=HG OR Y>=LG THEN
64000          BEGIN
64100      CHARM;
64200      LINETO(X,Y);
64300      WRITE(E1,/,X,Y);
64400      END;
64500      END;
64600      END;
64700      END;
64800      PROCEDURE MGRAPH2;
64900      BEGIN
65000      FOR W:=LF STEP W*1.1/20 UNTIL HF DO
65100          BEGIN
65200      G:=((R1(W,DA,A)**2+I1(W,DA,A)**2)/(R2(W,DB,B)**2+I2(W,DB,B)**2));
65300      IF G>0 THEN BEGIN
65400      Y:=10*LOG(G);
65500      X:=W;
65600      IF Y<=HG OR Y>=LG THEN
65700          BEGIN
65800      CHARM;
65900      LINETO(X,Y);
66000      WRITE(E1,/,X,Y);
66100      END;
66200      END;
66300      END;
66400      END;
66500      PROCEDURE MGRAPH3;
66600      BEGIN
66700      FOR W:=LF STEP W*1.1/20 UNTIL HF DO
66800          BEGIN
66900      G:=((R2(W,DA,A)**2+I2(W,DA,A)**2)/(R1(W,DB,B)**2+I1(W,DB,B)**2));
67000      IF G>0 THEN BEGIN

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67100	Y:=10*LOG(G);	00067100
67200	X:=W;	00067200
67300	IF Y<=HG OR Y>=LG THEN	00067300
67400	BEGIN	00067400
67500	CHARMD;	00067500
67600	LINETO(X,Y);	00067600
67700	WRITE(E1,/,X,Y);	00067700
67800	END;	00067800
67900	END;	00067900
68000	END;	00068000
68100	END;	00068100
68200	PROCEDURE MGRAPH4;	00068200
68300	BEGIN	00068300
68400	FOR W:=LF STEP W*1.1/20 UNTIL HF DO	00068400
68500	BEGIN	00068500
68600	G:=(R2(W,DA,A)**2+I2(W,DA,A)**2)/(R2(W,DB,B)**2+I2(W,DB,B)**2);	00068600
68700	IF G>0 THEN BEGIN	00068700
68800	Y:=10*LOG(G);	00068800
68900	X:=W;	00068900
69000	IF Y<=HG OR Y>=LG THEN	00069000
69100	BEGIN	00069100
69200	CHARMD;	00069200
69300	LINETO(X,Y);	00069300
69400	WRITE(E1,/,X,Y);	00069400
69500	END;	00069500
69600	END;	00069600
69700	END;	00069700
69800	END;	00069800
69900	PROCEDURE PGRAPH1;	00069900
70000	BEGIN	00070000
70100	REAL NNT,DNT;	00070100
70200	YB[0]:=0;	00070200
70300	FOR W:=LF STEP W*1.1/20 UNTIL HF DO	00070300
70400	BEGIN	00070400
70500	I:=*+1;	00070500
70600	NNT:=(I1(W,DA,A)/R1(W,DA,A))-(I1(W,DB,B)/R1(W,DB,B));	00070600
70700	DNT:=1+(I1(W,DA,A)/R1(W,DA,A))*(I1(W,DB,B)/R1(W,DB,B));	00070700
70800	YB[I]:=ARCTAN(NNT/DNT)*COF;	00070800
70900	IF YB[I]<=0 THEN	00070900
70902	BEGIN	00070902
70904	WRITE(E2,/,I,W,YB[I]);	00070904
70906	THREE(W,YB);	00070906
70908	END ELSE	00070908
70910	BEGIN	00070910
70912	YB[I]:=YB[I]-180;	00070912
70914	WRITE(E2,/,I,W,YB[I]);	00070914
70916	THREE(W,YB);	00070916
70918	END;	00070918
71100	END;	00071100
71200	END;	00071200
71300	PROCEDURE PGRAPH2;	00071300
71400	BEGIN	00071400
71500	REAL NNT,DNT;	00071500
71600	YB[0]:=0;	00071600
71700	FOR W:=LF STEP W*1.1/20 UNTIL HF DO	00071700
71800	BEGIN	00071800
71900	I:=*+1;	00071900
72000	NNT:=(I1(W,DA,A)/R1(W,DA,A))-(I2(W,DB,B)/R2(W,DB,B));	00072000
72100	DNT:=1+(I1(W,DA,A)/R1(W,DA,A))*(I2(W,DB,B)/R2(W,DB,B));	00072100
72200	YB[I]:=ARCTAN(NNT/DNT)*COF;	00072200
72210	IF YB[I]<=0 THEN	00072210

72220	BEGIN	00072220
72300	WRITE(E2,/,I,W,Y8[I]);	00072300
72400	THREE(W,Y8);	00072400
72410	END ELSE	00072410
72420	BEGIN	00072420
72421	Y8[I]:=Y8[I]-180;	00072421
72422	WRITE(E2,/,I,W,Y8[I]);	00072422
72424	THREE(W,Y8);	00072424
72426	END;	00072426
72500	END;	00072500
72600	END;	00072600
72700	PROCEDURE PGRAPH3;	00072700
72800	BEGIN	00072800
72900	REAL NNT,DNT;	00072900
73000	Y8[0]:=0;	00073000
73100	FOR W:=LF STEP W*1.1/20 UNTIL HF DO	00073100
73200	BEGIN	00073200
73300	I:=*+1;	00073300
73400	NNT:=(I2(W,DA,A)/R2(W,DA,A))-(I1(W,DB,B)/R1(W,DB,B));	00073400
73500	DNT:=1+(I2(W,DA,A)/R2(W,DA,A))*(I1(W,DB,B)/R1(W,DB,B));	00073500
73600	Y8[I]:=ARCTAN(NNT/DNT)*COF;	00073600
73700	IF Y8[I]<=0 THEN	00073700
73702	BEGIN	00073702
73704	WRITE(E2,/,I,W,Y8[I]);	00073704
73706	THREE(W,Y8);	00073706
73708	END ELSE	00073708
73710	BEGIN	00073710
73712	Y8[I]:=Y8[I]-180;	00073712
73714	WRITE(E2,/,I,W,Y8[I]);	00073714
73716	THREE(W,Y8);	00073716
73718	END;	00073718
73900	END;	00073900
74000	END;	00074000
74100	PROCEDURE PGRAPH4;	00074100
74200	BEGIN	00074200
74300	REAL NNT,DNT;	00074300
74400	Y8[0]:=0;	00074400
74500	FOR W:=LF STEP W*1.1/20 UNTIL HF DO	00074500
74600	BEGIN	00074600
74700	I:=*+1;	00074700
74800	NNT:=(I2(W,DA,A)/R2(W,DA,A))-(I2(W,DB,B)/R2(W,DB,B));	00074800
74900	DNT:=1+(I2(W,DA,A)/R2(W,DA,A))*(I2(W,DB,B)/R2(W,DB,B));	00074900
75000	Y8[I]:=ARCTAN(NNT/DNT)*COF;	00075000
75100	IF Y8[I]<=0 THEN	00075100
75102	BEGIN	00075102
75104	WRITE(E2,/,I,W,Y8[I]);	00075104
75106	THREE(W,Y8);	00075106
75108	END ELSE	00075108
75110	BEGIN	00075110
75112	Y8[I]:=Y8[I]-180;	00075112
75114	WRITE(E2,/,I,W,Y8[I]);	00075114
75116	THREE(W,Y8);	00075116
75118	END;	00075118
75300	END;	00075300
75400	END;	00075400
75500	PROCEDURE THREE(W,Y8);VALUE W;REAL W;ARRAY Y8[0];	00075500
75600	BEGIN	00075600
77900	X:=W;	00077900
78000	IF Y8[I]<=Y8[I-1] AND Y8[I]<=0 THEN	00078000
78100	BEGIN	00078100
78200	CHARMD,LINETD(X,Y8[I]);	00078200

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78300 WRITE(EE1,/,X,Y8[I]);
78400 END
78500 ELSE IF Y8[I]>=Y8[I-1] AND Y8[I]<=0 THEN
78600 BEGIN
78700 Y8[I]:=-180+Y8[I];
78800 IF Y8[I]<=Y8[I-1] THEN
78900 BEGIN
79000 CHARMD;LINETO(X,Y8[I]);
79100 WRITE(EE1,/,X,Y8[I]);
79200 END
79300 ELSE BEGIN
79400 Y8[I]:=-180+Y8[I];
79500 CHARMD;LINETO(X,Y8[I]);
79600 WRITE(EE1,/,X,Y8[I]);
79700 END;
79800 END;
81400 END;
81500 DEFINE
81600 GFDEC1=
81700 XRANGE(-250,1250);YRANGE(-250,1250);
81800 W:=LF;AT(110,1000);TXDIR(0);WRTARY(1,T2);
81900 AT(250,540);TXDIR(0);WRTARY(1,T1);
82000 SUBFRM(100,600,1000,970);YRANGE(LG,HG);
82100 XRANGE(LF,HF);
82200 GRID;
82300 YAXIS(LF,GI);
82400 MODE(3);
82500 GRID;
82600 XAXIS(0,-FI);NOLINE#;
82700 GFDEC2=
82800 XRANGE(-250,1250);YRANGE(-250,1250);
82900 W:=LF;AT(110,1000);TXDIR(0);WRTARY(1,T2);
83000 AT(250,540);TXDIR(0);WRTARY(1,T1);
83100 SUBFRM(100,600,1000,970);YRANGE(LG,HG);
83200 XRANGE(LF,HF);YAXIS(LF,GI);MODE(3);XAXIS(0,-FI);NOLINE#;
83300 PFDEC1=
83400 DRGFRM;MODE(0);XRANGE(-250,1250);YRANGE(-250,1250);
83500 W:=LF;AT(110,510);TXDIR(0);WRTARY(1,T3);
83600 AT(800,5);TXDIR(0);WRTARY(1,T4);
83700 SUBFRM(100,110,1000,500);YRANGE(LP,HP);XRANGE(LF,HF);
83800 GRID;
83900 YAXIS(LF,PI);
84000 MODE(3);
84100 GRID;
84200 XAXIS(-300,FI);NOLINE#;
84300 PFDEC2=
84400 DRGFRM;MODE(0);XRANGE(-250,1250);YRANGE(-250,1250);
84500 W:=LF;AT(110,510);TXDIR(0);WRTARY(1,T3);
84600 AT(800,5);TXDIR(0);WRTARY(1,T4);
84700 SUBFRM(100,110,1000,500);YRANGE(LP,HP);XRANGE(LF,HF);
84800 YAXIS(LF,PI);MODE(3);XAXIS(-300,FI);NOLINE#;
84900 BODESCALE=
85000 WRITE(FILE97,<"FREQUENCY RANGE?">);
85100 READ(FILE97,/,LF,HF);
85200 WRITE(FILE97,<"GAIN RANGE?">);
85300 READ(FILE97,/,LG,HG);
85400 WRITE(FILE97,<"PHASE RANGE?">);
85500 READ(FILE97,/,LP,HP);
85600 WRITE(FILE97,<"GAIN,PHASE AND FREQUENCY INCREMENT?">);
85700 READ(FILE97,/,GI,PI,FI)#;
85800 CHANGEBODESCALE=

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85900 WHEN(10);
86000 WRITE(FILE97,"DO YOU WANT TO CHANGE THE SCALE?");
86100 READ(FILE97,"SC");
86200 IF SC="Y" THEN BUDE(W,DA,DB,A,B)#;
86300 CDF:=180/3.14157;
86400 STARTP;
86500 DEVICE(3,1,0);EDGESR(1.312);
86600 IF DA<3 THEN
86700 BEGIN
86800 IF DB<3 THEN
86900 BEGIN
87000 LABEL BODE1;
87100 BODESCALE;
87200 WRITE(FILE97,"IS GRID WANTED?");
87300 READ(FILE97,"GR");
87400 FRAME("GRAPH");
87500 IF GR="Y" THEN
87600 BEGIN
87700 GFDEC1;
87800 MGRAPH1;
87900 PFDEC1;
88000 PGRAPH1;
88100 ENDP;
88200 CHANGEBODESCALE;
88300 END
88400 ELSE
88500 BEGIN
88600 GFDEC2;
88700 MGRAPH1;
88800 PFDEC2;
88900 PGRAPH1;
89000 ENDP;
89100 CHANGEBODESCALE;
89200 END;
89300 END
89400 ELSE
89500 BEGIN
89600 BODESCALE;
89700 WRITE(FILE97,"IS GRID WANTED?");
89800 READ(FILE97,"GR");
89900 FRAME("GRAPH");
90000 IF GR="Y" THEN
90100 BEGIN
90200 GFDEC1;
90300 MGRAPH2;
90400 PFDEC1;
90500 PGRAPH2;
90600 ENDP;
90700 CHANGEBODESCALE;
90800 END
90900 ELSE
91000 BEGIN
91100 GFDEC2;
91200 MGRAPH2;
91300 PFDEC2;
91400 PGRAPH2;
91500 ENDP;
91600 CHANGEBODESCALE;
91700 END;
91800 END;
91900 END

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92000 ELSE
92100 BEGIN
92200 IF DB<4 OR DB=4 THEN
92300 BEGIN
92400 LABEL BODE1;
92500 WRITE(FILE97,"<"IS GRID WANTED?">);
92600 READ(FILE97,"/GR");
92700 FRAME("GRAPH");
92800 IF GR="Y" THEN
92900 BEGIN
93000 GFDEC1;
93100 MGRAPH3;
93200 PFDEC1;
93300 PGRAPH3;
93400 ENDP;
93500 CHANGEBODESCALE;
93600 END
93700 ELSE
93800 BEGIN
93900 GFDEC2;
94000 MGRAPH3;
94100 PFDEC2;
94200 PGRAPH3;
94300 ENDP;
94400 CHANGEBODESCALE;
94500 END;
94600 END
94700 ELSE
94800 BEGIN
94900 BODESCALE;
95000 WRITE(FILE97,"<"IS GRID WANTED?">);
95100 READ(FILE97,"/GR");
95200 FRAME("GRAPH");
95300 IF GR="Y" THEN
95400 BEGIN
95500 GFDEC1;
95600 MGRAPH4;
95700 PFDEC1;
95800 PGRAPH4;
95900 ENDP;
96000 CHANGEBODESCALE;
96100 END
96200 ELSE
96300 BEGIN
96400 GFDEC2;
96500 MGRAPH4;
96600 PFDEC2;
96700 PGRAPH4;
96800 ENDP;
96900 CHANGEBODESCALE;
97000 END;
97100 END;
97200 END;
97300 END;
97400 COMMENT **** PROCEDURE NYQUIST ****;
97500 PROCEDURE NYQUIST(W,DA,DB,A,B);VALUE W,DA,DB;REAL DA,DB;
97600 REAL W;ARRAY A,B(0);
97700 BEGIN
97800 FILE EMI(KIND=DISK,MAXRECSIZE=14);
97900 REAL LF,HF,FI,LY,HY,YI,LX,HX,XI,DEM;
98000 VALUE ARRAY T1("NYQUIST DIAGRAM Y(W) VS X(W)");

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98100	VALUE ARRAY T2("Y");	00098100
98200	VALUE ARRAY T3("X");	00098200
98300	REAL SC,GR,G,X,Y;	00098300
98400	DEFINE	00098400
98500	NYQUISTSCALE=	00098500
98600	WRITE(FILE97,"<YRANGE AND Y INCREMENT?>");	00098600
98700	READ(FILE97,"/LY,HY,YI");	00098700
98800	WRITE(FILE97,"<XRANGE AND X INCREMENT?>");	00098800
98900	READ(FILE97,"/LX,HX,XI");	00098900
99000	WRITE(FILE97,"<FREQUENCY RANGE AND INCREMENT?>");	00099000
99100	READ(FILE97,"/LF,HF,FI)";	00099100
99200	CHANGENYQUISTSCALE=	00099200
99300	WHEN(10);	00099300
99400	WRITE(FILE97,"<DO YOU WANT TO CHANGE THE SCALE?>");	00099400
99500	READ(FILE97,"/SC");	00099500
99600	IF SC="Y" THEN NYQUIST(W,DA,DB,A,B);	00099600
99700	NFDEC1=	00099700
99800	W:=LF;AT(530,900);TXTDIR(0);WRTARY(1,T1);	00099800
99900	AT(510,1000);TXTDIR(0);WRTARY(1,T2);	00099900
100000	AT(950,450);TXTDIR(0);WRTARY(1,T3);	00100000
100100	YRANGE(LY,HY);XRANGE(LX,HX);GRID;YAXIS(0,YI);GRID;	00100100
100200	XAXIS(0,XI);NOLINE#;	00100200
100300	NFDEC2=	00100300
100400	W:=LF;AT(530,900);TXTDIR(0);WRTARY(1,T1);	00100400
100500	AT(520,1000);TXTDIR(0);WRTARY(1,T2);	00100500
100600	AT(950,430);TXTDIR(0);WRTARY(1,T3);	00100600
100700	YRANGE(LY,HY);XRANGE(LX,HX);YAXIS(0,YI);XAXIS(0,XI);NOLINE#;	00100700
100800	NPLOT=	00100800
100900	IF X<=HX OR X>=LX THEN	00100900
101000	IF Y<=HY OR Y>=LY THEN	00101000
101100	CHARMD;	00101100
101200	LINETO(X,Y);WRITE(EMI,"/W,X,Y)";	00101200
101300	STARTP;	00101300
101400	XRANGE(-250,1250);YRANGE(-250,1250);	00101400
101500	SUBFRM(0,0,1000,1000);	00101500
101600	DEVICE(3,1,0,0);	00101600
101700	IF DA<3 THEN	00101700
101800	BEGIN	00101800
101900	IF DB<3 THEN	00101900
102000	BEGIN	00102000
102100	LABEL NYQUIST2;	00102100
102200	NYQUISTSCALE;	00102200
102300	WRITE(FILE97,"<IS GRID WANTED?>");	00102300
102400	READ(FILE97,"/GR);	00102400
102500	FRAME("GRAPH");	00102500
102600	IF GR="Y" THEN	00102600
102700	BEGIN	00102700
102800	NFDEC1;GO TO NYQUIST2;	00102800
102900	END	00102900
103000	ELSE	00103000
103100	BEGIN	00103100
103200	NFDEC2;GO TO NYQUIST2;	00103200
103300	END;	00103300
103400	NYQUIST2:	00103400
103500	FOR W:=LF STEP FI UNTIL HF DO	00103500
103600	BEGIN	00103600
103700	DEM:=R1(W,DB,B)**2+I1(W,DB,B)**2;	00103700
103800	IF DEM NEQ 0 THEN BEGIN	00103800
103900	X:=(R1(W,DA,A)*R1(W,DB,B)+I1(W,DA,A)*I1(W,DB,B))/DEM;	00103900
104000	Y:=(I1(W,DA,A)*R1(W,DB,B)-R1(W,DA,A)*I1(W,DB,B))/DEM;	00104000
104100	NPLOT;	00104100

```

104200                                END;                                END;
104300                                END;                                END;
104400 ENDP;CHANGENYQUISTSCALE;
104500                                END;
104600                                END
104700 ELSE
104800                                BEGIN
104900 LABEL NYQUIST2;
105000 NYQUISTSCALE;
105100 WRITE(FILE97,"<"IS GRID WANTED?">);
105200 READ(FILE97,"/GR");
105300 FRAME("GRAPH");
105400 IF GR="Y" THEN
105500                                BEGIN
105600 NFDEC1;GO TO NYQUIST2;
105700                                END
105800 ELSE
105900                                BEGIN
106000 NFDEC2;GO TO NYQUIST2;
106100                                END;
106200 NYQUIST2:
106300 FOR W:=LF STEP FI UNTIL HF DO
106400                                BEGIN
106500 DEM:=R2(W,DB,B)**2+I2(W,DB,B)**2;
106600 IF DEM NEQ 0 THEN
106700 X:=(R1(W,DA,A)*R2(W,DB,B)+I1(W,DA,A)*I2(W,DB,B))/DEM;
106800 Y:=(I1(W,DA,A)*R2(W,DB,B)-R1(W,DA,A)*I2(W,DB,B))/DEM;
106900 NPLOT;
107000                                END;
107100                                END;
107200 ENDP;CHANGENYQUISTSCALE;
107300                                END;
107400                                END;
107500                                END
107600 ELSE
107700                                BEGIN
107800 IF DB<3 THEN
107900                                BEGIN
108000 LABEL NYQUIST2;
108100 NYQUISTSCALE;
108200 WRITE(FILE97,"<"IS GRID WANTED?">);
108300 READ(FILE97,"/GR");
108400 FRAME("GRAPH");
108500 IF GR="Y" THEN
108600                                BEGIN
108700 NFDEC1;GO TO NYQUIST2;
108800                                END
108900 ELSE
109000                                BEGIN
109100 NFDEC2;GO TO NYQUIST2;
109200                                END;
109300 NYQUIST2:
109400 FOR W:=LF STEP FI UNTIL HF DO
109500                                BEGIN
109600 DEM:=R1(W,DB,B)**2+I1(W,DB,B)**2;
109700 IF DEM NEQ 0 THEN
109800 X:=(R2(W,DA,A)*R1(W,DB,B)+I2(W,DA,A)*I1(W,DB,B))/DEM;
109900 Y:=(I2(W,DA,A)*R1(W,DB,B)-R2(W,DA,A)*I1(W,DB,B))/DEM;
110000 NPLOT;
110100                                END;
110200                                END;

```

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00104200
00104300
00104400
00104500
00104600
00104700
00104800
00104900
00105000
00105100
00105200
00105300
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00105500
00105600
00105700
00105800
00105900
00106000
00106100
00106200
00106300
00106400
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00106700
00106800
00106900
00107000
00107100
00107200
00107300
00107400
00107500
00107600
00107700
00107800
00107900
00108000
00108100
00108200
00108300
00108400
00108500
00108600
00108700
00108800
00108900
00109000
00109100
00109200
00109300
00109400
00109500
00109600
00109700
00109800
00109900
00110000
00110100
00110200

```

```

110300 ENDP;CHANGENYQUISTSCALE;
110400 END;
110500 ELSE
110600 BEGIN
110700 LABEL NYQUIST2;
110800 NYQUISTSCALE;
110900 WRITE(FILE97,"<"IS GRID WANTED?">);
111000 READ(FILE97,"GR");
111100 FRAME("GRAPH");
111200 IF GR="Y" THEN
111300 BEGIN
111400 NFDEC1;GO TO NYQUIST2;
111500 END
111600 ELSE
111700 BEGIN
111800 NFDEC2;GO TO NYQUIST2;
111900 END;
112000 NYQUIST2;
112100 FOR W:=LF STEP FI UNTIL HF DO
112200 BEGIN
112300 DEM:=R2(W,DB,B)**2+I2(W,DB,B)**2;
112400 IF DEM NEQ 0 THEN BEGIN
112500 X:=(R2(W,DA,A)*R2(W,DB,B)+I2(W,DA,A)*I2(W,DB,B))/DEM;
112600 Y:=(I2(W,DA,A)*R2(W,DB,B)-R2(W,DA,A)*I2(W,DB,B))/DEM;
112700 NPLOT;
112800 END;
112900 END;
113000 ENDP;CHANGENYQUISTSCALE;
113100 END;
113200 END;
113300 END;
113400 END;
113500 END;
113600 COMMENT **** PROCEDURE NICHOLS ****;
113700 PROCEDURE NICHOLS(W,DA,DB,A,B);VALUE W,DA,DB;REAL DA,DB;
113800 REAL W;ARRAY A,B(0);
113900 BEGIN
114000 FILE U(KIND=DISK,MAXRECSIZE=40);
114100 FILE CH(KIND=DISK,MAXRECSIZE=40);
114200 FILE PRS(KIND=DISK,MAXRECSIZE=40);
114300 FILE GRS(KIND=DISK,MAXRECSIZE=40);
114400 VALUE ARRAY T1("DBS"),T2("DEGS");
114500 ARRAY Y8(0:10000);
114600 REAL NNT,DNT;
114700 INTEGER I;
114800 REAL ML,MH,MI,PHAL,PHAH,PHAI,FAS;
114900 REAL M,F,P1,G,B1,GI,F1,SC,GR;
115000 REAL X,Y,COF,ICOF;
115100 REAL LF,HF,M1,G1;
115200 REAL GAIN1,GAIN2,GAININ;
115300 REAL FA1,FA2,FAI;
115400 REAL LG,HG,LP,HP,PI;
115500 REAL GAIN,P,F1;
115600 PROCEDURE SIX(Y8);ARRAY Y8(0);FORWARD;
115700 PROCEDURE SIX(Y8);ARRAY Y8(0);
115800 BEGIN
115900 IF GAIN>0 THEN
116000 BEGIN
116100 Y:=10*LOG(GAIN);
116200 IF Y8[I]<=Y8[I-1] AND Y8[I]<=0 THEN
116300 BEGIN
116400

```

```

00110300
00110400
00110500
00110600
00110700
00110800
00110900
00111000
00111100
00111200
00111300
00111400
00111500
00111600
00111700
00111800
00111900
00112000
00112100
00112200
00112300
00112400
00112500
00112600
00112700
00112800
00112900
00113000
00113100
00113200
00113300
00113400
00113500
00113600
00113700
00113800
00113900
00114000
00114100
00114200
00114300
00114400
00114500
00114600
00114700
00114800
00114900
00115000
00115100
00115200
00115300
00115400
00115500
00115600
00115700
00115800
00115900
00116000
00116100
00116200
00116300
00116400

```

```

118500 CHARMD;LINETO(Y8[I],Y);
118600 WRITE(CHI,7,Y8[I],Y);
118700 END
118800 ELSE IF Y8[I]>=Y8[I-1] AND Y8[I]<=0 THEN
118900 BEGIN
119000 Y8[I]:=-180+Y8[I];
119100 IF Y8[I]<=Y8[I-1] THEN
119200 BEGIN
119300 CHARMD;LINETO(Y8[I],Y);
119400 WRITE(CHI,7,Y8[I],Y);
119500 END
119600 ELSE BEGIN
119700 Y8[I]:=-180+Y8[I];
119800 CHARMD;LINETO(Y8[I],Y);
119900 WRITE(CHI,7,Y8[I],Y);
120000 END;
120100 END;
120200 END;
120300 END;
120400 END;
120500 END;
120600 END;
120700 END;
120800 END;
120900 END;
121000 END;
121100 END;
121200 END;
121300 END;
121400 END;
121500 END;
121600 END;
121700 END;
121800 END;
121900 END;
122000 END;
122100 END;
122200 END;
122300 END;
122400 END;
122500 END;
122600 END;
122700 END;
122800 END;
122900 END;
123000 END;
123100 END;
123200 END;
123300 END;
123400 END;
123500 END;
123600 END;
123700 END;
123800 END;
123900 END;
124000 END;
124100 END;
124200 END;
124300 END;
124400 END;
124500 END;
124600 END;
124700 END;
124800 END;
124900 END;
125000 END;
125100 END;
125200 END;
125300 END;
125400 END;
125500 END;
125600 END;
125700 END;
125800 END;
125900 END;
126000 END;

```

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00118500
00118600
00118700
00118800
00118900
00119000
00119100
00119200
00119300
00119400
00119500
00119600
00119700
00119800
00119900
00120000
00120100
00120200
00120300
00120400
00120500
00120600
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00121200
00121300
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00122500
00122600
00122700
00122800
00122900
00123000
00123100
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00124000
00124100
00124200
00124300
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00124500
00124600
00124700
00124800
00124900
00125000
00125100
00125200
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00125400
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00125800
00125900
00126000

```

```

DEFINE
NICHOL=
TXTSIZ(0);TXDIR(0);TXT("NICHOL");#
PHASECONTOURS=
G1:=TAN(P-F)/(SIN(P)-COS(P)*TAN(P-F));
IF G1>0 THEN
BEGIN
G:=20*LOG(G1);
X:=P*COF;Y:=G;CHARMD;LINETO(X,Y);
WRITE(PRS,7,F,G1,X,Y);
END#
GAINCONTOURS=
M1:=10*(M/20);G1:=10*(G/20);
B1:=-((M1*M1+G1*G1*(M1*M1-1))/(2*M1*M1*G1));
IF B1<=1 AND B1>=-1 THEN
BEGIN
X:=-ARCCOS(B1)*COF;Y:=G;CHARMD;LINETO(X,Y);WRITE(GRS,7,M,X,Y);
END#
STARTP;DEVICE(3,1,0,0);
XRANGE(-250,1250);YRANGE(-250,1250);
COF:=180/3.14157;
ICOF:=3.14157/180;
WRITE(FILE97,"<"GAIN RANGE AND INCREMENT?">);
READ(FILE97,/,LG,HG,G1);
WRITE(FILE97,"<"PHASE RANGE AND INCREMENT?">);
READ(FILE97,/,LP,HP,PI);
WRITE(FILE97,"<"GAIN RANGE VARIATION AND INCREMENT?">);
READ(FILE97,/,GAIN1,GAIN2,GAININ);
WRITE(FILE97,"<"PHASE RANGE VARIATION AND INCREMENT?">);
READ(FILE97,/,FA1,FA2,FAI);
WRITE(FILE97,"<"RANGE AND INCREMENT OF CONSTANT GAIN?">);
READ(FILE97,/,ML,MH,MI);
WRITE(FILE97,"<"RANGE AND INCREMENT OF CONSTANT PHASE?">);
READ(FILE97,/,PHAL,PHAH,PHAI);
WRITE(FILE97,"<"FREQUENCY RANGE AND INCREMENT?">);
READ(FILE97,/,LF,HF,FI);
FRAME("GRAPH");
AT(30,1000);WRTARY(1,T1);AT(1000,30);WRTARY(1,T2);
SUBFRM(80,80,980,980);
YRANGE(LG,HG);XRANGE(LP,HP);GRID;YAXIS(LP,G1);GRID;XAXIS(LG,PI);
NOLINE;
FOR M:=ML STEP MI UNTIL MH DO

```

126100
126200
126300
126400
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126800
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127000
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127200
127300
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127500
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129800
129900
130000
130100
130200
130300
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130900
131000
131100
131200
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131800
131900
132000
132100

```
BEGIN
NOLINE;
FOR G:=GAIN1 STEP GAININ UNTIL GAIN2 DO
  BEGIN
    GAINCONTOURS;
  END;
TXTDIR(0);
FLOATN(3,1,M);
END;
NOLINE;
NOLINE;
FOR F:=PHAL*ICOF STEP PHAI*ICOF UNTIL PHAH*ICOF DO
  BEGIN
    NOLINE;
    FOR P:=FA1*ICOF STEP FAI*ICOF UNTIL FA2*ICOF DO
      BEGIN
        PHASECONTOURS;
      END;
    F1:=F*COF;TXTDIR(0);FIXEDN(4,F1);
    END;
    IF DA<3 THEN
      BEGIN
        IF DB<3 THEN
          BEGIN
            W:=LF;NOLINE;
            FOR W:=LF STEP FI UNTIL HF DO
              BEGIN
                GAIN:=((R1(W,DA,A)**2+I1(W,DA,A)**2)/(R1(W,DB,B)**2+I1(W,DB,B)**2));
                I:=*+1;
                Y8[0]:=0;
                NNT:=(I1(W,DA,A)/R1(W,DA,A))-(I1(W,DB,B)/R1(W,DB,B));
                DNT:=1+(I1(W,DA,A)/R1(W,DA,A))*(I1(W,DB,B)/R1(W,DB,B));
                Y8[I]:=ARCTAN(NNT/DNT)*COF;
                IF Y8[I]<=0 THEN BEGIN
                  WRITE(U,/,W,GAIN,Y8[I]);
                  SIX(Y8);
                END ELSE
                  BEGIN
                    Y8[I]:=-180+Y8[I];
                    WRITE(U,/,W,GAIN,Y8[I]);
                    SIX(Y8);
                  END;
                END;
              END;
            END;
          END;
        END;
      END;
    ELSE IF DB>=4 THEN
      BEGIN
        W:=LF;NOLINE;
        FOR W:=LF STEP FI UNTIL HF DO
          BEGIN
            GAIN:=((R1(W,DA,A)**2+I1(W,DA,A)**2)/(R2(W,DB,B)**2+I2(W,DB,B)**2));
            I:=*+1;
            Y8[0]:=0;
            NNT:=(I1(W,DA,A)/R1(W,DA,A))-(I2(W,DB,B)/R2(W,DB,B));
            DNT:=1+(I1(W,DA,A)/R1(W,DA,A))*(I2(W,DB,B)/R2(W,DB,B));
            Y8[I]:=ARCTAN(NNT/DNT)*COF;
            IF Y8[I]<=0 THEN BEGIN
              WRITE(U,/,W,GAIN,Y8[I]);
              SIX(Y8);
            END ELSE
              BEGIN
                Y8[I]:=-180+Y8[I];
                WRITE(U,/,W,GAIN,Y8[I]);
                SIX(Y8);
              END;
            END;
          END;
        END;
      END;
    END;
  END;
END
```

00126100
00126200
00126300
00126400
00126500
00126600
00126700
00126800
00126900
00127000
00127100
00127200
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00127400
00127500
00127600
00127700
00127800
00127900
00128000
00128100
00128200
00128300
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00129200
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00130000
00130100
00130200
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00132100

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136100
136200
136300
136400
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136700
136800
136900
137000
137100
137200
137300
137400
137500
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137700
137800
137900
138000
138100
138200

```
Y8[I]:=-180+Y8[I];  
WRITE(U,/,W,GAIN,Y8[I]);  
SIX(Y8);  
END;  
END;  
NICHOL;  
END;  
END;  
ELSE  
IF DA=4 THEN  
BEGIN  
IF DB=3 THEN  
BEGIN  
W:=LF;NOLINE;  
FOR W:=LF STEP FI UNTIL HF DO  
BEGIN  
GAIN:=(R2(W,DA,A)**2+I2(W,DA,A)**2)/(R1(W,DB,B)**2+I1(W,DB,B)**2));  
I:=I+1;  
Y8[O]:=0;  
NNT:=(I2(W,DA,A)/R2(W,DA,A))-(I1(W,DB,B)/R1(W,DB,B));  
DNT:=1+(I2(W,DA,A)/R2(W,DA,A))*(I1(W,DB,B)/R1(W,DB,B));  
Y8[I]:=ARCTAN(NNT/DNT)*COF;  
IF Y8[I]<=0 THEN BEGIN  
WRITE(U,/,W,GAIN,Y8[I]);  
SIX(Y8);  
END ELSE  
BEGIN  
Y8[I]:=-180+Y8[I];  
WRITE(U,/,W,GAIN,Y8[I]);  
SIX(Y8);  
END;  
END;  
NICHOL;  
END;  
ELSE  
IF DB=4 THEN  
BEGIN  
W:=LF;NOLINE;  
FOR W:=LF STEP FI UNTIL HF DO  
BEGIN  
GAIN:=(R2(W,DA,A)**2+I2(W,DA,A)**2)/(R2(W,DB,B)**2+I2(W,DB,B)**2));  
I:=I+1;  
Y8[O]:=0;  
NNT:=(I2(W,DA,A)/R2(W,DA,A))-(I2(W,DB,B)/R2(W,DB,B));  
DNT:=1+(I2(W,DA,A)/R2(W,DA,A))*(I2(W,DB,B)/R2(W,DB,B));  
Y8[I]:=ARCTAN(NNT/DNT)*COF;  
IF Y8[I]<=0 THEN BEGIN  
WRITE(U,/,W,GAIN,Y8[I]);  
SIX(Y8);  
END ELSE  
BEGIN  
Y8[I]:=-180+Y8[I];  
WRITE(U,/,W,GAIN,Y8[I]);  
SIX(Y8);  
END;  
END;  
NICHOL;  
END;  
END;  
WHEN(10);  
WRITE(FILE97, <"DO YOU WANT TO CHANGE THE SCALE?">);
```

00132200
00132300
00132400
00132500
00132600
00132700
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00133400
00133500
00133600
00133700
00133800
00133900
00134000
00134100
00134200
00134300
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00136900
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00137500
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00138000
00138100
00138200

```

138300 READ(FILE#97,/,SC);
138400 IF SC="Y" THEN NICHOLS(W,DA,DB,A,B);
138500 END;
138600 COMMENT **** PROCEDURE RLOCUS ****;
138700 PROCEDURE RLOCUS(DPOLY1,DPOLY2,POLY1,POLY2);
138800 VALUE DPOLY1,DPOLY2,REAL DPOLY1,DPOLY2;
138900 ARRAY POLY2,POLY1(0);
139000 BEGIN
139100 FILE E(KIND=DISK,MAXRECSIZE=40);
139200 REAL X,Y,SC,GR,K1,KI,J,K;
139300 REAL LY,HY,YI,LX,HX,XI;
139400 VALUE ARRAY T1("**ROOT LOGI**");
139500 PROCEDURE REALPOLYZEROFINDER(DEGREE,P,ZEROR,ZEROI);
139600 VALUE DEGREE,INTEGER DEGREE,ARRAY P,ZEROR,ZEROI(0),FORWARD;
139700 PROCEDURE REALPOLYZEROFINDER(DEGREE,P,ZEROR,ZEROI);
139800 VALUE DEGREE;
139900 INTEGER DEGREE;
140000 ARRAY P,ZEROR,ZEROI(0);
140100 COMMENT PURPOSE
140200
140300 FIND ALL THE ZERUS OF A REAL POLYNOMIAL P
140400
140500 DESCRIPTION OF PARAMETERS
140600
140700 DEGREE = DEGREE OF THE POLYNOMIAL
140800 P = ARRAY OF COEFFICIENTS OF THE POLYNOMIAL, THE
140900 COEFFICIENT OF Z*J BEING STORED IN P[N-J]
141000 DIMENSION P(0:DEGREE)
141100 ZEROR = ARRAY OF REAL PART OF ZEROS OF THE POLYNOMIAL,
141200 THE REAL PART OF THE K-TH ZERO BEING STORED
141300 IN ZEROR[K]
141400 DIMENSION ZEROR(0:DEGREE)
141500 ZEROI = ARRAY OF IMAGINARY PARTS OF ZEROS OF THE
141600 POLYNOMIAL, THE IMAGINARY PART OF K-TH ZERO
141700 BEING STORED IN ZEROI[K]
141800 DIMENSION ZEROI(0:DEGREE)
141900
142000 METHOD
142100
142200 A THREE-STAGE VARIABLE-SHIFT ITERATION DUE TO M.A.
142300 JENKINS AND J.F. TRAUB IS USED. THE ZEROS ARE CALCULATED ONE OR
142400 TWO AT A TIME IN ROUGHLY INCREASING ORDER OF MAGNITUDE USING REAL
142500 ARITHMETIC WHEREVER POSSIBLE. FAILURE OF THE ITERATION TO
142600 CONVERGE FOR A ROOT WILL CAUSE AN ERROR MESSAGE TO BE PRINTED AND
142700 THE PROCEDURE EXITED WITH THE ZERUES ALREADY FOUND STORED IN THE
142800 ARRAYS ZEROR AND ZEROI. FAILURES WILL BE RARE.
142900
143000 REFERENCE
143100
143200 THIS PROCEDURE IS DUE TO M.A. JENKINS AND APPEARS IN
143300 "THREE-STAGE VARIABLE-SHIFT ITERATIONS FOR THE SOLUTION OF
143400 POLYNOMIAL EQUATIONS WITH A POSTERIORI ERROR BOUNDS FOR THE ZERUES"
143500 TECHNICAL REPORT NO. CS 138, PP 125-133, AUGUST 1969,
143600 COMPUTER SCIENCE DEPT., STANFORD UNIVERSITY, STANFORD,CALIF 94305 ;
143700 BEGIN
143800 COMMENT INITIALIZATION OF MACHINE DEPENDENT CONSTANTS ;
143900 DEFINE BASE = 8 #, % FLOATING POINT ARITHMETIC IS BASE 8
144000 PRECISION = 13 #, % NO. OF OCTAL DIGITS IN NUMBER
144100 % REPRESENTATION
144200 EXPRNGHI = 63 #, % UPPER EXPONENT RANGE
144300 LOEXPRNG = -51 #, % LOWER EXPONENT RANGE

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144400		ROUNDED = TRUE #, % ARITHMETIC IS ROUNDED	00144400
144500		ETA = 7.27595761420e-12 #, % RELATIVE REPRESENTATION ERROR	00144500
144600		INFINITY = 4.31359146672e68 #, % LARGEST FLOATING PT. NO.	00144600
144700		SMALLESTNO = 8.75811540210e-47 #, % SMALLEST FL. PT. NO.	00144700
144800		ZETA = 5.49755813890e11 #, % ZETA = BASE**PRECISION	00144800
144900		CONSTL = 3.85185988877e-34 #,	00144900
145000	%	LNBASE = CONSTL = ZETA * SMALLESTNO * BASE	00145000
145100		MRE = 7.27595961420e-12 #, % LN(8)	00145100
145200		MAX. RELATIVE ROUNDOFF ERROR.	00145200
145300		% FOR FL. PT. ADDITION AND MULT.	00145300
145400		ARE = 7.27595761420e-12 #, % ARE SAME ON B6700	00145400
145500		W1 = 1.45519152284e-11 #, % W1 = MRE+ARE	00145500
145600		W2 = 7.27595761420e-12 #, % W2 = MRE	00145600
145700		W3 = 6.54836185280e-11 #, % W3 = 5*MRE + 4*ARE	00145700
145800		W4 = 5.09317032993e-11 #, % W4 = 5*MRE + 2*ARE	00145800
145900		W5 = 1.45519152284e-11 #, % W5 = 2*MRE	00145900
146000	REAL	M1,M2,M3,M4,SR,SI,A,B,C,D,E,F,G,H,A1,A2,A3,A6,A7,	00146000
146100		MP,U,V,DU,DV,UI,VI,SVU,SVV;	00146100
146200	INTEGER	N,TYPE,ND,RANDOMSTART;	00146200
146300	ARRAY	QPIO:DEGREE, K,QK,SVK(0:DEGREE-1);	00146300
146400	LABEL	NEXTZERU,FAILURE,EXIT;	00146400
146500	FILE	PRINTER (KIND = 7, BUFFERS = 1, MAXRECSIZE = 15);	00146500
146600	FORMAT	ERR("FAILURE TO CONVERGE OCCURRED ON ",I2,"-TH ZERO");	00146600
146700	REAL PROCEDURE	CMD(A,B);	00146700
146800	VALUE	A,B;	00146800
146900	REAL	A,B;	00146900
147000	COMMENT	COMPLEX MODULUS OF (A,B);	00147000
147100		BEGIN	00147100
147200		A := ABS(A);	00147200
147300		B := ABS(B);	00147300
147400		CMOD := IF A LSS B THEN SQRT(1+(A/B)**2) * B ELSE	00147400
147500		IF A GTR B THEN SQRT(1+(B/A)**2) * A ELSE	00147500
147600		1.41421356237 * A; COMMENT SQRT(2);	00147600
147700		END CMOD;	00147700
147800	PROCEDURE	CDIVIDE(AR,AI,BR,BI,CR,CI);	00147800
147900	VALUE	AR,AI,BR,BI;	00147900
148000	REAL	AR,AI,BR,BI,CR,CI;	00148000
148100	COMMENT	COMPLEX DIVISION - (CR,CI) = (AR,AI)/(BR,BI);	00148100
148200		BEGIN	00148200
148300	REAL	D,R;	00148300
148400		IF BR = 0 AND BI = 0 THEN CR := CI := INFINITY ELSE	00148400
148500		IF ABS(BR) LSS ABS(BI) THEN	00148500
148600		BEGIN	00148600
148700		R := BR/BI;	00148700
148800		D := BR * R + BI;	00148800
148900		CR := (AR * R + AI)/D;	00148900
149000		CI := (AI * R - AR)/D;	00149000
149100		END	00149100
149200		ELSE	00149200
149300		BEGIN	00149300
149400		R := BI/BR;	00149400
149500		D := BI * R + BR;	00149500
149600		CR := (AI * R + AR)/D;	00149600
149700		CI := (AI - AR * R)/D;	00149700
149800		END	00149800
149900	PROCEDURE	CDIVIDE;	00149900
150000	VALUE	PRESCALE(N,P);	00150000
150100	INTEGER	N;	00150100
150200	ARRAY	P[0];	00150200
150300	COMMENT	SCALES THE POLYNOMIAL TO AVOID OVERFLOW AND UNDERFLOW IF	00150300
150400			00150400

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REAL
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REAL PROCEDURE
VALUE
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ARRAY
COMMENT

INTEGER
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ARRAY
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COMMENT
COMMENT

COMMENT
L:

COMMENT

```
POSSIBLE ;
BEGIN
  MAXM, MINM, S, SCALE;
  L := 1;
  MAXM := 0;
  MINM := INFINITY;
  FOR I := 0 STEP 1 UNTIL N DO
  BEGIN
    S := ABS(P[I]);
    MAXM := MAX(MAXM, S);
    IF S NEQ 0 THEN MINM := MIN(MINM, S);
  END;
  IF MINM LSS CONSTL OR MAXM GTR BASE ** (EXPRNGHI DIV 2)
  THEN
  BEGIN
    S := CONSTL / MINM;
    SCALE := IF S GTR 1 THEN (IF INFINITY / S LSS MAXM THEN 1.0
    ELSE S) ELSE (1 / SQRT(MAXM)) * (1 / SQRT(MINM));
    L := LN(SCALE) / LN(BASE);
    IF L NEQ 0 THEN
    BEGIN
      S := BASE ** L;
      FOR I := 0 STEP 1 UNTIL N DO
        P[I] := P[I] * S;
    END;
  END;
PRESCALE;
PROCEDURE CAUCHYBOUND(N, P);
N;
N;
P[0];
THIS PROCEDURE COMPUTES A LOWER BOUND ON THE MODULUS OF
THE ZEROS OF THE POLYNOMIAL;
BEGIN
  I;
  L;
  Q, PT[0:N];
  X, XM, DX, F, DF, BND;
  FOR I := 0 STEP 1 UNTIL N - 1 DO PT[I] := ABS(P[I]);
  PT[N] := -ABS(P[N]);
  INITIAL GUESS FOR LOWER BOUND;
  X := EXP((LN(-PT[N]) - LN(PT[0])) / N);
  COMMENT
  IF THE NEWTON STEP AT ZERO IS BETTER USE IT AS THE
  INITIAL GUESS;
  IF PT[N-1] NEQ 0 THEN
  BEGIN
    XM := -PT[N] / PT[N-1];
    X := MIN(X, XM);
  END;
  COMMENT
  CHOP THE INTERVAL (0, X) UNTIL F LEQ 0;
  L:
  XM := X * 0.1;
  F := PT[0];
  FOR I := 1 STEP 1 UNTIL N DO F := F * XM + PT[I];
  BEGIN
    IF F GTR 0 THEN
    BEGIN
      X := XM;
      GO TO L;
    END;
  END;
  COMMENT
  USE NEWTON RAPHSON TO FIND THE LOWER BOUND;
  DX := X;
  FOR X := X WHILE ABS(DX/X) GTR .005 DO
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BEGIN
  Q[0] := PT[0] ;
  FOR I := 1 STEP 1 UNTIL N DO Q[I] := Q[I-1] * X + PT[I] ;
  F := Q[N] ;
  DF := Q[0] ;
  FOR I := 1 STEP 1 UNTIL N-1 DO DF := DF * X + Q[I] ;
  DX := F/DF ;
  X := X - DX ;
END ;
CAUCHYBOUND := X ;
END CAUCHYBOUND ;
PROCEDURE QUADRATIC(A,B,C,SZR,SZI,LZR,LZI) ;
VALUE A,B,C ;
REAL A,B,C,SZR,SZI,LZR,LZI ;
COMMENT FINDS THE ROOTS OF THE QUADRATIC  $AIZ^2+2IBIZ+C$ , AVOIDING
UNNECESSARY OVERFLOW ;
BEGIN
  REAL D,E ;
  IF A = 0 THEN
  BEGIN
    SZR := SZI := LZI := 0 ;
    LZR := IF B = 0 THEN 0 ELSE -C/(2*B)
  END
  ELSE
    IF C = 0 AND B = 0 THEN SZR := SZI := LZR := LZI := 0 ELSE
  BEGIN
    IF ABS(B) GEQ ABS(C) THEN
    BEGIN
      E := 1 - (A/B) * (C/B) ;
      D := SQRT(ABS(E)) * ABS(B)
    END
    ELSE
    BEGIN
      E := B * (B/ABS(C)) - (IF C GTR 0 THEN A ELSE -A) ;
      D := SQRT(ABS(E)) * SQRT(ABS(C))
    END
  END ;
  IF E GEQ 0 THEN
  BEGIN
    LZR := (IF B GEQ 0 THEN -B-D ELSE -B+D)/A ;
    SZR := IF LZR = 0 THEN 0 ELSE (C/LZR)/A ;
    LZI := SZI := 0
  END
  ELSE
  BEGIN
    SZR := LZR := -B/A ;
    SZI := ABS(D/A) ;
    LZI := -SZI
  END
  ;
END
END QUADRATIC ;
BOOLEAN PROCEDURE PEVALANDTESTREAL(SR,QP) ;
VALUE SR ;
REAL SR ;
ARRAY QP[0] ;
COMMENT EVALUATES P AT SR BY SYNTHETIC DIVISION PUTTING THE
QUOTIENT POLYNOMIAL IN QP. AN UPPERBOUND ON THE ROUND OFF
ERROR IS COMPUTED. THE PROCEDURE IS TRUE IF THE MODULUS
OF THE POLYNOMIAL VALUE IS OF THE SAME ORDER OF MAGNITUDE
AS THE BOUND ON THE ROUND OFF ERROR ;
BEGIN
  REAL E,MS ;
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162700	INTEGER	I ;	00162700
162800		QP[0] := P[0] ;	00162800
162900		FOR I := 1 STEP 1 UNTIL N DO QP[I] := QP[I-1] * SR + P[I] ;	00162900
163000		MS := ABS(SR) ;	00163000
163100		E := ABS(QP[0]) * W1/W2 ;	00163100
163200		FOR I := 1 STEP 1 UNTIL N DO E := E * MS + ABS(QP[I]) ;	00163200
163300		MP := ABS(QP[N]) ;	00163300
163400		PEVALANDTESTREAL := MP LEQ 20 * (E * W2 - MP * W1)	00163400
163500	END	PEVALANDTESTREAL ;	00163500
163600	BOOLEAN PROCEDURE	COMPLEXPEVALANDTEST ;	00163600
163700	COMMENT	EVALUATES P AT (SR,SI) USING THE MORE GENERAL FORM OF	00163700
163800		SYNTHETIC DIVISION WHICH DIVIDES OUT A QUADRATIC FACTOR.	00163800
163900		THE QUOTIENT POLYNOMIAL IS QP AND THE REMAINDER IS	00163900
164000		A I Z + B. THE PROCEDURE IS TRUE IF THERE TWO ZEROS OF P	00164000
164100		WHICH FORM A QUADRATIC FACTOR CLOSE TO Z * 2 + U I Z + V.	00164100
164200		SEPARATE TESTS ARE MADE FOR A PAIR OF COMPLEX CONJUGATE	00164200
164300		ZERUS AND FOR TWO REAL ZEROS ;	00164300
164400	BEGIN		00164400
164500	REAL	E,MS ;	00164500
164600	ARRAY	GARB[0:N] ;	00164600
164700	BOOLEAN	B1 ;	00164700
164800	INTEGER	I ;	00164800
164900		QP[0] := P[0] ;	00164900
165000		QP[1] := P[1] - QP[0] * U ;	00165000
165100		FOR I := 2 STEP 1 UNTIL N DO	00165100
165200		QP[I] := P[I] - QP[I-1] * U - QP[I-2] * V ;	00165200
165300		A := QP[N] ;	00165300
165400		B := QP[N-1] ;	00165400
165500		QUADRATIC(1,U/2,V,M1,M2,M3,M4) ;	00165500
165600		IF M1 = M3 THEN	00165600
165700	BEGIN		00165700
165800		MS := SQRT(V) ;	00165800
165900		E := ABS(QP[0]) * W4 / W3 ;	00165900
166000		MP := M1 * B ;	00166000
166100		FOR I := 1 STEP 1 UNTIL N-1 DO	00166100
166200		E := E * MS + ABS(QP[I]) ;	00166200
166300		E := E * MS + ABS(A + MP) ;	00166300
166400		E := E * W3 - W4 * (ABS(A+MP) + ABS(B) * MS) + W5*ABS(MP) ;	00166400
166500		MP := CMOD(A + MP, M2 * B) ;	00166500
166600		B1 := MP LEQ 20 * E	00166600
166700	END		00166700
166800		ELSE	00166800
166900		B1 := PEVALANDTESTREAL(M1,GARB) AND	00166900
167000		PEVALANDTESTREAL(M3,GARB) ;	00167000
167100		MP := CMOD(A,B) ;	00167100
167200		COMPLEXPEVALANDTEST := B1	00167200
167300	END C	COMPLEXPEVALANDTEST ;	00167300
167400	PROCEDURE	DEFLATE(I) ;	00167400
167500	VALUE	I ;	00167500
167600	INTEGER	I ;	00167600
167700	COMMENT	DEFLATE STORES THE ZERO OR ZEROS FOUND BY THE PROGRAM,	00167700
167800		REPLACES P WITH THE DEFLATED POLYNOMIAL, AND REDUCES N ;	00167800
167900	BEGIN		00167900
168000		IF I =2 THEN	00168000
168100	BEGIN		00168100
168200		ZEROR[ND] := M1 ;	00168200
168300		ZERUI[ND] := M2 ;	00168300
168400		ZEROR[ND + 1] := M3 ;	00168400
168500		ZERUI[ND + 1] := M4 ;	00168500
168600		N := N - 2	00168600
168700	END		00168700

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PROCEDURE
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ELSE  
BEGIN  
  ZEROR[ND] := SR ;  
  ZEROI[ND] := 0 ;  
  N := N - 1  
END ;  
WRITE(P[*],N+1,QP[*]);  
GO TO NEXTZERO ; COMMENT IN MAIN ROUTINE ;  
END DEFLATE ;  
CALCULATECOEFF ;  
EVALUATES K(S) AND CALCULATES THE COEFFICIENTS NEEDED TO  
COMPUTE THE THE NEXT K POLYNOMIAL ;  
BEGIN  
  I ;  
  QK[0] := K[0] ;  
  QK[1] := K[1] - QK[0] * U ;  
  FOR I := 2 STEP 1 UNTIL N-1 DO  
  QK[I] := K[I] - QK[I-1] * U - QK[I-2] * V ;  
  C := QK[N-1] ;  
  D := QK[N-2] ;  
  TYPE := IF ABS(C) LEQ ABS(K[N-1]) * 10 * ETA THEN 3 ELSE  
  IF ABS(D) GEQ ABS(C) THEN 2 ELSE 1 ;  
  IF TYPE = 1 THEN  
  BEGIN  
    E := A/C ;  
    F := D/C ;  
    G := U * E ;  
    H := V * B ;  
    A3 := A2 := A * E + G * B + H * (B/C) ;  
    A1 := B - A * (D/C) ;  
    A6 := A7 := G * D + H * F + A  
  END  
  ELSE IF TYPE = 2 THEN  
  BEGIN  
    E := A/D ;  
    F := C/D ;  
    G := U * B ;  
    H := V * B ;  
    A3 := A2 := A * E + E * G + H * (B/D) ;  
    A1 := B * F - A ;  
    A6 := A7 := A * F + U * A + H  
  END  
  ;  
  NEXTESTIMATES ;  
  COMPUTES THE NEXT U AND V USING THE RESULTS OF  
  CALCULATECOEFF ;  
  BEGIN  
    A4,A5,B1,B2,C1,C2,C3,C4,TEMP ;  
    IF TYPE = 3 THEN UI := VI := 0 ELSE  
  BEGIN  
    IF TYPE = 1 THEN  
  BEGIN  
    A4 := U * B + H * F + A ;  
    A5 := V * (U * F) + U * D + C  
  END  
  ELSE  
  BEGIN  
    A4 := A * F + G * F + H ;  
    A5 := C * F + U * C + V * D  
  END  
  ;  
  B1 := - K[N-1]/P[N] ;
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174900		B2 := -(K[N-2] + P[N-1] * B1)/P[N] ;	00174900
175000		C1 := V * B2 * A1 ;	00175000
175100		C2 := B1 * A6 ;	00175100
175200		C3 := B1 * (B1 * A2) ;	00175200
175300		C4 := C1 - C2 - C3 ;	00175300
175400		TEMP := B1 * A4 + A5 - C4 ;	00175400
175500		IF TEMP = 0 THEN UI := VI := 0 ELSE	00175500
175600	BEGIN	UI := U - (U * (C3 + C2) + V * (B1 * A1 + B2 * A6))/TEMP ;	00175600
175700		VI := V * (C4/TEMP) + V	00175700
175800			00175800
175900	END		00175900
176000	END		00176000
176100	END	NEXTESTIMATES ;	00176100
176200	PROCEDURE	NEXTKPOLY ;	00176200
176300	COMMENT	CALCULATES THE NEXT K POLYNOMIAL USING THE RESULTS FROM	00176300
176400		CALCULATECOEFF ;	00176400
176500	BEGIN		00176500
176600	INTEGER	I ;	00176600
176700		IF TYPE LSS 3 THEN	00176700
176800	BEGIN		00176800
176900		IF ABS(A1) LEQ ABS(IF TYPE = 1 THEN B ELSE A) * ETA * 10	00176900
177000		THEN	00177000
177100	BEGIN		00177100
177200		FOR I := N-1 STEP -1 UNTIL 2 DO	00177200
177300		K[I] := QK[I-2] * (A3/QP[0]) - QP[I-1] * (A7/QP[0]) ;	00177300
177400		K[1] := A7 ;	00177400
177500		K[0] := 0	00177500
177600	END		00177600
177700		ELSE	00177700
177800	BEGIN		00177800
177900		A7 := A6/A1 ;	00177900
178000		A3 := A2/A1 ;	00178000
178100		FOR I := N-1 STEP -1 UNTIL 2 DO	00178100
178200		K[I] := QK[I-2] * A3 - QP[I-1] * A7 + QP[I] ;	00178200
178300		K[1] := QP[0] * A7 + QP[1] ;	00178300
178400		K[0] := QP[0]	00178400
178500	END		00178500
178600	END		00178600
178700		ELSE	00178700
178800	BEGIN		00178800
178900		FOR I := N-1 STEP -1 UNTIL 2 DO	00178900
179000		K[I] := QK[I-2] ;	00179000
179100		K[1] := K[0] := 0	00179100
179200	END		00179200
179300	END	NEXTKPOLY ;	00179300
179400	PROCEDURE	NOSHIFTPROCESS ;	00179400
179500	COMMENT	INITIALIZES K AND COMPUTES 5 NO SHIFT K POLYNOMIALS ;	00179500
179600	BEGIN		00179600
179700	INTEGER	I, J ;	00179700
179800	REAL	A, C, T ;	00179800
179900		FOR I := 0 STEP 1 UNTIL N-1 DO K[I] := P[I] * (N-I)/N ;	00179900
180000		A := P[N] ;	00180000
180100		B := P[N-1] ;	00180100
180200		FOR J := 1 STEP 1 UNTIL 5 DO	00180200
180300	BEGIN		00180300
180400		C := K[N-1] ;	00180400
180500		IF ABS(C) LEQ ABS(B) * ETA * 10 THEN	00180500
180600	BEGIN		00180600
180700		FOR I := N-1 STEP -1 UNTIL 1 DO K[I] := K[I-1] ;	00180700
180800		K[0] := 0	00180800
180900	END		00180900

181000		ELSE		00181000
181100		BEGIN		00181100
181200		T := - A/C ;		00181200
181300		FOR I := N-1 STEP -1 UNTIL 1 DO K[I] := K[I-1] * T + P[I] ;		00181300
181400		K[0] := P[0]		00181400
181500		END		00181500
181600		END	J	00181600
181700		END	NOSHIFTPROCESS ;	00181700
181800	PROCEDURE	CPLXITERATION ;		00181800
181900	COMMENT	ATTEMPTS 10 STEPS WITH THE VARIABLE-SHIFT PROCESS FOR A		00181900
182000		QUADRATIC FACTOR ;		00182000
182100		BEGIN		00182100
182200	REAL	DMP ;		00182200
182300	INTEGER	J ;		00182300
182400	LABEL	EXIT ;		00182400
182500		IF COMPLEXPEVALANDTEST THEN DEFLATE(2) ;		00182500
182600		FOR J := 1 STEP 1 UNTIL 10 DO		00182600
182700		BEGIN		00182700
182800		IF J GTR 2 AND MP GTR 10 * DMP THEN GO TO EXIT ;		00182800
182900		DMP := MP ;		00182900
183000		CALCULATECOEFF ;		00183000
183100		NEXTKPOLY ;		00183100
183200		CALCULATECOEFF ;		00183200
183300		NEXTESTIMATES ;		00183300
183400		IF VI NEQ 0 THEN		00183400
183500		BEGIN		00183500
183600		U := UI ;		00183600
183700		V := VI		00183700
183800		END		00183800
183900		IF COMPLEXPEVALANDTEST THEN DEFLATE(2) ;		00183900
184000		END	J ;	00184000
184100	EXIT:	END	CPLXITERATION ;	00184100
184200	PROCEDURE	REALITERATION ;		00184200
184300	COMMENT	ATTEMPTS 10 STEPS WITH THE VARIABLE-SHIFT PROCESS FOR A		00184300
184400		REAL ZERO ;		00184400
184500		BEGIN		00184500
184600	INTEGER	I, J ;		00184600
184700	REAL	T, DMP, KS ;		00184700
184800	LABEL	EXIT ;		00184800
184900		IF PEVALANDIESTREAL(SR, QP) THEN DEFLATE(1) ;		00184900
185000		FOR J := 1 STEP 1 UNTIL 10 DO		00185000
185100		BEGIN		00185100
185200		IF J GTR 2 THEN		00185200
185300		BEGIN		00185300
185400		IF ABS(T) < 0.01 * ABS(SR-T) AND MP > DMP THEN		00185400
185500		BEGIN		00185500
185600	COMMENT	A CLUSTER OF ZEROS NEAR THE REAL AXIS HAS BEEN ENCOUNTERED		00185600
185700		WE ATTEMPT TO FIND A QUADRATIC FACTOR ;		00185700
185800		U := -2 * SR ;		00185800
185900		V := SR * SR ;		00185900
186000		CPLXITERATION ;		00186000
186100	COMMENT	RETURNS ONLY IF NO QUADRATIC FACTOR IS FOUND ;		00186100
186200		GO TO EXIT		00186200
186300		END		00186300
186400		ELSE		00186400
186500		IF MP GTR 10 * DMP THEN GO TO EXIT		00186500
186600		END		00186600
186700		DMP := MP ;		00186700
186800		QK[0] := K[0] ;		00186800
186900		FOR I := 1 STEP 1 UNTIL N-1 DO		00186900
187000		QK[I] := QK[I-1] * SR + K[I] ;		00187000

187100		IF ABS(QK[N-1]) LEQ ABS(K[N-1]) * 10 * ETA THEN	00187100
187200		BEGIN	00187200
187300		FOR I := N-1 STEP -1 UNTIL 1 DO K[I] := QK[I-1] ;	00187300
187400		K[0] := 0	00187400
187500		END	00187500
187600		ELSE	00187600
187700		BEGIN	00187700
187800		T := -QP[N]/QK[N-1] ;	00187800
187900		FOR I := N-1 STEP -1 UNTIL 1 DO K[I] := QK[I-1] * T + QP[I] ;	00187900
188000		K[0] := QP[0]	00188000
188100		END	00188100
188200		KS := K[0] ;	00188200
188300		FOR I := 1 STEP 1 UNTIL N-1 DO KS := KS * SR + K[I] ;	00188300
188400		T := IF ABS(KS) LEQ ABS(K[N-1]) * 10 * ETA THEN 0 ELSE	00188400
188500		= QP[N]/KS ;	00188500
188600		SR := SR + 1 ;	00188600
188700		IF PEVALANDTESTREAL(SR, QP) THEN DEFLATE(1)	00188700
188800		J ;	00188800
188900		END	00188900
189000	EXIT:	REALITERATION ;	00189000
189100	PROCEDURE	FIXEDSHIFTPROCESS(LIMIT) ;	00189100
189200	VALUE	LIMIT ;	00189200
189300	INTEGER	LIMIT ;	00189300
189400	COMMENT	CARRIES OUT THE STAGE TWO CALCULATION OF K POLYNOMIALS	00189400
189500		WITH TESTING. LIMIT IS THE MAXIMUM NUMBER OF STEPS ;	00189500
189600	REAL	BEGIN	00189600
189700	BOOLEAN	TR, TI, OTR, OTI, OVI, SVU, SVV ;	00189700
189800	INTEGER	TEST, RTEST, PASSED, RPASSED ;	00189800
189900		I, J ;	00189900
190000		U := -2 * SR ;	00190000
190100		V := SR * SR + SI * SI ;	00190100
190200		TEST := RTEST := TRUE ;	00190200
190300		PASSED := RPASSED := FALSE ;	00190300
190400		IF COMPLEXPEVALANDTEST THEN DEFLATE(2) ;	00190400
190500		CALCULATECOEFF ;	00190500
190600		FOR J := 1 STEP 1 UNTIL LIMIT DO	00190600
190700		BEGIN	00190700
190800		NEXTKPOLY ;	00190800
190900		CALCULATECOEFF ;	00190900
191000		NEXTTESTIMATES ;	00191000
191100		IF TYPE = 3 OR K[0] = 0 THEN TR := TI := 0 ELSE	00191100
191200		CDIVIDE(SR * B - A, -SI * B, C - SR * D, SI * D, TR, TI) ;	00191200
191300		IF RTEST AND J GTR 1 THEN	00191300
191400		BEGIN	00191400
191500		IF CMOD(TR - OTR, TI - OTI) LSS 0.5 * CMOD(SR + TR, SI + TI)	00191500
191600		THEN	00191600
191700		BEGIN	00191700
191800		IF RPASSED THEN	00191800
191900		BEGIN	00191900
192000		SVU := SR ;	00192000
192100		WRITE(SVK[*], N, K[*]) ;	00192100
192200		SR := SR + TR ;	00192200
192300	COMMENT	REALITERATION ;	00192300
192400		RETURNS ONLY IF THE ITERATION FAILS ;	00192400
192500		RTEST := FALSE ;	00192500
192600		SR := SVU ;	00192600
192700		WRITE(KL[*], N, SVK[*]) ;	00192700
192800		IF COMPLEXPEVALANDTEST THEN DEFLATE(2) ;	00192800
192900		END	00192900
193000		ELSE	00193000
193100		RPASSED := TRUE	00193100

193200		ELSE		00193200
193300		PASSED := FALSE		00193300
193400		END RTEST ;		00193400
193500		OIR := IR ;		00193500
193600		OTI := TI ;		00193600
193700		IF TEST AND J GTR 1 THEN		00193700
193800		BEGIN		00193800
193900		IF ABS(OVI-VI) LSS 0.5 * ABS(VI) THEN		00193900
194000		BEGIN		00194000
194100		IF PASSED THEN		00194100
194200		BEGIN		00194200
194300		SVU := U ;		00194300
194400		SVV := V ;		00194400
194500		WRITE(SVK[*],N,K[*]);		00194500
194600		U := UI ;		00194600
194700		V := VI ;		00194700
194800		CPLXITERATION ;		00194800
194900	COMMENT	RETURNS ONLY IF THE ITERATION FAILS ;		00194900
195000		TEST := FALSE ;		00195000
195100		U := SVU ;		00195100
195200		V := SVV ;		00195200
195300		WRITE(K[*],N,SVK[*]);		00195300
195400		IF COMPLEXPEVALANDTEST THEN DEFLATE(2) ;		00195400
195500		CALCULATECOEFF		00195500
195600		END		00195600
195700		ELSE		00195700
195800		PASSED := TRUE		00195800
195900		END		00195900
196000		ELSE		00196000
196100		PASSED := FALSE		00196100
196200		END TEST ;		00196200
196300		OVI := VI		00196300
196400		END J ;		00196400
196500		WRITE(SVK[*],N,K[*]);		00196500
196600		CPLXITERATION ;		00196600
196700	COMMENT	RETURNS ONLY IF THE ITERATION FAILS ;		00196700
196800		WRITE(K[*],N,SVK[*]);		00196800
196900		END FIXEDSHIFTPROCESS ;		00196900
197000	COMMENT	VARIABLES LOCAL TO MAIN ROUTINE ;		00197000
197100	REAL	BND,ANGLE ;		00197100
197200	INTEGER	LIMIT,CNT1,CNT2 ;		00197200
197300	ARRAY	INITIAL[0:DEGREE-1] ;		00197300
197400	COMMENT	BODY OF PROCEDURE REALPOLYZEROFINDER STARTS HERE ;		00197400
197500	COMMENT	SET INITIAL VALUE FOR PROCEDURE RANDOM ;		00197500
197600		RANDOMSTART := 17 ;		00197600
197700		N := DEGREE ;		00197700
197800		IF P[0] = 0 THEN GO TO FAILURE ;		00197800
197900		PRESCALE(N,P) ;		00197900
198000	NEXTZERO:	IF N = 1 THEN		00198000
198100	BEGIN	ZEROR[DEGREE] := - P[1]/P[0] ;		00198100
198200		ZEROI[DEGREE] := 0		00198200
198300		END		00198300
198400		ELSE		00198400
198500		IF N = 2 THEN		00198500
198600		QUADRATIC(P[0],P[1]/2,P[2],ZEROR[DEGREE-1],ZEROI[DEGREE-1]		00198600
198700		,ZEROR[DEGREE],ZEROI[DEGREE])		00198700
198800		ELSE		00198800
198900		IF N GTR 2 THEN		00198900
199000	BEGIN			00199000
199100	COMMENT	MAIN ALGORITHM FOR FINDING ONE OR TWO ZEROS ;		00199100
199200				00199200

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199300 ND := DEGREE - N + 1 ;
199400 IF P[N] = 0 THEN
199500 BEGIN
199600 ZEROR[ND] := ZEROI[ND] := 0 ;
199700 N := N-1 ;
199800 GO TO NEXTZERO
199900 END ;
200000 NOSHIFTPROCESS ;
200100 WRITE(INITIAL[*],N,K[*]);
200200 BND := CAUCHYBOUND(N,P) ;
200300 FOR CNT1 := 1,2 DO
200400 BEGIN
200500 LIMIT := 10 ;
200600 FOR CNT2 := 1 STEP 1 UNTIL 8 DO
200700 BEGIN
200800 ANGLE := RANDOM(RANDOMSTART) * 3.14159265359 ;
200900 SR := BND * COS(ANGLE) ;
201000 SI := BND * SIN(ANGLE) ;
201100 FIXEDSHIFTPROCESS(LIMIT) ;
201200 COMMENT IF WE RETURN FROM THE FIXED SHIFT STAGE THEN A SUCCESSFUL
201300 ITERATION HAS NOT BEEN STARTED. A NEW SHIFT IS SELECTED
201400 AND THE FIXED SHIFT STAGE IS INCREASED ;
201500 LIMIT := LIMIT + 10
201600 END CNT2 ;
201700 COMMENT IF 8 SHIFTS DO NOT SUCCEED WE TRY RESTARTING WITH THE
201800 INITIAL K POLYNOMIAL AND USE A DIFFERENT SEQUENCE OF
201900 SHIFTS ;
202000 WRITE(K[*],N,INITIAL[*]);
202100 END CNT1 ;
202200 GO TO FAILURE
202300 END OF MAIN ALGORITHM ;
202400 GO TO EXIT ;
202500 FAILURE: WRITE(PRINTER,ERR,DEGREE="N+1");
202600 EXIT:
202700 END REALPOLYZEROFINDER ;
202800 STARTP;
202900 X RANGE(-250,1250); Y RANGE(-250,1250);
203000 SUBFRM(0,0,1000,1000);
203100 DEVICE(3,1,0,0);
203200 BEGIN
203300 ARRAY ZR1,ZI1[0:DPOLY1];
203400 ARRAY ZR2,ZI2[0:DPOLY2];
203500 ARRAY POLY3,ZR3,ZI3[0:DPOLY2];
203600 ARRAY POLYK[0:DPOLY1];
203700 ARRAY K[0:0];
203800 WRITE(FILE97,"FINAL VALUE OF K AND K INCREMENT?">);
203900 READ(FILE97,/,K1,KI);
204000 WRITE(FILE97,"Y RANGE AND INCREMENT?">);
204100 READ(FILE97,/,LY,HY,YI);
204200 WRITE(FILE97,"X RANGE AND INCREMENT?">);
204300 READ(FILE97,/,LX,HX,XI);
204400 FRAME("GRAPH");
204500 AT(510,1000); TXDIR(0); TXSIZ(0); TXT("IM");
204600 AT(1000,1000); TXDIR(0); TXSIZ(0); TXT("RE");
204700 AT(20,1000); TXDIR(0); WRITARY(1,T1);
204800 YRANGE(LY,HY); XRANGE(LX,HX);
204900 XAXIS(LY+1,XI); YAXIS(0,YI);
205000 NDLIN;
205100 IF DPOLY1 >= 1 THEN
205200 BEGIN
205300 REALPOLYZEROFINDER(DPOLY1,POLY1,ZR1,ZI1);

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205400 FOR J:=1 STEP 1 UNTIL DPOLY1 DO
205500 BEGIN
205600 X:=ZR1[J];Y:=ZI1[J];
205700 AT(X,Y);SYMSEL(2);SYMSIZ(1);PUTSYM;
205800 END;
205900 END;
206000 NOLINE;
206100 REALPOLYZEROFINDER(DPOLY2,POLY2,ZR2,ZI2);
206200 FOR J:=1 STEP 1 UNTIL DPOLY2 DO
206300 BEGIN
206400 X:=ZR2[J];Y:=ZI2[J];
206500 AT(X,Y);SYMSEL(8);SYMSIZ(1);PUTSYM;
206600 END;
206700 NOLINE;
206800 FOR KI0:=0 STEP KI*1.1/20 UNTIL K1 DO
206900 BEGIN
207000 POLYMUL(DPOLY1,POLY1,0,K,DPOLY1,POLYK);
207100 POLYADD(DPOLY1,POLYK,DPOLY2,POLY2,DPOLY2,POLY3);
207200 REALPOLYZEROFINDER(DPOLY2,POLY3,ZR3,ZI3);
207300 FOR J:=1 STEP 1 UNTIL DPOLY2 DO
207400 BEGIN
207500 X:=ZR3[J];Y:=ZI3[J];
207600 WRITE(E,/K,X,Y);
207700 CHARM;
207800 AT(X,Y);LINETO(X,Y);
207900 END;
208000 END;
208100 WHEN(10);
208200 WRITE(FILE97,<"DO YOU WANT TO CHANGE THE SCALE?">);
208300 READ(FILE97,/SC);
208400 IF SC="Y" THEN RLOCUS(DPOLY1,DPOLY2,POLY1,POLY2);
208500 END;
208600 ENDP;
208700 END;
208800 DEFINE
208900 READK=
209000 WRITE(FILE97,<"VALUE OF K?">);
209100 READ(FILE97,/K10) #;
209200 PART2=
209300 WRITE(FILE97,<"COEFFICIENTS OF C1(S)?">);
209400 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DC1 DO C1[J]);
209500 WRITE(FILE97,<"COEFFICIENTS OF C2(S)?">);
209600 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DC2 DO C2[J]);
209700 WRITE(FILE97,<"COEFFICIENTS OF G1(S)?">);
209800 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DG1 DO G1[J]);
209900 WRITE(FILE97,<"COEFFICIENTS OF G2(S)?">);
210000 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DG2 DO G2[J]);
210100 WRITE(FILE97,<"COEFFICIENTS OF H1(S)?">);
210200 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DH1 DO H1[J]);
210300 WRITE(FILE97,<"COEFFICIENTS OF H2(S)?">);
210400 READ(FILE97,/FOR J:=0 STEP 1 UNTIL DH2 DO H2[J]) #;
210500 READCGH1=
210600 READK;
210700 WRITE(FILE97,<"DEGREE OF C1,C2,G1,G2,H1,H2?">);
210800 READ(FILE97,/DC1,DC2,DG1,DG2,DH1,DH2);
210900 DCG1:=DC1+DG1;DCG2:=DC2+DG2;DPOLY1:=DCG1+DH1;
211000 DA:=DPOLY1;DPOLY2:=DCG2+DH2;DB:=DPOLY2;
211100 BEGIN
211200 LABEL LAB2;
211300 ARRAY C1[0:DC1],C2[0:DC2],G1[0:DG1],G2[0:DG2];
211400 ARRAY H1[0:DH1],H2[0:DH2],CG1[0:DCG1],CG2[0:DCG2];
00205400
00205500
00205600
00205700
00205800
00205900
00206000
00206100
00206200
00206300
00206400
00206500
00206600
00206700
00206800
00206900
00207000
00207100
00207200
00207300
00207400
00207500
00207600
00207700
00207800
00207900
00208000
00208100
00208200
00208300
00208400
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00208900
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00209100
00209200
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00209400
00209500
00209600
00209700
00209800
00209900
00210000
00210100
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00210500
00210600
00210700
00210800
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00211000
00211100
00211200
00211300
00211400

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211500 ARRAY A,AA[0:DA],B,BB[0:DB],POLY1[0:DPOLY1],POLY2[0:DPOLY2];
211600 PART2;
211700 POLYMUL(DC1,C1,DG1,G1,DCG1,CG1);
211800 POLYMUL(DCG1,CG1,DH1,H1,DPOLY1,POLY1);
211900 POLYMUL(DPOLY1,POLY1,U,K,DA,A);
212000 POLYMUL(DC2,C2,DG2,G2,DCG2,CG2);
212100 POLYMUL(DCG2,CG2,DH2,H2,DPOLY2,POLY2);
212200 POLYMUL(DCG2,CG2,DH2,H2,DB,B);
212300 U[0]:=1;INB[0]:=1/B[0];
212400 IF B[0]=1 THEN
212500 BEGIN
212600 POLYMUL(DA,A,0,U,DA,AA);
212700 POLYMUL(DB,B,0,U,DB,BB);
212800 END
212900 ELSE
213000 BEGIN
213100 POLYMUL(DA,A,0,INB,DA,AA);
213200 POLYMUL(DB,B,0,INB,DB,BB);
213300 END#;
213400 READCGH2=
213500 READK;
213600 WRITE(FILE97,<"DEGREE OF C1,C2,G1,G2,H1,H2?">);
213700 READ(FILE97,/,DC1,DC2,DG1,DG2,DH1,DH2);
213800 DCG2:=DC2+DG2;DKCG1:=DC1+DG1;
213900 DA:=DC1+DG1+DH2;DPOL1:=DC1+DG1+DH1;DPOL2:=DC2+DG2+DH2;
214000 IF DPOL1>DPOL2 THEN DB:=DPOL1 ELSE DB:=DPOL2;
214100 BEGIN
214200 LABEL LAB2;
214300 ARRAY C1[0:DC1],C2[0:DC2],G1[0:DG1],G2[0:DG2];
214400 ARRAY H1[0:DH1],H2[0:DH2],KC1[0:DC1],A,AA[0:DA];
214500 ARRAY B,BB[0:DB],POL1[0:DPOL1],POL2[0:DPOL2];
214600 ARRAY CG2[0:DCG2],KCG1[0:DKCG1];
214700 PART2;
214800 POLYMUL(O,K,DC1,C1,DC1,KC1);
214900 POLYMUL(DC1,KC1,DG1,G1,DKCG1,KCG1);
215000 POLYMUL(DKCG1,KCG1,DH2,H2,DA,A);
215100 POLYMUL(DKCG1,KCG1,DH1,H1,DPOL1,POL1);
215200 POLYMUL(DC2,C2,DG2,G2,DCG2,CG2);
215300 POLYMUL(DCG2,CG2,DH2,H2,DPOL2,POL2);
215400 POLYADD(DPOL1,POL1,DPOL2,POL2,DB,B);
215500 U[0]:=1;INB[0]:=1/B[0];
215600 IF B[0]=1 THEN
215700 BEGIN
215800 POLYMUL(DA,A,0,U,DA,AA);
215900 POLYMUL(DB,B,0,U,DB,BB);
216000 WRITE(FILE97,/,FOR J:=0 STEP 1 UNTIL DB DO BB[J]);
216100 END
216200 ELSE
216300 BEGIN
216400 POLYMUL(DA,A,0,INB,DA,AA);
216500 WRITE(FILE97,/,FOR J:=0 STEP 1 UNTIL DA DO A[J]);
216600 POLYMUL(DB,B,0,INB,DB,BB);
216700 END#;
216800 STARTPLOTTER=
216900 LAB2:WRITE(FILE97,<"WHICH GRAPH?"-BODE,NYQUIST,NICHOLS OR RLOCUS?">);
217000 READ(FILE97,/,GRAPH);
217100 IF GRAPH="B" THEN BODE(W,DA,DB,A,B);
217200 IF GRAPH="NY" THEN NYQUIST(W,DA,DB,A,B);
217300 IF GRAPH="NI" THEN NICHOLS(W,DA,DB,A,B);
217400 IF GRAPH="R" THEN RLOCUS(DPOLY1,DPOLY2,POLY1,POLY2);
217500 WRITE(FILE97,<"DO YOU WANT ANYMORE GRAPH?">);

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217600 READ(FILE97,/,MORE);
217700 IF MORE="Y" THEN GO TO LAB2 ELSE
217800 BEGIN
217900 WRITE(FILE97,<"DO YOU WANT TO CHANGE TO NEW VALUES?">);
218000 READ(FILE97,/,NEW);
218100 END;
218200 IF NEW="Y" THEN GO TO LAB4 ELSE GO TO LAB5#;
218300 WRITE(FILE97,<"TIME OR FREQUENCY DOMAIN?">);
218400 READ(FILE97,/,DMM);
218500 IF DMM="T" THEN
218600 LAB8: BEGIN
218700 WRITE(FILE97,<"O/P OR CLOSED LOOP?">);
218800 READ(FILE97,/,OP);
218900 IF OP="O/P" THEN
219000 BEGIN
219100 ARRAY K[0:0];
219200 LABEL LAB10;
219300 LAB10: READCGH1;
219400 TIMEDOMAIN(DA,DB,AA,BB);
219500 WRITE(FILE97,<"WANT TO CHANGE TO NEW VALUES?">);
219600 READ(FILE97,/,NEW);
219700 IF NEW="Y" THEN GO TO LAB10;
219800 END;
219900 END
220000 ELSE
220100 BEGIN
220200 ARRAY K[0:0];
220300 LABEL LAB10;
220400 LAB10: READCGH2;
220500 TIMEDOMAIN(DA,DB,AA,BB);
220600 WRITE(FILE97,<"WANT TO CHANGE TO NEW VALUES?">);
220700 READ(FILE97,/,NEW);
220800 IF NEW="Y" THEN GO TO LAB10;
220900 END;
221000 END;
221100 WRITE(FILE97,<"WANT ANYMORE TIME-DOMAIN PLOT?">);
221200 READ(FILE97,/,MTDP);
221300 IF MTDp="Y" THEN GO TO LAB8 ELSE GO TO LAB7;
221400 END
221500 ELSE
221600 IF DMM="F" THEN
221700 LAB6: BEGIN
221800 WRITE(FILE97,<"WHICH PLOT?">);
221900 READ(FILE97,/,OP);
222000 IF OP="O/P" THEN
222100 BEGIN
222200 LABEL LAB4;
222300 ARRAY K[0:0];
222400 LAB4: READCGH1;
222500 STARTPLOTTER;
222600 END
222700 ELSE IF OP="CLOSED" THEN
222800 BEGIN
222900 LABEL LAB4;
223000 ARRAY K[0:0];
223100 LAB4: READCGH2;
223200
223300 LAB2: WRITE(FILE97,<"WHICH GRAPH?">);
223400 READ(FILE97,/,GRAPH);
223500 IF GRAPH="B" THEN BODE(W,DA,DB,A,B);
223600

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2223700	IF GRAPH="NY" THEN NYQUIST(W,DA,DB,A,B);	00223700
2223800	IF GRAPH="NI" THEN NICHOLS(W,DA,DB,A,B);	00223800
2223900	WRITE(FILE97,<"DO YOU WANT ANYMORE GRAPH?">);	00223900
2224000	READ(FILE97,/,MORE);	00224000
2224100	IF MORE="Y" THEN GO TO LAB2 ELSE	00224100
2224200	BEGIN	00224200
2224300	WRITE(FILE97,<"DO YOU WANT TO CHANGE TO NEW VALUES?">);	00224300
2224400	READ(FILE97,/,NEW);	00224400
2224500	END;	00224500
2224600	END;	00224600
2224700	IF NEW="Y" THEN GO TO LAB4 ELSE GO TO LAB5;	00224700
2224800	END;	00224800
2224900	END;	00224900
2225000	LAB5: BEGIN END;	00225000
2225100	WRITE(FILE97,<"DO YOU WANT ANYMORE FREQUENCY"DOMAIN PLOT?">);	00225100
2225200	READ(FILE97,/,DUI);	00225200
2225300	IF DUI="Y" THEN GO TO LAB6 ELSE GO TO LAB7;	00225300
2225400	LAB7: IF DMM="T" THEN	00225400
2225500	BEGIN	00225500
2225600	WRITE(FILE97,<"WHAT ABOUT FREQ. DOMAIN PLOT?">);	00225600
2225700	READ(FILE97,/,DUI);	00225700
2225800	IF DUI="Y" THEN GO TO LAB6 ELSE GO TO LAB9;	00225800
2225900	END	00225900
2226000	ELSE IF DMM="F" THEN	00226000
2226100	BEGIN	00226100
2226200	WRITE(FILE97,<"WHAT ABOUT TIME"DOMAIN RESPONSE?">);	00226200
2226300	READ(FILE97,/,DUI);	00226300
2226400	IF DUI="Y" THEN GO TO LAB8 ELSE GO TO LAB9;	00226400
2226500	END;	00226500
2226600	LAB9: END.	00226600